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A COMPUTER CODE FOR THE SOLUTION OF THE  
EQUATIONS GOVERNING A LAMINAR, PREMIXED,  
ONE-DIMENSIONAL FLAME

Terence P. Coffee

April 1982



US ARMY ARMAMENT RESEARCH AND DEVELOPMENT COMMAND  
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## I. INTRODUCTION

We are interested in determining validated sets of elementary chemical reactions for use in predictive combustion models. For this purpose, a simulation of the laminar, premixed, one dimensional, steady state flame has been implemented. This has the advantage of being a relatively simple combustion problem that also yields predicted temperature and species profiles that can be compared with suitable burner experiments.

The basic idealism is of an infinite column of premixed gas. At some point the gas is ignited. A flame front forms and propagates along the column of gas. All effects are assumed to be one-dimensional. Eventually, the flame will reach steady state. A basic characteristic of the solution is the flame speed, the velocity with which the flame front moves with reference to the undisturbed gas mixture. The flame front is marked by steep temperature and chemical species profiles. The above formulation is referred to as an unbounded, or adiabatic flame.

In practice, a flame is usually stabilized on a burner. That is, the flame will propagate to the burner surface. Loss of heat to the burner will stabilize the flame. This creates different boundary conditions.

This report describes a method of numerically solving the equations governing a laminar, premixed, one-dimensional, steady state flame. Both unbounded flames and burner stabilized flames are discussed. To solve such systems, we have modified a standard package for integrating one-dimensional partial differential equations so that it efficiently handles flame equations. The basic procedure is to integrate the equations in time until the steady state solution is reached. Changes have been made in the basic structure of the code, as well as adding a number of subroutines. The program has been implemented for the ozone flame and the  $H_2-O_2-N_2$  flame.

Section II describes the partial differential equations governing the flames. In Section III, the basic code used to solve the equations is introduced. Section IV describes the modifications made in the code, and Section V the additional subroutines added to the program. Section VI discusses the changes in the equations and the code necessary to model burner stabilized flames. Section VII reports some of the numerical considerations in successfully using the code. Finally, the Appendix gives a complete listing of the code, plus the output for a run modeling a typical  $H_2-O_2-N_2$  flame.

## II. THE UNBOUNDED FLAME EQUATIONS

The equations for a multicomponent reacting ideal gas mixture can be found in the literature.<sup>1-4</sup> We are interested in the equations that describe a one-dimensional, laminar, premixed flame that propagates in an unbounded medium. The effects of radiation, viscosity and body forces are ignored. The momentum equation

$$\hat{\rho} \frac{\partial \hat{v}}{\partial t} = - \hat{\rho} \hat{v} \frac{\partial \hat{v}}{\partial x} - \frac{\partial \hat{p}}{\partial x}$$

can be eliminated. Here  $\hat{v}$  is the fluid velocity,  $\hat{\rho}$  is the density, and  $\hat{p}$  is the pressure. The independent variables are time  $\hat{t}$  (sec) and distance  $\hat{x}$  (cm). For flames, the fluid velocity is much less than the speed of sound, that is,  $\hat{v}^2 \ll \hat{p}/\hat{\rho}$ . For steady state flow, this implies that the pressure gradient is negligibly small. So the pressure is assumed to be constant.

The pertinent equations are then, overall continuity.

$$\frac{\partial \hat{\rho}}{\partial t} = - \frac{\partial (\hat{\rho} \hat{v})}{\partial x} \quad (1)$$

That is, any change in density with respect to time is due to the overall convection of the mixture, which equals the gradient of the total mass flux  $\hat{\rho} \hat{v}$ .

Continuity of species:

$$\hat{\rho} \frac{\partial \hat{Y}_k}{\partial t} = - \hat{\rho} \hat{v} \frac{\partial \hat{Y}_k}{\partial x} - \frac{\partial}{\partial x} (\hat{\rho} \hat{Y}_k \hat{V}_k) + \hat{R}_k M_k, \quad k = 1, 2, \dots, N \quad (2)$$

<sup>1</sup>F. A. Williams, *Combustion Theory*, Addison-Wesley, Reading, MA, Chapter 1, 1965.

<sup>2</sup>R. B. Bird, W. E. Stewart and E. N. Lightfoot, *Transport Phenomena*, John Wiley and Sons, NY, Chapter 18, 1960.

<sup>3</sup>R. M. Fristram and A. A. Westenberg, *Flame Structure*, McGraw-Hill, NY, Chapter V-1, 1965

<sup>4</sup>J. O. Hirshfelder, C. F. Curtiss and R. B. Bird, *Molecular Theory of Gases and Liquids*, 2nd Printing corrected, with notes, John Wiley and Sons, NY, Chapter 11.1, 1964.

where  $\hat{Y}_k$  is the mass fraction of the  $k^{\text{th}}$  species,  $\hat{V}_k$  is its diffusion velocity,  $M_k$  is its molecular weight, and  $\hat{R}_k$  is the rate at which the species is produced or consumed by the chemical reactions. So changes in the concentration of the  $k^{\text{th}}$  species can be due to the convection of the species, the diffusion of the species in the mixture, or the production or consumption of the species by chemistry. The term  $\rho \hat{Y}_k \hat{V}_k$  is the mass flux of the  $k^{\text{th}}$  species (relative to  $\hat{v}$ ). Note that the total mass flux for the  $k^{\text{th}}$  species is  $\hat{\rho}(\hat{v} + \hat{V}_k) \hat{Y}_k$ .

Conservation of Energy.

$$\hat{\rho} \hat{c}_p \frac{\partial \hat{T}}{\partial t} = - \hat{\rho} \hat{v} \hat{c}_p \frac{\partial \hat{T}}{\partial x} + \frac{\partial}{\partial x} \left( \hat{\lambda} \frac{\partial \hat{T}}{\partial x} \right) - \sum_{k=1}^N \hat{R}_k M_k \hat{h}_k - \hat{\rho} \sum_{k=1}^N \hat{c}_{pk} \hat{Y}_k \hat{V}_k \frac{\partial \hat{T}}{\partial x} \quad (3)$$

where  $\hat{T}$  is the temperature,  $\hat{c}_{pk}$  is the specific heat of the  $k^{\text{th}}$  species,  $\hat{c}_p$  is the specific heat of the mixture,  $\hat{\lambda}$  is the thermal conductivity of the mixture, and  $\hat{h}_k$  is the specific enthalpy of the  $k^{\text{th}}$  species. So changes in the temperature can be due to the convection of heat, the conduction of heat, the production or consumption of energy by the chemical reactions, and a small amount due to the diffusion of species with different specific heats. We have not written a negligibly small term due to the species gradients (Dufour effect).

The thermal equation of state is given by the ideal gas law

$$\hat{p} = \hat{\rho} R \hat{T} \sum_{k=1}^N \hat{Y}_k / M_k, \quad (4)$$

where  $R$  is the gas constant. The caloric equation of state is

$$\hat{h}_k = \hat{h}_k^0 + \int_{\hat{T}_0}^{\hat{T}} \hat{c}_{pk} d\hat{T}, \quad (5)$$

where  $\hat{h}_k^0$  is the specific enthalpy of the  $k^{\text{th}}$  species at some reference temperature  $\hat{T}_0$ . The specific heat of the mixture is given by

$$\hat{c}_p = \sum_{k=1}^N \hat{c}_{pk} \hat{Y}_k. \quad (6)$$

From conservation of mass we have the relation

$$\sum_{k=1}^N \hat{Y}_k = 1 \quad (7)$$

and

$$\sum_{k=1}^N \hat{i}_k \hat{V}_k = 0. \quad (8)$$

The boundary conditions are the following. For  $x = -\infty$ .

$$\hat{T} = \hat{T}_U \text{ and } \hat{Y}_k = \hat{Y}_{kU}, \quad (k=1,2,\dots,N), \quad (9)$$

where  $\hat{T}_U$  is the temperature and the  $\hat{Y}_{kU}$  are the mass fractions of the original, undisturbed mixture. For  $x = \infty$

$$\hat{T} = \hat{T}_B \text{ and } \hat{Y}_k = \hat{Y}_{kB}, \quad k=1,2,\dots,N. \quad (10)$$

where  $\hat{T}_B$  is the temperature and  $\hat{Y}_{kB}$  are the mass fractions of the burned mixture.  $\hat{T}_B$  is called the adiabatic temperature. Since we are assuming no heat loss to the surroundings,  $T_B$  and  $Y_{kB}$  depend only on the chemistry, and can be calculated in advance.

In practice, the integration will be over a finite interval  $(\hat{x}_L, \hat{x}_R)$ , where the flame front will be located roughly in the center of the interval. The boundary conditions (9) will be applied to  $\hat{x}_L$ . However, the conditions (10) are not convenient. After the flame front, there is a long recombination period before the mixture reaches adiabatic conditions. Choosing the interval of integration long enough to cover the entire recombination zone would be computationally expensive. So normally the flame will be cut off before it reaches adiabatic conditions,

and the weaker boundary conditions

$$\frac{\partial \hat{T}}{\partial \hat{x}} = \frac{\partial Y_k}{\partial \hat{x}} = 0, \quad k=1,2,\dots,N, \quad (11)$$

will be applied at  $\hat{x}_R$ .

At this stage the partial differential Eq. (1) for the density  $\hat{\rho}$  can be eliminated by introducing a new coordinate  $\hat{\psi}$  such that

$$\hat{\psi}(\hat{x}, \hat{t}) = \int_{\hat{x}_L}^{\hat{x}} \hat{\rho}(\hat{x}', \hat{t}) d\hat{x}'. \quad (12)$$

Then  $\frac{\partial \hat{\psi}}{\partial \hat{x}} = \hat{\rho}$  and  $\frac{\partial \hat{\psi}}{\partial \hat{t}} = -\hat{\rho}\hat{v} + \hat{m}_0(\hat{t})$ , where  $\hat{m}_0(\hat{t}) = \hat{\rho}\hat{v}|_{\hat{x}=0}$ . With this

notation Eqs. (2) and (3) become

$$\frac{\partial Y_k}{\partial \hat{t}} = -\tilde{m}_0 \frac{\partial Y_k}{\partial \hat{\psi}} - \frac{\partial}{\partial \hat{\psi}} (\tilde{P} Y_k \tilde{V}_k) + \tilde{R}_k M_k / \tilde{\rho} \quad (13)$$

and

$$\frac{\partial \hat{T}}{\partial \hat{t}} = -\tilde{m}_0 \frac{\partial \hat{T}}{\partial \hat{\psi}} + \frac{1}{\tilde{c}_p} \left\{ \frac{\partial}{\partial \hat{\psi}} \left( \tilde{\rho} \tilde{\lambda} \frac{\partial \hat{T}}{\partial \hat{\psi}} \right) - \sum_{k=1}^N \tilde{R}_k M_k \tilde{h}_k / \tilde{\rho} - \sum_{k=1}^N \tilde{c}_{pk} \tilde{Y}_k \tilde{V}_k \frac{\partial \hat{T}}{\partial \hat{\psi}} \right\}, \quad (14)$$

where the tilde variables are functions of  $\tilde{t} = \hat{t}$  (sec) and  $\tilde{\psi}$  (gm/cm<sup>2</sup>).

For numerical convenience, dimensionless forms of Eqs. (13) and (14) are integrated. That is, we will define  $T = \hat{T}/T_\infty$ ,  $t = \tilde{t}/t_\infty$  and  $\psi = \tilde{\psi}/\psi_\infty$ , where  $t_\infty$  and  $\psi_\infty$  are chosen so as to obtain reasonable time and space scales for a given flame, and  $T_\infty$  is chosen so that  $T$  is the same order of magnitude as the larger  $Y_k$ 's. Then the equations are

$$\frac{\partial Y_k}{\partial \tau} = - \frac{t_\infty}{\psi_\infty} m_0 \frac{\partial Y_k}{\partial \psi} - \frac{t_\infty}{\psi_\infty} \frac{\partial}{\partial \psi} (\rho Y_k V_k) + t_\infty R_k M_k / \rho, \quad (15)$$

and

$$\begin{aligned} \frac{\partial T}{\partial \tau} = & - \frac{t_\infty}{\psi_\infty} m_0 \frac{\partial T}{\partial \psi} + \frac{t_\infty}{c_p} \left\{ \frac{1}{\psi_\infty^2} \frac{\partial}{\partial \psi} (\rho \lambda \frac{\partial T}{\partial \psi}) \right. \\ & \left. - \sum_{k=1}^N R_k M_k h_k / (\rho T_\infty) - \frac{1}{\psi_\infty} \sum_{k=1}^N c_{pk} Y_k V_k \frac{\partial T}{\partial \psi} \right\}, \end{aligned} \quad (16)$$

where the variables are functions of  $\tau$  and  $\psi$ . The boundary conditions are

$$T = T_U, \quad Y_k = Y_k U, \quad k = 1, 2, \dots, N \quad (17)$$

at  $\psi = \psi_L$  and

$$\frac{\partial T}{\partial \psi} = \frac{\partial Y_k}{\partial \psi} = 0, \quad k = 1, 2, \dots, N \quad (18)$$

at  $\psi = \psi_R$ .

The integration can start from any initial profiles, as long as there is enough energy available to begin the combustion. As shown in Section V, the mass flux  $m_0$  through the origin is modified as the integration proceeds so as to keep the flame front in the center of the interval of integration  $(\psi_L, \psi_R)$ . The integration proceeds until  $m_0$  approaches a constant, and the terms  $\frac{\partial Y_k}{\partial \tau}$  and  $\frac{\partial T}{\partial \tau}$  approach zero. The equation for species  $N$  is not integrated;  $Y_N$  is found from the relation (7).

The burning velocity,  $S_b$ , can be calculated at steady state. From the continuity equation  $\partial(\rho v)/\partial \psi = 0$ , or  $\rho v$  is constant with respect to  $\psi$ . Then we can take any of the equations (15) and integrate over any interval  $(a, b)$  to obtain

$$\rho v_k [Y_k(b) - Y_k(a)] = \psi_\infty \int_a^b \rho^{-1} R_k M_k d\psi - \rho Y_k V_k \Big|_a^b. \quad (19)$$

Then

$$v_k(-\infty) = \frac{\psi_\infty \int_a^b \rho^{-1} R_k M_k d\psi - \rho Y_k V_k \Big|_a^b}{\rho(-\infty) [Y_k(b) - Y_k(a)]}. \quad (20)$$

At steady state, all of the  $v_k$  are equal. Since in the present coordinate system the flame does not move,  $v_k(-\infty)$  is the speed at which the unburned gas is approaching the flame. Conversely,  $S_V = v_k(-\infty)$  is the speed at which the flame is propagating into the unburned mixture.

The parameters of the equations (15) and (16) are the specific heats  $c_{pk}$ , the specific enthalpies  $h_k$ , the chemistry production terms  $R_k$ , and the diffusion velocities  $V_k$  for each species, plus the thermal conductivity  $\lambda$  of the mixture. The  $c_{pk}$  and  $h_k$  are functions only of the temperature, and can be evaluated very accurately using sixth degree polynomial fits<sup>5</sup>. The chemistry is generally the least well known of the input data. We need to know which species are involved, which reactions can occur, and the rate constant for each reaction. The rate constants  $k_j$  will be of the form  $A T^b \exp(-C/T)$ , where either  $b$  or  $C$  can be zero. We can then find the rate  $r_j$  for each reaction by multiplying the rate constant times the concentrations of the reactants (concentrations of the  $k^{\text{th}}$  species  $= \rho Y_k / M_k$ ). Each  $R_k$  is then found by adding the rates of the reactions in which the  $k^{\text{th}}$  species is a reactant. The transport parameters  $\lambda$  and  $V_k$  are in general very complicated functions of temperature and species concentrations. Because of the numerical complexity, various approximations to these quantities have been used. We have the capability to run the code with a number of different levels of approximation. This is discussed in detail in another paper<sup>6</sup>.

<sup>5</sup>S. Gordon and B. J. McBride, "Computer Program for Calculation of Complex Chemical Equilibrium Compositions, Rocket Performance, Incident and Reflected Shocks, and Chapman-Jouguet Detonations", NASA-SP-273, 1971, (1976 program version).

<sup>6</sup>T. P. Coffee and J. M. Heimerl, "Transport Algorithms for Premixed, Laminar, Steady-State Flames", to be published in Combustion and Flame.

For future reference, we will discuss here the simplest transport algorithm used in the code. First assume that Ficks law,

$$\hat{Y}_k \hat{V}_k = - \hat{D}_{k m} \frac{\partial \hat{Y}_k}{\partial \hat{x}} \quad (21)$$

is valid. This is only strictly true for a binary mixture where thermal diffusion is negligible.  $\hat{D}_{k m}$  is a multicomponent diffusion coefficient. Equation (21) can be written as

$$\rho Y_k V_k = - \frac{\rho^2 D_{k m}}{\psi_\infty} \frac{\partial Y_k}{\partial \psi} \quad (22)$$

As a first approximation, we can take  $\rho^2 D_{k m}$  and  $\rho \lambda$  to be constant. Also, we can assume that  $c_p = c_{p k}$ ,  $k=1,2,\dots,N$ . Here  $c_p$  must be chosen so as to obtain the proper adiabatic temperature  $T_B$ . The procedure for choosing these constants and the justification of this approximation is given in reference 6. This level of approximation is reasonably accurate, if the constants are properly chosen.

Equations (15) and (16) now become

$$\frac{\partial Y_k}{\partial t} = - \frac{t_\infty}{\psi_\infty} m_0 \frac{\partial Y_k}{\partial \psi} + \frac{t_\infty}{\psi_\infty} \rho^2 D_{k m} \frac{\partial Y_k}{\partial \psi^2} + t_\infty R_k M_k / \rho \quad (23)$$

and

$$\frac{\partial I}{\partial t} = - \frac{t_\infty}{\psi_\infty} m_0 \frac{\partial T}{\partial \psi} + \frac{t_\infty}{c_p} \left\{ \frac{\rho \lambda}{\psi_\infty^2} \frac{\partial^2 T}{\partial \psi^2} - \sum_{k=1}^N \frac{R_k M_k h_k}{\rho T_\infty} \right\} \quad (24)$$

Equation (5) now simplifies to

$$h_k = h_k^0 + c_p (T - T_0) \quad (25)$$

so the required input parameters for this level of approximation are the constants  $\rho^2 D_{k m}$ ,  $\rho \lambda$ ,  $c_p$ ,  $h_k^0$  and the functions  $R_k(T, Y_1, \dots, Y_N)$ .

### III. THE NUMERICAL METHOD - PDECOL

The package PDECOL, developed by Madsen and Sincovec<sup>7</sup>, was used to solve the equations. This package is designed to solve a general system of  $N$  nonlinear partial differential equations of at most second order on a finite interval. In our coordinate system the appropriate form is

$$\frac{\partial \vec{u}}{\partial t} = \vec{f}(t, \psi, \vec{u}, \vec{u}_\psi, \vec{u}_{\psi\psi}), \quad (26)$$

where

$$\vec{u} = (Y_1, \dots, Y_{N-1}, T) \quad (27)$$

$$\vec{u}_\psi = \left( \frac{\partial Y_1}{\partial \psi}, \dots, \frac{\partial Y_{N-1}}{\partial \psi}, \frac{\partial T}{\partial \psi} \right),$$

$$\vec{u}_{\psi\psi} = \left( \frac{\partial^2 Y_1}{\partial \psi^2}, \dots, \frac{\partial^2 Y_{N-1}}{\partial \psi^2}, \frac{\partial^2 T}{\partial \psi^2} \right).$$

Fairly general boundary conditions of the form

$$\vec{b}(\vec{u}, \vec{u}_\psi) = \vec{Z}(t) \quad (28)$$

are allowed, where  $\vec{b}$  and  $\vec{Z}$  are arbitrary vector valued functions with  $N$  components. Each solution component is assumed to be a known function of  $\psi$  at the initial time  $t = t_0$ . That is,  $Y_k(t_0, \psi)$ ,  $k=1,2,\dots,N-1$ , and  $T(t_0, \psi)$  are known functions. The initial conditions must be consistent with the boundary conditions.

The spatial discretization is accomplished by finite element collocation methods based on B-splines<sup>8</sup>. The user must supply a set of

<sup>7</sup>N. K. Madsen and R. F. Sincovec, "PDECOL: General Collocation Software for Partial Differential Equations", Preprint UCRL-78263 (Rev 1), Lawrence Livermore Laboratory, (1977).

<sup>8</sup>C. de Boor, "Package for Calculating with B-Splines", Siam, J. Numer. Anal. 14, 441-472, (1977).

NB breakpoints, that is, a set of strictly increasing locations where the polynomials are joined. He must also supply the order KORD of the splines and the number of continuity conditions, NCC, to be applied at the breakpoints. That is, if NCC=1, the approximating function is continuous; if NCC=2, it is continuous and smooth; and so on. PDECOL then generates a set of  $NC = KORD(NB-1) - NCC(NB-2)$  basis functions and collocation points. The basis functions  $B_i(\psi)$  are piecewise polynomials of order KORD - 1. The basic assumption is that the solution can be written in the form

$$u_k = \sum_{i=1}^N c_k^{(i)}(t) B_i(\psi), \quad k=1, \dots, N, \quad (29)$$

where the basis functions  $B_i(\psi)$  span the solution space for any fixed  $t$  to within a small error tolerance. The time dependent coefficients  $c_k^{(i)}$  are determined uniquely by requiring that the expansions above satisfy the given boundary conditions and that they satisfy the partial differential equations exactly at the (N-2) interior (collocation) points. Since by definition a B-spline is zero except over a small interval, at any collocation point no more than KORD of the B-splines are non-zero. So the system of ODE's for the coefficients  $c_k^{(i)}$  will not be fully coupled.

The boundary conditions (28) must also be changed into ordinary differential equations. This can be done by taking the derivative with respect to time, which results in

$$\sum_{j=1}^N \left[ \frac{\partial b_k}{\partial u_j} \frac{\partial u_j}{\partial t} + \frac{\partial b_k}{\partial u_{j\psi}} \frac{\partial u_{j\psi}}{\partial t} \right] = \frac{dz_k}{dt}, \quad k=1, 2, \dots, N. \quad (30)$$

When equation (29) is substituted into equation (30), a set of ODE's in the  $c_k^{(i)}$  results.

Unlike a finite difference code, the program can be run with no boundary condition at either the left or the right boundary. The program simply collocates at the boundary, using the same procedure as for the interior points. It must be, of course, the user's responsibility to define a mathematically meaningful PDE problem.

This system of ODE's is integrated in time, using a variant of the Gear stiff integrator<sup>9</sup>. This is a fully implicit, predictor-corrector method. The required banded Jacobian is generated internally by the program. Once the integrator has reached a desired output time, the values of  $Y_k$  and  $T$  can be obtained for any  $\psi$  by substituting into the expansions (29). The accuracy of the time integration is determined by a user supplied error tolerance  $\epsilon$ .

For the actual integration, the code requires the values of the basis functions and the first two space derivatives only at the collocation points. Since computing these values is complicated, this is done once at the beginning of the program and the values are saved.

In general, a user must supply a main program and three subroutines. The program MAIN sets the values of the required parameters, calls the integrator, and writes any desired output. The subroutine UINIT gives the initial conditions. Given a collocation point  $\psi$ , the subroutine must return the vector  $\vec{u}(t_0, \psi)$ . The subroutine F evaluates the function  $\vec{F}$  given by Eq. (26). Given  $t, \psi, u, u_\psi$ , and  $u_{\psi\psi}$ , the routine must return the vector  $\partial u / \partial t$ . The program automatically converts this to the corresponding set of ODE's involving the  $c_k^{(i)}$ . The subroutine BNDRY specifies the boundary conditions. That is, given  $t, \psi, \vec{u}$ , and  $\vec{u}_\psi$ , the subroutine returns the quantities  $\partial b_k / \partial u_j, \partial b_k / \partial u_j, \partial Z_k / \partial t$ ,  $j, k = 1, 2, \dots, N$ , for both  $\psi_L$  and  $\psi_R$ .

#### IV. MODIFICATIONS OF PDECOL

The time integration is controlled by a user supplied error tolerance  $\epsilon$ . Single step error estimates divided by  $C_{MAX_k}^{(i)}$  will be kept less than  $\epsilon$  in the root-mean-square norm. In PDECOL,  $C_{MAX_k}^{(i)}$

is initially set to the maximum of  $|c_k^{(i)}|$  and 1.0. Thereafter,  $C_{MAX_k}^{(i)}$  is the largest value of  $c_k^{(i)}$  seen so far, or the initial  $C_{MAX_k}^{(i)}$  if that is larger. However, this error criterion does not produce the desired accuracy for flame simulations, because radical species with small concentrations will control the flame, and must be computed accurately. But mass fractions much smaller than 1.0 will not be computed accurately using the original PDECOL criterion. An alternate criterion is the purely relative error criterion, that is,  $C_{MAX_k}^{(i)} = |c_k^{(i)}|$ . This criterion is also unacceptable because some species at some locations will approach zero, and excess computation will result

<sup>9</sup>A. C. Hindmarsh, "Preliminary Documentation of GEARIB: Solution of Implicit Systems of Ordinary Differential Equations with Banded Jacobian", Rep. UCID-30130, Lawrence Livermore Laboratory, (1976).

in accurately computing these negligible concentrations. Consequently a semi-relative error control is used.  $C_{MAX_k}^{(i)}$  is chosen as the maximum of  $c_k^{(i)}$  and a user supplied parameter SREC. Thus, mass fractions less than SREC will be computed less accurately. We have normally used  $SREC=10^{-6}$ .

The original program PDECOL is fully implicit. That is, it generates a set of  $NO = N \times NC$  ordinary differential equations of the form

$$A \frac{d\vec{c}}{dt} = \vec{g}(t, \vec{c}) \quad , \quad (31)$$

where  $c = (c_1^{(1)} \dots c_N^{(1)}, c_1^{(2)} \dots c_N^{(2)} \dots)$ . The resulting Jacobian  $\partial \vec{g} / \partial \vec{c}$  is a banded  $NO$  by  $NO$  matrix. The band width is  $3 \times ML + 1$ , where  $ML = N(KORD - 1) - 1$ . To advance a time step, the program first computes an explicit predictor for the values of the  $c_k^{(i)}$  at the next time step. Then it solves a set of linear algebraic equations involving the Jacobian to correct the values. For a stiff system, such as one involving chemistry, this allows time steps orders of magnitude larger than those of an explicit method<sup>9</sup>. The drawback is the storage and execution time required to work with the Jacobian.

For example, consider a system where  $KORD = 4$ ,  $N = 10$ , and  $NC = 20$ . The bandwidth of the Jacobian is 88, and we have a system of 300 ODE's. We then have 26,400 possible non-zero elements in the banded matrix. The amount of storage required also increases rapidly. For instance, if  $N = 20$  instead of 10 the bandwidth will be 178, and we will require 106,800 words to store the Jacobian. Since we want to be able to solve systems at least this large, the storage requirements become almost prohibitive. In addition, solving such large systems of linear algebraic equations is very time consuming.

To avoid this problem, we essentially uncouple the partial differential equations and solve them successively. The basic procedure is illustrated below. Suppose we are at time  $t_n$  and we want to advance a time step to  $t_{n+1}$ . We first integrate the  $n$  equation for  $Y_1$ , under the assumption that  $Y_2, \dots, Y_{N-1}$ ,  $T$  are constant at their  $t_n$  values. This uncouples the first PDE from the system. Subsequently, we integrate the second PDE for  $Y_2$ , using the new value of  $Y_1$  at  $t_{n+1}$ , and the old values for the other variables. Continuing this process, we finally solve for  $T(t_{n+1})$ , using updated values for all the mass fractions.

In general, this method is restricted to smaller step sizes than a fully implicit method. However, as steady state is approached, all the variables approach constants with respect to time, and the temporal coupling vanishes.

Now consider how this assumption affects the associated system (31) of ODE's. In the Jacobian, we will have

$$\partial g_k^{(i)} / \partial c_\ell^{(j)} = 0 \text{ if } k \neq \ell.$$

That is, changes in the  $\ell^{\text{th}}$  PDE will not affect the  $k^{\text{th}}$  PDE if  $k \neq \ell$ . So most of the terms in the Jacobian will become zero.

The remaining nonzero elements are not in banded form. However, we can accomplish this by rearranging the vector  $\vec{c} = (c_1^{(1)}, c_1^{(2)}, \dots, c_1^{(\text{NC})}, c_2^{(1)}, \dots, c_2^{(\text{NC})}, \dots)$ . That is, the coefficients for each PDE are grouped together instead of the coefficients for each collocation point.

The Jacobian matrix can now be decomposed into  $N$  smaller  $\text{NC} \times \text{NC}$  matrices on the main diagonal. Moreover, the band width of these matrices is only  $3\text{XML}+1$ , where  $\text{ML} = \text{KORD} - 2$ . The Jacobian is essentially  $N$  separate Jacobians, each for one PDE with  $\text{NC}$  collocation points.

The savings in storage space is dramatic. For our previous example of  $\text{KORD} = 4$ ,  $N = 10$ , and  $\text{NC} = 30$ , we have 2100 nonzero elements instead of 26,400. If  $N = 20$ , we have only 4200 nonzero elements instead of 106,800.

The following procedure is used to actually integrate a time step. The predictor of the predictor-corrector method is used to obtain first estimates for all the  $c_k^{(1)}$ . Then the corrected  $c_1^{(1)}$  are computed, using the first small Jacobian and assuming that the other  $c_k^{(1)}$  do not change. Then the corrected values for  $c_1^{(1)}$  and the predicted values for  $c_k^{(1)}$ ,  $k > 2$ , are used, and the corrected  $c_2^{(1)}$  are computed. This process continues through the  $N$  PDE's. The estimated error is calculated. If necessary, the above procedure is iterated. It is more efficient to use the predicted values rather than the values at the previous time step. In using the procedure, it is more efficient to integrate the minor species first, since they change most rapidly, then the major species, and finally the temperature.

The above procedure is similar to one developed by Spalding and Stephenson for use in a finite difference code<sup>10</sup>.

<sup>10</sup> D.B. Spalding and P.L. Stephenson, "Laminar Flame Propagation in Hydrogen + Bromine Mixtures", *Proc. R. Soc. Lond. A.* **324**, 315-337 (1971).

PDECOL was rewritten to use the successive calculation method. This also means that we must change the user supplied routines F and BNDRY.

Because of the rearrangement of the vector  $\vec{C}$ , the core integrater will call F first at each collocation point, asking for the values of  $\partial Y_1 / \partial t$ . It will then repeat for each PDE. The routine F is written so as to return only the desired time derivative.

Since the problem can no longer directly handle coupling terms, the boundary conditions (28) must be uncoupled, that is, the boundary conditions must be of the form

$$b_k(u_k, u_{k\psi}) = Z_k(t) \quad (32)$$

The rewritten subroutine BNDRY evaluates the quantities  $\partial b_k / \partial u_k$ ,  $\partial b_k / \partial u_{k\psi}$ ,  $\partial Z_k / \partial t$ ,  $k = 1, 2, \dots, N$ .

Comparisons of the execution time of the fully implicit method versus the successive calculation method have not been made. However, our main purpose was not to reduce the execution time but to reduce the storage requirements. This iterative procedure accomplishes this goal and simultaneously gives accurate results in reasonable run times.

## V. ADDITIONS TO PDECOL

The computer program PDECOL requires a set of user supplied subroutines. We have written a set of subroutines that casts the equations into a computationally efficient form and which generates the required output. These routines are the first seven listed in the appendix; namely MAIN, F, UINIT, FLSP, BNDRY, BKPT, and RT. In addition, several auxiliary programs are mentioned, but without providing a listing. These routines generate sets of input data or actual subroutines that are used frequently, or analyze the output of the flame code.

The code can be run with several different options. We first describe the case of an unbounded flame (NBURN = 0) using the simplest constant transport algorithm (NTRAN = 1). There is no information available about the flame speed or the solution profiles (NSTART = 1). Later in this section we describe restarting the integration (NSTART = 2), and using a more complicated transport algorithm (NTRAN = 2). Burner stabilized flames (NBURN = 1) are discussed in the next section.

Initially, a chemistry scheme must be chosen. In the appendix an  $H_2 - O_2 - N_2$  system is used, with nine chemical species ( $H$ ,  $OH$ ,  $O$ ,  $HO_2$ ,  $H_2O_2$ ,  $H_2$ ,  $O_2$ ,  $H_2O$ ,  $N_2$ ) and a set of thirty reactions involving these species, each with a rate constant of the form  $a T^b \exp(c/T)$  (either  $b$  or  $c$  can be zero). This information is used as input to an auxiliary

code. This code writes the subroutine RT that computes the chemistry terms  $R_k M_k / \rho$ . The procedure is analogous to that used to write the transport subroutines<sup>6</sup>. This subroutine is attached to the flame code in the job stream and can be used for any problem involving this set of kinetics.

The actual subroutine RT has three main parts. The rate constants are evaluated for the current temperature and stored in the vector RK. Since the code uses a successive calculation method, the subroutine will be called N times at each time step and each collocation point. The rate constants are only recomputed if the temperature has been changed. Otherwise this section is skipped. Each rate constant is multiplied by the concentrations of the appropriate reactants divided by  $\rho$  to obtain  $r_j / \rho$ . The terms  $R_k M_k / \rho$  (stored in the vector R) are calculated by adding the rates for the reactions in which  $Y_k$  is a product, subtracting the rates for which  $Y_k$  is a reactant, and multiplying by  $M_k$ .

The choice of a transport algorithm determines the form of the subroutine F. The version given in the appendix is for constant transport. Common statements are used to make the appropriate constants available for all subroutines. The chemistry terms required are found by calling RT.

The initial temperature  $T_U$  and mass fractions  $Y_{kU}$  of the unburned gas are input quantities. Another auxiliary code determines the adiabatic temperatures  $T_B$  and mass fractions  $Y_{kB}$ , and the constants  $\rho^2 D_{km}^0$ ,  $h_k^0$ ,  $\rho \lambda$ , and  $c_p$  needed by the constant transport algorithm (see reference 6). This information is saved on a data file and attached to TAPE 11 when the code is run.

The remaining data necessary to run the program is read in on cards (TAPE 5). The pressure  $p$  and the normalizing constants  $t_\infty$ ,  $\psi_\infty$ , and  $T_\infty$  are specified. The option parameters NSTART, NTRAN, and NBURN are chosen. The numerical parameters required by the code are specified; that is,  $\psi_L$  and  $\psi_R$ , the final integration time  $t_{FINAL}$ , the time integration error control parameters  $\epsilon$  and SREC, and a set of breakpoints.

The major difficulty in efficiently solving the flame equations is choosing an appropriate set of breakpoints. These must be close enough that spatial errors are minimized and yet not so dense that one's computer resources are exceeded. The breakpoints should be densest in the flame front, where the gradients are very steep.

The technique in the code is to use a static mesh, with the breakpoints most closely spaced near the center of the interval of integration. The flame front is then forced to remain near the center of the interval. This is done by adjusting  $m_0$ , the mass flux through the origin. At steady state, the mass flux through the flame is a constant. So  $m_0$  is

iteratively modified to match the steady state mass flow through the flame. This leads to a coordinate system in which the flame front is at rest. The transient behavior can cause the flame front to drift away from the center. This can also be corrected by modifying  $m_0$ .

This method requires a procedure for tracking the flame front. To do this the position of a specific temperature  $T_{cn}$  is monitored. The code attempts to keep this temperature located at the center of the interval of integration. As a heuristic rule this temperature is defined by

$$T_{cn} = T_U + 0.4 (T_B - T_U) \quad (33)$$

The average of  $T_U$  and  $T_B$  is not used. This is because the mixture will not normally reach the adiabatic temperature at the end of the flame front. Rather, there is a radical overshoot, and the recombination of these radicals will very slowly raise the temperature.  $T_{cn}$  as defined will usually be close to the center of the flame front. The details of this iterative procedure are discussed in reference 11.

So the breakpoints may be chosen to be densest in the center of the interval of integration. However, it is tedious to have to choose an entire breakpoint sequence for each problem. We have developed a procedure to generate an appropriate type of breakpoint sequence from a small number of parameters. By varying three parameters, a wide variety of breakpoint sequences can be generated.

The user must supply NINT, the number of intervals ( $NB = NINT+1$ ), NCN, the number of intervals of equal length that will be at the center of the interval, and FC, the ratio between the longest intervals (on the boundaries) and the shortest intervals. Also let  $L$  be the total length of the interval of integration, that is,  $L = \psi_R - \psi_L$ . The program generates a set of intervals whose lengths increase by a constant factor  $\alpha$ , where

$$\alpha = \log^{-1} [2(\log FC)/(NINT-NCN)] \quad (34)$$

The common length  $LC$  of the NCN center shortest intervals is

$$LC = L/[NCN + 2 \alpha(\alpha^{(NINT-NCN)/2} - 1)/(\alpha - 1)] \quad (35)$$

The procedure can best be seen by example. Suppose we have  $\psi_L = 0$ ,

<sup>11</sup> T.P. Coffee and J.M. Heimerl, "A Method for Computing the Flame Speed of a Laminar, Premixed, One Dimensional Flame", BRL Technical Report ARBRL-TR-02212, January 1980.

$\psi_R = 10$ ,  $NINT = 12$ ,  $NCN = 4$ , and  $FC = 6$ . Then  $L = 10$ ,  $\alpha = 1.5651$ , and  $LC = .3155$ . The resulting breakpoint sequence is given in Table 1. Note that the 4 center intervals are of the same length, the length of the intervals then increases by a factor of  $\alpha$ , and the two intervals by the boundaries are 6 times as long as the central intervals. So the procedure automatically generates a set of breakpoints that are closest together near the center, with the spacing increasing smoothly toward the boundaries.

Some experimentation is necessary to choose the proper value of the above parameters. However, choosing  $NINT = 12$ ,  $NCN = 4$ , and  $FC$  between 4 and 8 has worked in most of the cases we have tried. Normally we experiment using the constant transport algorithm, and then use a more realistic transport subroutine once we have a good breakpoint sequence.

TABLE 1. THE SET OF BREAKPOINTS GENERATED BY  $L = 10$ ,  $NINT = 12$ ,  $NCN = 4$  and  $FC = 6$ .

<u>Breakpoints</u>	<u>Interval Lengths</u>
0.0	1.8929
1.8929	1.2094
3.1024	.7728
3.8752	.4937
4.3689	.3155
4.6844	.3155
5.0000	.3155
5.3155	.3155
5.6310	.4937
6.1247	.7728
6.8975	1.2094
8.1069	1.8929
10.0000	

The breakpoint sequence is written by the subroutine BKPT. It also generates a larger set of evaluation points by interpolating between the breakpoints. Using Eq. (29), the subroutine VALUES can evaluate the  $Y_k$  and  $T$  at this larger set of points. This information is useful in generating detailed output, such as graphs, or in performing numerical integrations (see below).

The subroutine UINIT writes the initial profiles, where

$$\begin{aligned}\psi_1 &= \psi_L + 0.24 (\psi_R - \psi_L) \\ \psi_2 &= \psi_L + 0.64 (\psi_R - \psi_L)\end{aligned}\tag{36}$$

and

$$Y_k(t_o, \psi) = \begin{cases} Y_{kU}, & \psi_L \leq \psi \leq \psi_1. \\ Y_{kU} + (Y_{kB} - Y_{kU}) \sin \left[ \frac{\pi}{2} \left( \frac{\psi - \psi_1}{\psi_2 - \psi_1} \right)^2 \right]^2, & \psi_1 \leq \psi \leq \psi_2 \\ Y_{kB}, & \psi_2 \leq \psi \leq \psi_R. \end{cases}\tag{37}$$

The temperature  $T$  is defined similarly.

This particular definition will give us  $T = T_{cn}$  at  $\psi_{cn} = 0.5(\psi_L + \psi_R)$ . The choice of the particular function that defines the  $Y_k$  and  $T$  between  $\psi_1$  and  $\psi_2$  is not important. We have used a straight line with success, but defining a smooth function is slightly more efficient.

A requirement of PDECOL is that the initial profiles satisfy the boundary conditions, in our case given by Eqs. (17) and (18). The above initial profiles have the proper unburned values at  $\psi_L$  and are constant (space derivative zero) near  $\psi_R$ .

For this case the subroutine BNDRY has a simple form. At  $\psi_L$ , the subroutine returns the values  $\partial b_k / \partial u_k = 1$ ,  $\partial b_k / \partial u_{k\psi} = 0$ , and  $\partial Z_k / \partial t = 0$ . At  $\psi_R$ , the conditions are  $\partial b_k / \partial u_j = 0$ ,  $\partial b_k / \partial u_{k\psi} = 1$ , and  $\partial Z_k / \partial t = 0$ .

To begin the integration, a starting value for  $m_0$ , the mass flux through the origin is required. To do this, the code ignores the time dependent terms, assumes that the mass flux  $\rho v$  is constant, and uses Eq. (20) to obtain a value of  $\rho v$ . This gives a reasonable starting value for  $m_0$ .

The evaluation of Eq. (20) is carried out in the subroutine FLSP. The integral is approximated using the trapezoidal rule, where the integrand  $R_k M_k / \rho$  is calculated at the evaluation points. The flame speed  $v_k(-\infty)$  is evaluated for each species  $k$  and for several intervals  $(a,b)$ , where  $a = \psi_L$ . The initial value for  $m_0$  is based on species  $N-1$  integrated over the entire interval  $(\psi_L, \psi_R)$ .

The integration is performed over several intervals  $(a,b)$  because of the behavior of the minor species. These are normally close to zero at  $\psi_L$ , reach a peak in the flame front, and are close to zero again at  $\psi_R$ . For these species, integrating from the left boundary to the flame front is much more accurate.

The chemistry terms in Eq. (20) are found by calling RT. The diffusion terms for this case are found using Ficks law, Eq. (22). Since we are assuming that  $\rho^2 D_{km}$  is constant, this value is stored in the vector R2D. The subroutine FLSP also computes the  $x$  values from the relation

$$x(t, \psi) = \psi_\infty \int_{\psi_L}^{\psi} \rho^{-1} d\psi, \quad (38)$$

using the trapezoidal rule, and computes an estimate of the flame thickness.

As the time integration proceeds, control returns to MAIN at a series of output times. The present spatial location of the temperature  $T_{cn}$  is found, and this is used to find the average value of the mass flux since the last output time. The function  $m_0(t)$  is redefined at these times so as to keep the flame front in roughly the same position. The output times are chosen by the program so that the flame front will not drift too far between evaluations. Also FLSP is called so the user can see if the flame speeds computed from the different species profiles are approaching a common value.

At the final time  $t_{FINAL}$ , FLSP also writes an output file. It consists of all the evaluation points  $\psi_i$ , the corresponding  $x_i$ , the

values of  $Y_k$  and  $T$ , plus their first and second derivatives with respect to  $\psi$ . This file can be attached to an output routine. By also attaching  $F$  and  $RT$ , we can compute and print out any quantity in the steady state solution in which we are interested. This file can also be attached to a graphics routine.

Similarly, the program MAIN writes a restart file. This consists of the present location of  $T_{cn}$ , the present value of  $m_0$ , and the values of  $Y_k$  and  $T$  at the collocation points. This file can be used to restart the time integration. It is attached to TAPE1, and the parameter NSTART is set equal to 2. The input parameters read in on cards can now be changed if desired. UINIT will translate these input values to center the flame front, and will use interpolation to find the appropriate values of the starting profiles at the new collocation points. The old value of  $m_0$  is used to start the integration.

So far only the constant transport case ( $NTRAN = 1$ ) has been considered. For more realistic algorithms  $F$  is written by an auxiliary code and attached to the flame code ( $NTRAN = 2$ ). Several different levels of approximation can be used.<sup>6</sup>

In these more complicated algorithms, the diffusion velocities  $V_k$  are coupled, and they must be computed simultaneously. But since the code uses successive calculation, only the value of one  $V_k$  is required on each call to  $F$ . Because the computation is time-consuming, it is preferable not to compute the  $V_k$   $N$  times at each time step for each collocation point. To economize computer time, all the thermodynamic and transport quantities required at all the collocation points for  $k = 1$  are computed, and stored in vectors. For  $k > 1$ , we use the same values, even though some of the  $Y_k$  terms have changed slightly. Only the chemistry terms are reevaluated. Since the chemistry normally changes much more rapidly than the transport, this will still be a good approximation.

In generating an approximation to the Jacobian (using finite differences) it is necessary to recompute the transport each time  $F$  is called, since what is of interest is the effect of changes in  $Y_k$  and  $T$  on the time derivatives. Ignoring the changes in transport leads to an inaccurate Jacobian. However, the rate constants  $k_j$  and the thermodynamic quantities  $c_p$  and  $h_k$  need be recomputed only if  $T$  is changed, since they only depend on temperature.

As will be seen in the next section it will still be useful to formulate the mass flux in a Ficks law form Eq. (22). For the more complicated transport subroutines, we define

$$\rho^{.2} D_{km} = - \frac{\psi_{\infty} \rho Y_k V_k}{\partial Y_k / \partial \psi} . \quad (39)$$

This new quantity  $\rho^2 D_{km}$  is no longer constant with respect to space or time.

## VI. BURNER STABILIZED FLAMES

So far we have only discussed flames that propagate in an unbounded medium. In actual experiments, the flame will usually be stabilized by a burner. It is useful to be able to model this type of experiment so that we can compare experimental and calculated profiles.

Our basic idealization is of a cylindrical, porous plug burner. The premixed gas exits the plug with a constant velocity  $v$ , but the plug prevents back diffusion of the products into the burner. When the gas is ignited, a flame front develops and propagates toward the burner. The gas velocity  $v$  at the burner must be less than the flame velocity, or the flame will be blown off the burner. In the model this will look like an unbounded flame. In an experiment the flame will be extinguished by the surrounding atmosphere. As the flame approaches the burner, the burner surface acts as a heat sink for the flame. The loss of heat slows down the flame velocity, until the flame stabilizes near the surface of the burner at the gas velocity. As the gas velocity  $v$  is decreased, the flame loses more heat to the burner, and stabilizes closer to the burner surface. If  $v$  is made too small in an experiment, the flame can flash back into the burner.

For this kind of burner, air is entrained along the outside edges of the flame. However, the center of the flame will correspond closely to a premixed, laminar, one-dimensional flame.

The only change in the equations is in the left boundary condition at the surface of the burner. Because of back diffusion from the flame to the burner surface, the mass fractions of the unburned mixture are not conserved. However, due to conservation of mass, the mass flux fractions, defined as

$$\epsilon_k = \frac{\rho Y_k v + \rho Y_k V_k}{\rho v} = Y_k + \frac{\rho Y_k V_k}{\rho v} \quad (40)$$

are conserved. Within the burner, the diffusion velocities are zero, and  $\epsilon_{kU} = Y_{kU}$ . So the appropriate boundary conditions at the burner surface are

$$\epsilon_k = Y_{kU} \quad (41)$$

The boundary conditions for the temperature equation depends on how heat

is extracted. We assume that the burner is maintained at a constant temperature. Then the boundary condition is the same as for an unbounded flame,

$$T = T_U. \quad (42)$$

This idealization is discussed by Hirshfelder, Curtiss and Bird<sup>12</sup>.

To implement this in the flame code, the input parameter NBURN is set equal to 1. The fluid velocity at the burner must be specified. Then  $m_0$  is a predetermined constant instead of an adjustable parameter. All the other input data remains the same. However, the code will handle this data differently.

The breakpoints are generated differently, since the flame front will be near the left boundary instead of in the center of the interval of integration. NCN is the number of intervals of equal length at the left boundary. FC is the ratio between the longest interval (at the right boundary) and the shortest interval (at the left boundary). The constant  $\alpha$  is now defined by

$$\alpha = \log^{-1} [(\log FC)/(NINT-NCN)]. \quad (43)$$

The common length LC of the NCN shortest intervals is now

$$LC = L/[NCN + \alpha(\alpha^{(NINT-NCN)} - 1)/(\alpha - 1)]. \quad (44)$$

As an example, consider the same input parameters that we used for an unbounded flame;  $\psi_L = 0$ ,  $\phi_R = 10$ ,  $NINT = 12$ ,  $NCN = 4$ , and  $FC = 6$ .

Then  $L = 10$ ,  $\alpha = 1.2510$  and  $LC = .4181$ . The resulting breakpoint sequence is given in Table 2. The shortest intervals are near the burner surface, where fairly steep gradients are expected.

In general, it is possible to choose a shorter interval of integration  $L$  for a burner stabilized flame. There is no longer a need for a relatively long interval to the left of the flame front in order to approach the unburned conditions. The number of breakpoints can then also be reduced.

The initial profiles are also chosen differently. The code sets

<sup>12</sup>J.O. Hirshfelder, C.F. Curtiss, and R.B. Bird, *op. cit.*, pp. 761-763.

TABLE 2. THE SET OF BREAKPOINTS GENERATED BY L=10, NINT = 12,  
NCN = 4 and FC = 6 (NBURN = 1)

<u>Breakpoints</u>	<u>Interval Lengths</u>
0	.3458
.3458	.3458
.6916	.3458
1.0374	.3458
1.3832	.4326
1.8159	.5412
2.3571	.6771
3.0342	.8471
3.8812	1.0597
4.9409	1.3257
6.2666	1.6585
7.9251	2.0749
10.0000	

$$\psi_1 = \psi_L + 0.1 (\psi_R - \psi_L)$$

(45)

$$\psi_2 = \psi_L + 0.3 (\psi_R - \psi_L) ,$$

and then uses Eq. (37) as before. This flame front is more likely to be close to the position of the final steady state stabilized flame. Also, the flame front is chosen to be narrower. If Eq. (36) is used, the initial transient flame velocity may be less than the velocity of the fluid. The flame then drifts toward the right, and can be carried completely outside the interval of integration before it stabilizes. Choosing a steeper flame front leads to a larger initial flame velocity.

The boundary condition, Eq. (41), can cause some difficulty. It is fairly straightforward for the constant transport case. Using Fick's law, Eq. (22), it can be rewritten as

$$Y_k - \left( \frac{\rho^2 D_{km}}{\psi_\infty m_0} \frac{\partial Y_k}{\partial \psi} \right) \Big|_{\psi_L} = Y_U \quad (46)$$

Recall that PDECOL converts this to a time dependent equation by taking the time derivative. In the form used by PDECOL, Eq. (46) becomes

$$\left[ \frac{\partial Y_k}{\partial t} - \frac{\rho^2 D_{km}}{\psi_\infty m_0} \frac{\partial}{\partial t} \left( \frac{\partial Y_k}{\partial \psi} \right) \right] \Big|_{\psi_L} = 0 \quad (47)$$

So the subroutine BNDRY must now return the values  $\partial b_k / \partial u_k = 1$ ,  $\partial b_k / \partial u_k = -\rho^2 D_{km} / \psi_\infty m_0$ , and  $\partial Z_k / \partial t = 0$  at  $\psi_L$  for  $k = 1, 2, \dots, N-1$ . The mass flux fractions will then have the proper values as the integration proceeds.

For more realistic transport algorithms the mass flux  $\rho Y_k V_k$  is a complicated function of all the mass fractions. But by using Eq. (39) to define  $\rho^2 D_{km}$ , the boundary condition can be put in the same form as Eq. (46), except that  $\rho^2 D_{km}$  is not a constant. Taking the time derivative we obtain

$$\left\{ \frac{\partial Y_k}{\partial t} - \frac{\rho^2 D_{km}}{\psi_\infty m_0} \frac{\partial}{\partial t} \left( \frac{\partial Y_k}{\partial \psi} \right) - \frac{1}{\psi_\infty m_0} \frac{\partial}{\partial t} (\rho^2 D_{km}) \frac{\partial Y_k}{\partial \psi} \right\} \Big|_{\psi_L} = 0 \quad (48)$$

The time derivative of  $\rho^2 D_{km}$  can not be evaluated. It is not possible to write it as an explicit function of the  $Y_k$  and  $\partial Y_k / \partial \psi$ . Moreover, since the code uses successive calculation, cross coupling terms between the different species are not allowed. By necessity, the last term in Eq. (48) must be ignored.

It is possible to just use the expression

$$\left\{ \frac{\partial Y_k}{\partial t} - \frac{\rho^2 D_{km}}{\psi_\infty m_0} \frac{\partial}{\partial t} \left( \frac{\partial Y_k}{\partial \psi} \right) \right\} \Big|_{\psi_L} = 0 \quad (49)$$

as the boundary condition. At steady state, the mass flux fractions will approach constants. But because of the transient behavior of  $\rho^2 D_{km}$ ,

the mass flux fractions will converge to incorrect values.

What is required is a correction term that will cause the code to converge to the proper steady state limit. Moreover, it must be of a form that can be used in the code. This can be done by choosing the boundary condition

$$\left\{ \frac{\partial Y_k}{\partial t} - \frac{\rho^2 D_{km}}{\psi_{\infty 0}} \frac{\partial}{\partial t} \left( \frac{\partial Y_k}{\partial \psi} \right) \right\} \Big|_{\psi_L} = \frac{\partial Z}{\partial t}, \quad (50)$$

where

$$\frac{\partial Z}{\partial t} = \frac{Y_{kU} - \epsilon_k|_{\psi_L}}{0.1 t_{\text{FINAL}}}. \quad (51)$$

The rationale is that  $\rho^2 D_{km}$  changes rather slowly. As it changes the mass flux fraction at  $\psi_L$  will change slowly from the proper value. We use a heuristically defined function  $\partial Z/\partial t$  to modify the value of  $\epsilon_k$  until it again equals  $Y_k$ . So as the time integration proceeds, the value of  $\epsilon_k$  at  $\psi_L$  will vary slightly, but will approach the proper steady state limit.

It is necessary to check that the boundary conditions are still consistent with the initial conditions. For the initial profiles defined by Eq. (37) this will be the case. The mass fractions  $Y_k$  are constant near  $\psi_L$ , so  $\partial Y_k/\partial \psi$  at  $\psi_L$  will be zero. Also  $\partial T/\partial \psi$  at  $\psi_L$  is zero. For any transport algorithm all the  $V_k$  will be zero at  $\psi_L$ , and  $\epsilon_k = Y_k$ . So the initial profile is consistent with the boundary conditions.

When the integration is restarted (NSTART = 2), the space derivatives at  $\psi_L$  are normally non-zero. It is necessary to modify the way PDECOL determines the initial values of  $c_k^{(i)}$ . Normally this is done by the subroutine INITAL. It calls the user supplied subroutine UINIT to obtain the values  $u_k(t_0, \psi_j)$ , where the  $\psi_j$ ,  $j = 1, 2, \dots, NC$  are the collocation points. Then by substituting into the expansion (29), it obtains a set of linear algebraic equations of the form

$$u_k(t_0, \psi_j) = \sum_{i=1}^{NC} c_k^{(i)}(t_0) B_i(\psi_j) \quad k = 1, 2, \dots, N, \quad j = 1, 2, \dots, NC. \quad (52)$$

These systems of equations are solved to obtain the initial  $c_k^{(i)}(t_0)$ .

For a burner stabilized flame, the appropriate values of the mass fractions at  $\psi_1 = \psi_L$  are not known but the values of the mass flux fractions  $\epsilon_k$  are. Thus, we use the boundary condition (46), substitute the expansion (29), and obtain

$$Y_{kU} = \sum_{i=1}^{NC} c_k^{(i)}(t_0) \left[ B_i(\psi_1) - \frac{\rho^2 D_{km}}{\psi_\infty} \frac{\partial B_i(\psi_1)}{\partial \psi} \right], \quad k=1,2,\dots,N-1. \quad (53)$$

The values of  $\rho^2 D_{km}$  from the last run of the code are used. These values are also saved in the restart file. So for a burner stabilized flame, the subroutine INITIAL has been changed so that it generates this new set of equations for  $j = 1, k = 1, 2, \dots, N - 1$ .

Note that the above procedure is not strictly necessary. Since we have a correction term  $\partial Z / \partial t$ , the boundary conditions will approach the proper values, even if they are incorrect at the start of the integration. But it is more efficient to begin with the boundary conditions as accurate as possible.

To actually make comparisons with experiments, some of the boundary conditions may have to be further modified. For instance, evidence exists that hydrogen atoms H combine very rapidly on the burner surface to form  $H_2$ .<sup>13</sup> For practical purposes, we consider this to be instantaneous. Then if our first species is H and our second species is  $H_2$ , their boundary conditions are

$$Y_1 = 0 \quad \text{at } \psi_L \quad (54)$$

and

$$\epsilon_1 + \epsilon_2 = Y_{2u} \quad \text{at } \psi_L. \quad (55)$$

These changes can easily be made in UINIT and INITIAL.

As the time integration proceeds, control returns to the routine MAIN at five equally spaced output times. The code calls subroutine FLSP so the integration process can be monitored. There is no need to adjust  $m_0$ , since this is now a predetermined constant.

<sup>13</sup>J. Warnatz, "Calculation of the Structure of Laminar Flat Flames III: Structure of Burner-Stabilized Hydrogen-Oxygen and Hydrogen-Fluorine Flames", Verlag Chemie GmbH, BdB 8/78 E 4018.

## VII. NUMERICAL CONSIDERATIONS

This code has been applied to the  $H_2-O_2-N_2$  system, with nine chemical species. An earlier version of the code has been applied to the ozone system, with three species<sup>14</sup>. Several calculations have been performed for methane-air flames. In each case the code integrates in time until the steady state solution has been reached. What is desired is accurate values for the flame speed and accurate temperature and species profiles. Results are not discussed here, since they are given in the papers referenced above. Instead, we will discuss some of the considerations necessary to obtain accurate results, based on our experiences with the above systems.

The flame speed can be calculated from any of the species profiles using Eq. (20). Normally the time integration proceeds until the flame speeds based on the major species (reactants and products) agree to within a small fraction of a percent. At this point the solution is very close to steady state. The values of the flame speed calculated from the minor species (radicals) are very sensitive, and require more accuracy (both spatial and temporal) in order to achieve the same agreement.

In general, it is easier to obtain an accurate solution for a fast flame than for a slow flame. The slow flame requires many more integration steps before all the oscillations die out and a steady state solution is achieved.

To obtain an accurate solution, both the spatial and temporal accuracy must be sufficient. If the temporal accuracy is low, the calculated flame speeds can oscillate around the correct value. If the spatial accuracy is low, the problem may come very close to convergence, but to the wrong value, since the communication between the collocation points is inadequate. As a consequence of the spatial or temporal accuracy becoming too small, the integration may break down completely.

Spatial accuracy is more important in the flame front, where we have steep gradients. Hence, our algorithms for choosing breakpoints concentrates them in the region to be occupied by the flame front.

The need for spatial and temporal accuracy is connected. Suppose the number of breakpoints is increased, but the temporal error tolerance  $\epsilon$  is not decreased. The calculated solutions will oscillate, and eventually the solution will break up. On the other hand, suppose  $\epsilon$  is decreased but the breakpoint sequence is not changed. We can still obtain an

<sup>14</sup>J.M. Heimerl and T.P. Coffee, "The Detailed Modeling of Premixed, Laminar Steady-State Flames. I. Ozone", Combustion and Flame, Volume 39, pp. 301-315, 1980.

answer, but the integration will take longer, and the accuracy of the solution is not increased. At this stage, the appropriate value of  $\epsilon$  for a given breakpoint sequence is a matter of trial and error.

An additional problem can occur with very slow flames. If the interval of integration is chosen too small, there may be a noticeable gradient in the temperature profile at the cold boundary. Since the gas is entering through the cold boundary quite slowly, this can result in a substantial heat loss through the boundary. This heat loss will slow down the flame. The integration will converge, but the computed flame speed will be too low. This effect can occur for any flame. However, for most flames, a noticeable heat loss is due to a fairly steep temperature gradient at the boundary. It is then obvious that the interval of integration must be increased. But for a slow flame (less than 20 cm/sec) a noticeable error in the flame speed can result (5 to 10 percent), even when the gradient at the cold boundary appears negligible.

Most of the cases we have run have been with  $NINT = 12$ ,  $KORD = 4$ , and  $NCC = 2$  (13 breakpoints, 26 collocation points). We have let  $\epsilon = 10^{-3}$ , and  $SREC = 10^{-6}$ . The number of central intervals  $NCN$  was 4, and the ratio  $FC$  between the longest and shortest intervals was between 4 and 8.

The length of the interval of integration  $L$ , as well as the optimum value for  $FC$ , is a matter of trial and error. However, a reasonable set of breakpoints can be found fairly easily by experimenting with the above scheme, using the constant transport subroutine. The code can then be run with a more realistic transport algorithm, using the restart option.

To make sure that convergence was obtained, the code was also run with  $NINT = 16$  and  $\epsilon = 3 \times 10^{-4}$  for a number of cases. Only negligible differences occurred.

The program was run on the BRL CYBER 76. For hydrogen-oxygen flames, the run time varied from about 30 seconds with the simplest transport subroutine to 5 minutes or more for the most complicated case.

An earlier version of this code was applied to the solution of an unbounded ozone flame<sup>14</sup>. The successive calculation procedure was not implemented at that time. However, since the number of species is so small, the savings in reducing the size of the Jacobian is not very important.

At that time we reported using  $NINT = 57$  to  $NINT = 70$ ,  $KORD = 6$  and  $NCC = 5$ . The number of collocation points was then between 62 and 75. We have now been able to solve this case with equal accuracy using  $NINT = 12$ ,  $KORD = 4$ , and  $NCC = 2$ . Partly this is due to choosing  $NCC = 2$ . The program then generates a special choice of collocation points (Gauss - Legendre quadrature points in each subinterval) which gives increased spatial accuracy. More important is the improved algorithm for choosing breakpoints. By using a little more care, a much smaller set of breakpoints can still reproduce the flame with sufficient accuracy. So the careful choice of breakpoints is probably the most important consideration in running the code efficiently.

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APPENDIX A

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## APPENDIX A

A listing of the computer code follows. The subroutines MAIN, F, UINIT, FLSP, BNDY, BKPT, and RT are written for laminar flame problems. The subroutine F given is for the constant transport assumption. The subroutine RT given is for a  $H_2-O_2-N_2$  flame. There are nine chemical species and thirty reactions.

The rest of the subroutines are from PDECOL. They have been modified as discussed in the text.

After the listing, the job stream and output is given for a typical hydrogen-oxygen flame. The initial unburned gas is 50%  $H_2$  and 50% air, where air is 21%  $O_2$  and 79%  $N_2$ . The nitrogen is considered to be a diluent, and does not react. The required starting information is attached to TAPE11. The main program and the subroutine RT are attached in a compiled form. The integration begins from the initial profiles (37). At the final output time, the flame speed and the temperature and species profiles are given, as well as the corresponding  $x$  values. The restart file is written on TAPE 2, and the output file on TAPE 9. Both files are catalogued for possible future use.

```

*** 03/10/80 SCOME 2.1.5 R H L VER 004 *** 04/13/80 M0106
SYS DEVICES R19/ 4/PF FLS=377K FLL=1750K MKS=300K MXL=1305K MXH=1105H

PHLMM.55 CPU SECOND UPGIN
11.06.56.MFA. 42
11.07.03 00000.003 MFZ.
11.07.03 00000.003 JOR.
11.07.04 00000.026 JOR.
11.07.05 00000.029 MFZ.
11.07.05 00000.030 LOD.
11.07.42 00000.045 USR.
11.07.44 00000.171 USR.
11.07.44 00000.172 LOD.
11.08.53 00001.490 USR.
11.08.54 00001.495 JOR.
11.08.54 00001.499 MFZ.
11.08.54 00001.499 LOD.
11.09.01 00001.914 USR.
11.09.01 00001.941 USR.
11.09.01 00001.942 LOD.
11.09.56 00002.416 USR.
11.09.56 00002.422 JOR.
11.09.57 00002.425 MFZ.
11.09.57 00002.426 LOD.
11.10.11 00002.440 USR.
11.10.35 00002.901 USR.
11.10.35 00002.902 LOD.
11.20.41 00006.645 USR.
11.20.41 00006.647 MFZ.
11.20.41 00006.647 MFZ.
11.20.41 00006.647 MFZ.
11.20.41 00006.648 MFZ.
11.20.41 00006.648 MFZ.
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HKL NOS/HE 1.3 L499 VER 004 03/11/80
-TERMY,STMFZ,TJU,P1. PRINT PLFLSC, CY=5 FOR HRL REPORT.
-ACCOUNT,PU***.
-ATTACH,OLDPL1,PLFLSC1,ID=SHJMMH.
PF254 - CYCLE 5 ATTACHED FROM SH=SYSTEM
-UPDATE(P=OLDPL1,OLD=1)
READING INPUT
UPDATE COMPLETED
-FTN(I=COMPILE,SL,H=0,T)
1.714 CP SECONDS COMPIATION TIME
-ATTACH,OLDPL3,PLH202HTW30N4,LD=SHJMMH.
PF254 - CYCLE 1 ATTACHED FROM SH=SYSTEM
-UPDATE(P=OLDPL3,OLD=1)
READING INPUT
UPDATE COMPLETED
-FTN(I=COMPILE,SL,H=0,T)
.470 CP SECONDS COMPIATION TIME
-ATTACH,OLDPL2,PLFLSC2,ID=SHJMMH.
PF254 - CYCLE 5 ATTACHED FROM SH=SYSTEM
-UPDATE(P=OLDPL2,OLD=1)
READING INPUT
UPDATE COMPLETED
-FTN(I=COMPILE,SL,H=0,T)
3.739 CP SECONDS COMPIATION TIME
JM166 - MAXIMUM USER SCM 47000H WOKDS
JM167 - MAXIMUM USER LCM 7000H WOKDS
JM170 - MAXIMUM JS+IO LCM 61H HUFFERS
RM770 - MAXIMUM ACTIVE FILES 5
RM771 - OPEN/CLOSE CALLS 74
RM772 - DATA TRANSFER CALLS 37.754
RM773 - CONTRUL/POSITIONING CALLS 245
RM774 - RM DATA TRANSFER CALLS 3.348
RM775 - BM CONTRUL/POSITIONING CALLS 342
RM776 - QUEUE MANAGER CALLS 744
RM777 - RECALL CALLS 574
SCM 133.251 KWS
LCM 112.623 KWS
I/O 0.242 HW
RMS 0.172 MWS
USER 3.201 SEC
JOR 6.657 SEC
DIO 1.604.967 KW
SS 8.154 SEC
COST ESTIMATE $1.24
SC050 - 000060 SC/LC SWAP

```

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76/76 OPT=1 ROUNDO=0.0/ 1-ACE

PROGRAM MAIN

```

1      PROGRAM MAIN(INPUT,OUTPUT,TAPE5=INPUT,TAPE3=OUTPUT,TAPE1,
2      * TAPE2,TAPE4,TAPE9,TAPE10,TAPE11)
3      C *****
4      C THIS VERSION USES SUCCESSIVE CALCULATION.
5      C EACH PDE IS INTEGRATED IN TIME INDEPENDENTLY.
6      C *****
7      COMMON/ENDPT/PH0,PH5
8      COMMON/TARP/PRESS,PSM,NPUEM
9      COMMON/TARSM/SMALL
10     COMMON/GEAR0/DTUSED,NJ,NSTEPS,NP,NJ
11     COMMON/GEAR1/DUR(5),GROUNDD,IDUM(4)
12     COMMON/OPTION/NOGAUSS,MAXUER
13     COMMON/IDUNIT/LUIT
14     COMMON/GEARS/HMX
15     COMMON/TARIN/PH2,PH3,PHCTD,PHCT
16     COMMON/START/NSSTART,NBURN,NTRAN,KORD,NSKIP
17     COMMON/TARAH/ASP,ESP,TPN,PHN,IM1,IM2,IM3,IM4,IM5,IM6,IM7,IM8,IM9,IM10,
18     COMMON/TARFM/FM
19     COMMON/TARUM/NDIM1,NDIM2
20     DIMENSION LBINP(7)
21     COMMON/TARPK/PHKPT(3)
22     COMMON/TARLB/LB(21)
23     DIMENSION UPH(20),UUF(20)
24     COMMON/TARCT/RL,CPMA,H0(20),R2D(20),H2DM(20)
25     COMMON/TARNT/CPINV,K2OF(20)
26     COMMON/TARMF/RHUV(1000),ORHUV(1000)
27     COMMON/TARV/PHVAL(401),UN(401),U(20,401),UC(20),UH(20),W(20)
28     COMMON/MAIN/SCICH(25),WORK(12000),IWORK(1750)
29     COMMON/ISTART/IW1,IW2,IW3,IW4,IW5,IW6,IW7,IW8,IW9,IW10,
30     * IW11,IW12,IW13,IW14,IW15,IW16,IW17,IW18
31     COMMON/TARCH/DZM
32     DATA LDIM1,NDIM2/20,401/
33     DATA LDIM1,NDIM2/20,401/
34     DATA IWORK(1),IWORK(2)/12000,750/
35     C *****
36     C THE FOLLOWING INPUT DATA MUST BE SPECIFIED.
37     C DIMENSIONING IN MAIN, F, UINIT, FLSP, PI, HNDRY, HKPT, AND INITIAL
38     C MUST BE CONSISTENT.
39     C SUBROUTINE PT MUST BE WRITTEN.
40     C IF NTHAN = 2, SUBROUTINE F MUST BE WRITTEN.
41     C *****
42     HEAD(5,402) LHINP
43     FORMAT(1A10)
44     WRITE(3,450) LHINP
45     FORMAT(77A10//)
46     READ(5,430) PRESS
47     WRITE(3,47) PRESS
48     FORMAT(175X,10PHPRESSURE = 1PE14.4,2X,4HATM:/)
49     HEAD(11,400) NPDE
50     FORMAT(14)
51     WRITE(3,405) NPDE
52     FORMAT(10X,6HNPDE = 14/)
53     WRITE(3,455)
54     FORMAT(1726X,1HW,11X,2HJC,10X,2H4H,10X,2HMO,10X,3HRTD/)
55     DO 425 K=1,NPDE
56     HEAD(11,410) LH(K) = (K), UC(K), UH(K), H0(K), K2D(K)
57     FORMAT(10X,1P5E14.4)
58     C *****

```



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PROGRAM MAIN 76/76 OPT=1 ROUNDO=0.0/ 1-ACE

```

115 MF=22
    PSK=PRESS/R2.05
    TMSPH=TMH/PHN
    TMSPH2=TMSPH/PHN
    TMSIP=TMN/TPN
    UZM=10.0/TFINAL
    DO 9 K=1,NPDE
      R20(K)=R20(K)/PHN
    9 ALL R20 = RH*RH*DM/PHN.
    C R20 IS THE VALUE AT THE LEFT BOUNDARY.
    C R20M IS THE VALUE AT THE RIGHT END OF AN INTERVAL USED FOR
    C COMPUTING THE FLAME SPEED.
    C FUM NTHAN = 1, THEY ARE THE SAME.
    NCPTS=(KORD-NCC)*NINT*NCC
    DO 602 K=1,NPDE
      R20M(K)=R20(K)
    602
    C FIND THE APPROPRIATE NORMALIZED VALUES FOR USE IN F (INTRAN=1).
    CPINV=TMSTP/CPMX
    DO 605 K=1,NPDE
      R20F(K)=R20(K)*TMSPH
      RL=HL*TMSPH2/CPMX
      NPUEM=NPDE-1
      NPUEP=NPUE+1
      NCPTS=(KORD-NCC)*NINT*NCC
      YSSW=0.0
      YN=1.0
      DO 5 K=1,NPDEM
        YN=YN-UC(K)
        YSSM=YSSM+UC(K)/W(K)
        YSSM=YSSM+YN/W(NPUE)
        T=UC(NPDE)*TPN
        MHU=PSR/(1+YSSM)
        WRITE(3,*)RHO
        FORMAT(5X,17HINITIAL DENSITY =,1PE12.4/)
        NPTS=NINT*1
        T=TFINAL*1.0E-10
        TOUT=0.0
    5 C MMX IS THE LARGEST TIME STEP ALLOWED IN THE INTEGRATION.
        HMX=TFINAL/20.0
        INDEX=1
    C DEFINE THE CENTER OF THE FLAME.
        KCEN=NPDE
        VCEN=UC(KCEN)*0.4*(UH(KCEN)-UC(1+KCN))
        WRITE(3,573)KCEN,VCEN
    573
    C INITIAL VALUE FOR THE SPEED OF THE ORIGIN (MASS FLOW).
        IF (INSTART.EQ.2) REAU(1,255)=SPEED0
        IF (INSTART.EQ.2) WMIT(3,552)=SPEED0
        IF (INSTART.EQ.1) SPEED0=0.0
        ASP=SPEED0
        IF (INHUM.EQ.0.0) GO TO 550
        FM=RHU*FLUSP
        WRITE(3,67)FM,FLUSP
        SPEED0=FM*TMSPH
        SPNI=SPEED0
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116 MAIN
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120 MAIN
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122 MAIN
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THACE

PROGRAM MAIN

76/76

OPT=1 KOUND=

```

175      ASP=SPEEDO
176      HSP=0.0
177      CONTINUE
178      FORMAT(5X,8HSPEDUO =,1PE12.4/)
179      FORMAT(1PE14.6)
180      FT=SECOND(CP)
181      C CALL INTEGRATOR AND WRITE OUTPUT.
182      C CALL PDECOL(TO,TOUT,DT,PHAKPT,EPS,NINT,KOMU,NCC,NPDF,MF,
183      * INDEX,WORK,IWORK,SECT)
184      IF (INDEX.NE.0) GO TO 70
185      GT=SECOND(CP)
186      HT=GT-FT
187      IF (HT.GT.7MAX) TFINAL=TOUT
188      WRITE(3,30) TOUT,DT,USED,NSTEPS
189      FORMAT(//10X,3HT =,1PE12.4,4X,4.HT =,1PE12.4,4X,
190      * 13HTOTAL STEPS =,1M/)
191      WRITE(3,32) NF,NJ
192      FORMAT(//10X,6HNF =,1M,5X,4HNJ =,1M/)
193      WRITE(3,45) HT
194      FORMAT(//10X,10HHT TIME =,1PE12.4/)
195      CALL VALUES(PHVAL,U,SECTCH,NDIM,IM2,NVPTS,0,WORK)
196      DO 25 K=1,NVPTS
197      UN(K)=1.0
198      DO 25 J=1,NPDEM
199      UN(K)=UN(K)-U(J,K)
200      IF (TOUT.LT.TPRINT) GO TO 65
201      DO 35 K=1,NPDEM
202      WRITE(3,37) LB(K)
203      FORMAT(//10X,A10/)
204      WRITE(3,39) U(K,1),I=1,NVPTS,NSKIP)
205      FORMAT(//10X,12.4)
206      CONTINUE
207      WRITE(3,37) LB(NPDE)
208      WRITE(3,39) U(NPDE,1),I=1,NVPTS,NSKIP)
209      CONTINUE
210      CALL FLSP(NPDE,TOUT,TPRINT,TFINAL,NVPTS,NLPTS,
211      * PHO,KCEN,FSP,UPH,UOF)
212      FORMAT(5X,8HSPEDUO =,1PE12.4,6X,4HUS =,1PE12.4/)
213      IF (TOUT.GE.TPRINT) TPRINT=TPRINT+IFINAL/5.0
214      IF (INURN.EQ.0) GO TO 100
215      DO 610 J=1,NPDEM
216      UOF(J)=U(J,1)-UPH(J)*P2D(J)/FM
217      UOF(NPDE)=1.0
218      DO 612 J=1,NPDEM
219      UOF(NPDE)=UOF(NPDE)-UOF(J)
220      WRITE(3,615) UOF(L),L=1,NPDE)
221      FORMAT(//2X,29HSPEDUO MASS FLUX FRACTIONS =,1PE12.4/)
222      C FIND THE FORMULA FOR ADJUSTING THE ORIGIN SPEED AND CENTERING
223      C THE FLAME.
224      DO 110 K=1,NVPTS
225      KP=K+1
226      UMAX=AMAX1(U(KCEN,K),U(KCEN+KP,K))
227      UMIN=AMIN1(U(KCEN,K),U(KCEN+KP,K))
228      IF (UMAX.GT.VCEN+AND,UMIN.LT.VCEN) KS=K
229      IF (UMAX.GT.VCEN+AND,UMIN.LT.VCEN) GO TO 115

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04/15/80 11.07.44

FIN 4.8.4.94

PROGRAM MAIN 76/76 OPT=1 MOUND=0.0/ 1-ACE

```

230      KP=KS*1
231      PC=(VCEN-U(KCEN*KS))/(U(KCEN*KP)-U(KCEN*KS))
232      PHNE=PHVAL(KS)*PC*(PHVAL(KP)-PHVAL(KS))
233      IF (NHURN.EQ.1) GO TO 560
234      IF (TOUT.GT.0.0) GO TO 150
235      FM=PHO*FSP
236      IF (INSTART.EQ.1) SPEEDU=FM*TMSPH
237      WRITE(3,447)FM,FSP
238      FORMAT(1/5X,4HMO =,1PE12.4,6X,6HLSP =,1PE12.4/)
239      US=0.0
240      IF (SPEEDO.GT.0.0) SPEEDO=0.0
241      WRITE(3,443)SPEEDO,US
242      TINC=TFINAL/100.0
243      TOLD=TOUT
244      TOUT=TINC
245      DPH=PHNEW-PHCT
246      UT=TOUT-TOLD
247      PHOLD=PHNEW-SPEEDO*UT
248      HSP=0.0
249      ASP=SPEEDO
250      SPEEDN=ASP*HSP*TOUT
251      SPN1=SPEEDN
252      GO TO 20
253      SPEEDO=SPEEDN
254      SPN1=SPN1
255      SPN1=(PHNEW-PHOLD)/(TOUT-TOLD)
256      FM=-SPN1/TMSPH
257      FSP=FM/RHO
258      WRITE(3,47)FM,FSP,PHNEW
259      FORMAT(1/5X,4HMO =,1PE12.4,6X,6HLSP =,1PE12.4,6X,
260      * 7HPHNEW =,1PE12.4/)
261      US=SPN1-SPN1
262      TOLD=TOUT
263      C ESTIMATE CHANGE IN FLAME SPEED PER UNIT TIME.
264      USG=ABS(US)/TINC
265      C CHOOSE A NEW TINC SO THAT THE ESTIMATED DRIFT OF THE FLAME
266      C FROM THE CENTER IS VERY SMALL.
267      DPHMX=(PH3-PH2)/50.0
268      TINC=SQRT(DPHMX/USG)
269      TINC=MAX1(TINC,TFINAL/100.0)
270      TINC=MIN1(TINC,TFINAL/4.0)
271      TOUT=TOUT+TINC
272      IF (TOUT.GT.TFINAL) TOUT=TFINAL
273      IF (TOLD.EQ.TFINAL) GO TO 201
274      UT=TOUT-TOLD
275      DPH=PHNEW-PHCT
276      SPN2=2.0*SPN1-SPEEDO*2.0*DPH/UT
277      SPEEDN=0.5*(SPN1+SPN2)
278      C IF DPH IS SMALL ENOUGH, DON'T HURRY TO RECENTRE THE FLAME.
279      IF (ABS(DPH).LT.0.01*(PH5-PH0)) SPEEDN=SPN1
280      C FLAME SPEED CANNOT BE NEGATIVE.
281      IF (SPEEDN.GT.0.0) SPEEDN=0.0
282      HSP=(SPEEDN-SPEEDO)/(TOUT-TOLD)
283      ASP=SPEEDO-TOLD*HSP
284      US=SPEEDN-SPEEDO
285      WRITE(3,43)SPN1,SPN2,SPEEDN,US
286      FORMAT(1/5X,6HSPN1 =,1PE12.4,6X,6HSPN2 =,1PE12.4,6X,

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11.07.44

04/15/HU

FTN 4.8.49H

76/76 OPT=1 KOUND=0.0/ TRACE

PROGRAM MAIN

```

*  HXSPEEDN =,IPE12.4,6X,4HDS =,IPE12.4/
PHOLD=PHNEW-DT*(ASP+SP*(TOUT+IULU)*0.5)
GO TO 570
560 TOUT=TOUT+TFINAL/4.0
570 CONTINUE
IF(TOUT.LE.TFINAL)GO TO 20
201 CONTINUE
C  CREATE DATA FILE FOR RESTART
WRITE(3,220)SPH1
WRITE(2,220)SPH1
WRITE(3,220)PHNEW
WRITE(2,220)PHNEW
WRITE(3,200)
200 FORMAT(/10X,14HCOLLOCATION POINTS/)
WRITE(2,210)NCPTS
WRITE(3,210)NCPTS
210 FORMAT(JA)
IEND=IW3+NCPTS-1
WRITE(2,220)(WORK(L),L=IW3,IEND)
WRITE(3,220)(WORK(L),L=IW3,IEND)
CALL VALUES(WORK(IW3),U,SCICH,N,IW1,NULIM2,NCPTS,0,WORK)
DO 250 K=1,NPDE
WRITE(2,220)(UIK,L),L=1,NCPTS)
WRITE(3,220)(UIK,L),L=1,NCPTS)
250 CONTINUE
WRITE(2,220)(R2U(L),L=1,NPDE)
WRITE(3,220)(R2U(L),L=1,NPDE)
220 FORMAT(1PE14.6)
500 CONTINUE
70 WRITE(3,RO)INDEX
80 FORMAT(/10X,7HINDEX =,I3)
STOP
END

```

```

1      SUBROUTINE F (TIME,PH,U,UPH,UPH2,FVAL,NPDE,NPUE,IC,KSKT,KSKR)
2      DIMENSION U(NPDE),UPH(NPDE),UPH2(NPDE)
3      COMMON/TABAR/ASP,ASP,TPH,PHN,IM,IMSPH,IMSPH2,IMSTP,TPENT
4      COMMON/TAMP/PHESP,PHSP,NPDEM
5      DIMENSION H(20),R(20)
6      COMMON/TARCT/RL,CPH,H(20),H2U(20),H2UM(20)
7      COMMON/TABNT/CPINV,H2DF(20)
8      COMMON/TABRY/T,HMY1,Y2,Y3,Y4,Y5,Y6,Y7,Y8,Y9,Y10
9      *****
10     C AT EACH CALL THE TIME RATE OF CHANGE FOR ONE PDE IS RETURNED IN FVAL.
11     C *****
12     YN=1.0
13     DO 10 K=1,NPDEM
14     YN=YN-U(K)
15     CALL RT(U,YN,H,NPDE,NPUE,IC,KSKR)
16     SP=ASP+HSP*TIME
17     IF (KPDE.EQ.NPDE) GO TO 50
18     FVAL=SP*UPH(NPDE)+H2DF(KPDE)*UPH2(NPDE)+TMN*R(KPDE)
19     RETURN
20
21     DT=TPN*U(NPDE)-TPENT
22     DO 20 K=1,NPDE
23     H(K)=H(K)+CPMX*DT
24     KTEMP=0.0
25     DO 25 K=1,NPDE
26     KTEMP=KTEMP-R(K)+H(K)
27     FVAL=SP*UPH(NPDE)+HL*UPH2(NPDE)+KTEMP*CPINV
28     RETURN
29     END

```

04/15/80 11.07.44

FIN 4.R.4.4H

I=ACE

76/76 OPT=1 KOUND=--\*/

SUBROUTINE UNIT

```

1  SUBROUTINE UNIT(PH,U,NPDE,KPDE)
COMMON/TA4IN/PH2,PH3,PHCTO,PHCT
COMMON/ENIPT/PHU,PHS
COMMON/TARAH/ASP,HSP,TPH,PHN,IM,TMSPH,TMSPH2,TMSTP,TPENT
COMMON/START/NS,(HURR,ENTRAN,KORI,NSKIP
COMMON/TARV/PHVAL(401),PHS(401),UP(20*401),UC(20),UH(20),W(20)
C PIS2=PI / 2.0.
C DATA PIS2/1.5707963/
C *****
C DETERMINE THE INITIAL STARTING PROFILE.
C X WILL BE A COLLOCATION POINT.
C U STORES THE CORRESPONDING FUNCTION VALUES.
C *****
C IF (NS.GT.1) GO TO 50
C DETERMINE AN INITIAL GUESSED PROFILE FROM VALUES AT THE
C HURNED AND UNHURNED ENDS.
C IF (PH.GT.PH2) GO TO 5
C U=UC(KPDE)
C RETURN
5 IF (PH.GT.PH3) GO TO 10
C PHU=(PH-PH2)/(PH3-PH2)
C PHM=PIS2*(PHD**2)
C UU=UH(KPDE)-UC(KPDE)
C U=UC(KPDE)+UD*(TSIN(PHM)**2)
C RETURN
10 U=UH(KPDE)
C RETURN
50 IF (NS.GT.2) GO TO 100
C READ THE COLLOCATION VALUES AND THE CORRESPONDING FUNCTION
C VALUES FROM A PREVIOUS RUN.
C HEAD(1:65)PHCTO
C HEAD(1:55)NPS
C FORMAT(18)
55 NM=NPS-1
C HEAD(1:65) (PHS(L),L=1,NPS)
C DO 60 K=1,NPDE
C HEAD(1:65) (UP(K,L),L=1,NPS)
C WE DO NOT WANT ANY NEGATIVE CONCENTRATIONS.
C DO 58 L=2,NM
C IF (UP(K,L).GE.0.0) GO TO 54
C LM=L-1
C LP=L+1
C UP(K,L)=0.5*(UP(K,LM)+UP(K,LP))
54 CONTINUE
60 CONTINUE
65 FORMAT(1PRE14.6)
C IF (HURN.EQ.1) GO TO 75
C CENTER THE FLAME.
C DO 70 K=1,NPS
C PHS(K)=PHS(K)+PHCT-PHCTO
70 CONTINUE
75 CONTINUE
C PHAL=PHS(2)-PHU
C PHAR=PHS-PHS(NPS)
C IS=3
C U=UC(KPDE)
C RETURN

```

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FTN 4.8.498

SUBROUTINE UINIT 76/76 OPT=1 KOUNB=0.0/ 1-ACE

C DETERMINE THE NEW VALUES BY INTERPOLATION.

```

100 K=2
   IF (PH.GT.PH0) GO TO 102
   U=UC(KPDE)
   RETURN
102 IF (PH.GT.PHS(2)) GO TO 104
   P=(PH-PH0)/PHAL
   U=UC(KPDE)+P*(UP(KPDE,2)-UC(KPDE))
   RETURN
104 IF (PH.LT.PHS(NPS)) GO TO 108
   IF (PHXR.NE.0.0) P=(PH-PHS(NPS))/PHXR
   IF (PHXR.EQ.0.0) P=0.0
   U=UP(KPDE,NPS)+P*(UB(KPDE)-UP(KPDE,NPS))
   RETURN
108 CONTINUE
110 IF (PH.LE.PHS(K)) GO TO 120
   K=K+1
   GO TO 110
120 K=K-1
   P=(PHS(K)-PH)/(PHS(K)-PHS(KM))
   U=UP(KPDE,K)+P*(UP(KPDE,K)-UP(KPDE,KM))
   RETURN
END

```



```

50      DO 60 J=1,NPDE
60      RINT1(J)=RINT(J)
100     CONTINUE
110     DO 110 J=1,NPDE
110     FLS (J)=PHNPHI(J)
C      FIND APPROPRIATE VALUES AT NMID.
C      CALL VALUES(PHVAL(NMID),URT,SCIC,MNDIMI,1.1,2.0,WORK)
C      IF (INTRAN.EQ.1) GO TO 114
C      IF THE LONG VERSION OF F IS USED, WE NEED TO FIND OUT WHAT THE VALUE
C      OF W2U IS AT THE APPROPRIATE POINT IN THE FLAME.
C      IC=NCPTS+1
C      CALL FITOUT(PHVAL(NMID),URT(1.1),URT(1.2),URT(1.3),FVAL,
C      * NPDE,1,IC,1,1)
C      IST=NCPTS+NPDE
C      DO 115 J=1,NPDEM
115     W2UM(J)=-RHUV(IST+J)/URT(J,2)
118     DO 120 J=1,NPDEM
120     U(J)=W2UM(J)*URT(J,2) -W2U(J)*ULT(J,2)
C      THE SUM OF THE DIFFUSION TERMS MUST ADD UP TO ZERO.
C      U(NPDE)=0.0
C      DO 124 J=1,NPDEM
124     U(NPDE)=U(NPDE)-U(J)
C      CONTINUE
130     FLS (J)=FLS (J)*U(J)
135     CONTINUE
C      DIFF=U(NMID)-U(1)
C      IF (ABS(DIFF).LE.SMALL) FLS(NPDE)=0.0
C      IF (ABS(DIFF).GT.SMALL) FLS(NPDE)=FLS(NPDE)/(RH0*DIFF)
C      DO 150 J=1,NPDEM
150     DIFF=U(J,NMID)-U(J,1)
C      IF (ABS(DIFF).LE.SMALL) FLS(J)=0.0
C      IF (ABS(DIFF).GT.SMALL) FLS(J)=FLS(J)/(RH0*DIFF)
C      CONTINUE
152     WHITE(3,152)=(FLS (L)*L+1,NPDE)
155     FORMAT(1/2X,13HFLAME SPEED =,1P10E12.4,/)
C      FSP=FLS(NPDEM)
C      FIND THE X VALUES USING THE TRAPEZOIDAL RULE.
C      W(1)=0.0
C      X(1)=0.0
C      DO 200 K=1,NVPTS
200     T=U(NPDE,K)*TPN
C      YSSM=U(N,K)/W(NPDE)
C      DO 160 J=1,NPDEM
160     YSSM=YSSM+U(J,K)/W(J)
C      W(N)=PSH/IT*YSSM
C      WINT(1)=1.0/RH
C      IF (K.EQ.1) GO TO 140
C      KM=K-1
C      DPM=PHVAL(K)-PHVAL(KM)
C      W(1)=W(1)+0.5*(WINT(1)+RINT1(1))*DPM
C      X(K)=PHNPHI(1)
C      WINT(1)=RINT1(1)
C      CONTINUE
C      IF (TOUT-L1,TPWINT) GO TO 250
C      WHITE(3,201)

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115 201 FORMAT(10X,7H 1st CNV)
      WRITE(3,202)(X(L),L=1,NVPTS,NSKIP)
202 FORMAT(1P10E12.4)
250 CONTINUE
C COMPUTE THE FLAME THICKNESS.
      THMAX=MAX1(U(INPDE,NVPTS),U(INPDE,1))
      THMIN=AMIN1(U(INPDE,NVPTS),U(INPDE,1))
      DT=0.1*(THMAX-THMIN)
      TH=THMAX-DT
      TL=THMIN+DT
      DO 310 K=1,NVPTS
        KP=K+1
        IF (TL.GT.U(INPDE,K).AND.U(LT,U(INPDE,KP))){KLE=K
          IF (TH.GT.U(INPDE,K).AND.TH.LT.U(INPDE,KP))KHE=K
          CONTINUE
          KLP=KL+1
          KHP=KH+1
          PL=(U(INPDE,KL)-TL)/(U(INPDE,KL)-U(1,KLP))
          PH=(U(INPDE,KH)-TH)/(U(INPDE,KH)-U(1,KHP))
          PHL=PHVAL(KL)+PL*(PHVAL(KLP)-PHVAL(KL))
          PHH=PHVAL(KH)+PH*(PHVAL(KHP)-PHVAL(KH))
          XLW=X(KL)+PL*(X(KLP)-X(KL))
          XMH=X(KH)+PH*(X(KHP)-X(KH))
          FTH=ABS(XLW-XH)
          WRITE(3,314)PHL,PHH
          PHCEN=0.5*(PHL+PHH)
          PHF=AMIN1(PHL,PHH)
          PHFF=AMAX1(PHL,PHH)
          FORMAT(10X,23HFLAME FRONT FROM PH[*]=1PE12.4,2X,
            * 6HNO PH[*]=1PE12.4/)
          WRITE(3,318)XLW,XH
          FORMAT(10X,20HFLAME FRONT FROM X =1PE12.4,2X,
            * 6HNO X =1PE12.4/)
          WRITE(3,330)FTH
          FORMAT(10X,17HFLAME THICKNESS =1PE12.4,2X,2HCM/)
          IF (TOUT.LT.TFINAL)RETURN
C CHARGE OUTPUT FILE.
          WRITE(9,1208)NPDE,NVPTS
          WRITE(3,1208)NPDE,NVPTS
          FORMAT(514)
          DO 1215 K=1,NPDE
            WRITE(9,1212)CHK(X(K),HD(K),K/1(K)
            WRITE(3,1212)CHK(X(K),HD(K),K/1(K)
          FORMAT(10,1P3E16.4)
          CONTINUE
          LHT=10HT
          WRITE(9,1212)LHT
          WRITE(3,1212)LHT
          WRITE(9,1210)TOUT
          WRITE(3,1210)TOUT
          WRITE(9,1210)ASP,HSP,TPN,PHN,IM,
            * 6HNO ASP,HSP,TPN,PHN,IM,
          WRITE(9,1210)PHF,SS,PSP
          WRITE(3,1210)CPMA,KL,TPENT
          WRITE(9,1210)CPMA,KL,TPENT
          WRITE(3,1210)PHF,SS,PSP
          FORMAT(1P4F16.4)
1210 1210
      FLSP
116 FLSP
117 FLSP
118 FLSP
119 FLSP
120 FLSP
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122 FLSP
123 FLSP
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172 FLSP

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FTN 4.8.498

FACE

76/76 OPT=1 MOUNU=0-0/

SUBROUTINE FLSP

FLSP 173  
FLSP 174  
FLSP 175  
FLSP 176  
FLSP 177  
FLSP 178  
FLSP 179

DO 1200 K=1,NPTS  
CALL VALUES(PHVAL(K),URT,SCICH,K,IM1,1,1,2,MUNK)  
WRITE (9,1210) PHVAL(K),X(K), (URT(L,1),L=1,NPUE),  
• (URT(L,2),L=1,NPUE), (URT(L,3),L=1,NPUE)  
1200 CONTINUE  
RETURN  
END

175

```

1      SUBROUTINE BNDHY(TIME,PH,U,UPH,DU,DUUPH,DZDT,NPDE)
2      DIMENSION U(NPDE),UPH(NPDE),DZDT(NPDE)
3      DIMENSION DUUPH(NPDE),DUUPH(NPDE)
4      COMMON/ENDPT/PHL,PHR
5      COMMON/TARP/PRESS,PSH,NPDEM
6      C FOR BOUNDARY CONDITIONS, SEE HJMSHFLDER, CJRTISS,
7      C AND HIRD, PP. 762,763.
8      COMMON/TARCT/RL,CPMA,NO(20),R2D(10),RZDM(20)
9      COMMON/TARHF/RHUV(1000),DRHUV(1000)
10     COMMON/TARSM/SMALL
11     COMMON/TARFM/FM
12     COMMON/START/NSTART,NHURN,NTRAN,KORO
13     COMMON/TARCM/DZM
14     COMMON/TARV/PHVAL(401),UN(401),U(20),UH(20),W(20)
15     DO 5 I=1,NPDE
16     DZDT(I)=0.0
17     UH(I)=0.0
18     UHUPH(I)=0.0
19     IF (PH.GT.PHL)GO TO 50
20     IF (NHURN.EQ.1)GO TO 7
21     DO 10 J=1,NPDE
22     UHUPH(J)=1.0
23     RETURN
24     C HURNER STABILIZED FLAME. WE NEED TO FIND THE MASS FLUX
25     C FRACTIONS AT THE COLD BOUNDARY.
26     IF (NTRAN.EQ.1)GO TO 12
27     C FIND R2D AT THE LEFT BOUNDARY.
28     DO 8 J=1,NPDEM
29     IF (ABS(UPH(J)).LT.SMALL)UPH(J)=SMALL
30     R2D(J)=-RHUV(J)/UPH(J)
31     CONTINUE
32     DUUPH(NPDE)=1.0
33     DO 20 J=1,NPDEM
34     DUUPH(J)=1.0
35     DUUPH(J)=1.0
36     DUUPH(J)=R2D(J)/FM
37     C IF THE MASS FLUX FRACTION EP DOES NOT HAVE THE CORRECT VALUE,
38     C WE USE DZDT TO MOVE IT TOWARDS Y COLD.
39     C DZM IS CHOSEN SO THAT, IF R2D DOES NOT CHANGE, EP WOULD BE EQUAL
40     C Y COLD AFTER A TIME INTERVAL TFINAL/10.0.
41     DO 25 J=1,NPDEM
42     EP=U(J)-R2D(J)*UPH(J)/FM
43     DZDT(J)=(UC(J)-EP)*DZM
44     RETURN
45     DO 30 J=1,NPDE
46     DUUPH(J)=1.0
47     RETURN
48     END

```

04/14/80 11.07.66

FTN 4.4.494

76/76 OPT=1 MOUN=0.0/ 1PAC

SUBROUTINE HKPT

```

1      SUBROUTINE HKPT(NINT,NCN,FC,NVP,IS)
COMMON/ENDPT/PH0,PH5
COMMON/TARXK/PHKPT(3)
COMMON/TARV/PHVAL(40),UN(40),...((20,40),UC(20),UH(20),W(20))
COMMON/TARIN/PH2,PH3,PHCTO,PHCT
COMMON/START/START,SHURN,NTRAN,KORD,NSK,JP
C      COMPUTE THE BREAKPOINTS.
C      IF (SHURN.EQ.1)GO TO 650
C      UNKNOWN FLAME.
PHCT=PH0*0.5*(PH5-PH0)
PH2=PH0*0.24*(PH5-PH0)
PH3=PH0*0.44*(PH5-PH0)
ALP=2.0*ALOG10((FC)/FLOAT(NINT-NCN))
ALP=10.0*ALP
I=(NINT-NCN)/2
P=(PH5-PH0)/(FLOAT(NCN)*2.0*ALP*(ALP**[-1.0]/(ALP-1.0)))
PHKPT(1)=PH0
DP=FC*P
NU=(NINT-NCN)/2
DO 11 K=1,NU
  PHKPT(KP)=PHKPT(K)*DP
  DP=DP/ALP
  NL=NU*1
  NU=NU*NCN
  DP=P
  DO 12 K=NL,NU
    KP=K*1
    PHKPT(KP)=PHKPT(K)*DP
    NL=NU*1
    NU=NU*1
    DO 13 K=NL,NU
      KP=K*1
      DP=DP*ALP
      PHKPT(KP)=PHKPT(K)*DP
      PHKPT(NINT*1)=PH5
    GO TO 660
C      MURNER STABILIZED FLAME.
650  PH2=PH0*0.1*(PH5-PH0)
      PH3=PH0*0.3*(PH5-PH0)
      ALP=ALOG10((FC)/FLOAT(NINT-NCN))
      ALP=10.0*ALP
      I=NINT-NCN
      P=(PH5-PH0)/(FLOAT(NCN)*ALP*(ALP**[-1.0]/(ALP-1.0)))
      PHKPT(1)=PH0
      DO 653 K=1,NCN
        KP=K*1
        PHKPT(KP)=PHKPT(K)*P
        NL=NCN*1
        NU=NINT
        DO 657 K=NL,NU
          KP=K*1
          P=P*ALP
          PHKPT(KP)=PHKPT(K)*P
          CONTINUE
        NU=NU*1
        NITE(3,39)=(PHKPT(1),L=1,NU)

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04/15/80 11.07.44

FIN 4.4.44

SUBROUTINE HKPT 76/76 OPT=1 KOUND=\*\*/ 1\*ACE

```

C  DEFINE THE EVALUATION POINTS.
NSP=R
NVPTS=NSP*NINT+1
DO 17 K=1,NVPTS
  KV=1+(K-1)*NSP
  PHVAL(KV)=PHHKPT(K)
  NSVP=NSP+1
  DO 19 K=NSVP,NVPTS,NSP
    KM=K-NSP
    UP=(PHVAL(K)-PHVAL(KM))/FLOAT(NSP)
    DO 19 J=2,NSP
      KT=KM+J-1
      PHVAL(KT)=PHVAL(KM)+UP*FLOAT(J-1)
    CONTINUE
  NSKIP=NINT/10
  IF (NSKIP.LT.1) NSKIP=1
  WRITE(3,99) (PHVAL(I),I=1,NVPTS,NSKIP)
  FORMAT(1P)E12.4)
  RETURN
END

```

HKPT 59  
 HKPT 60  
 HKPT 61  
 HKPT 62  
 HKPT 63  
 HKPT 64  
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04/15/80 11.09.01

FTN 4.8.498

76/76 OPT=1 MOUN=0.0/ 1-ACE

SUBROUTINE RT

CARD NR. SEVERITY DETAILS DIAGNOSIS OF PROBLEM

14	I	U	ARRAY REFERENCE OUTSIDE DIMENSION HOUNUS.
15	I	U	ARRAY REFERENCE OUTSIDE DIMENSION HOUNUS.
16	I	U	ARRAY REFERENCE OUTSIDE DIMENSION HOUNUS.
17	I	U	ARRAY REFERENCE OUTSIDE DIMENSION HOUNUS.
18	I	U	ARRAY REFERENCE OUTSIDE DIMENSION HOUNUS.
19	I	U	ARRAY REFERENCE OUTSIDE DIMENSION HOUNUS.
20	I	U	ARRAY REFERENCE OUTSIDE DIMENSION HOUNUS.
91	I	K	ARRAY REFERENCE OUTSIDE DIMENSION HOUNUS.
94	I	K	ARRAY REFERENCE OUTSIDE DIMENSION HOUNUS.
97	I	K	ARRAY REFERENCE OUTSIDE DIMENSION HOUNUS.
100	I	K	ARRAY REFERENCE OUTSIDE DIMENSION HOUNUS.
102	I	K	ARRAY REFERENCE OUTSIDE DIMENSION HOUNUS.
104	I	K	ARRAY REFERENCE OUTSIDE DIMENSION HOUNUS.
105	I	K	ARRAY REFERENCE OUTSIDE DIMENSION HOUNUS.
108	I	K	ARRAY REFERENCE OUTSIDE DIMENSION HOUNUS.

11.10.37

04/15/80

FTH 4.8.49H

74/76 OPT=1 MOUNDED=0/ TRACE

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1 SUBROUTINE PDECOL(T0,TOUT,DT,XM,PT,EP,S,NINT,KORD,NCC,NPDE,MH,
5 INDEX,WORK,IWORK,SHRC)
C-----
C THIS IS THE SEPT 23, 1977 VERSION OF PDECOL.
C THIS PACKAGE WAS CONSTRUCTED SO AS TO CONFORM TO AS MANY ANSI-FORTHAN
C RULES AS WAS CONVENIENTLY POSSIBLE. THE FORTHAN USED VIOLATES ANSI
C STANDARDS IN THE TWO WAYS LISTED BELOW....
C 1. SUBSCRIPTS OF THE GENERAL FORM C(V1 + V2 * V3 ARE USED
C (POSSIBLY IN A PERMUTED ORDER). WHERE C IS AN INTEGER CONSTANT
C AND V1, V2, AND V3 ARE INTEGER VARIABLES.
C 2. ARRAY NAMES APPEAR SINGLY IN UNIA STATEMENTS IN THE ROUTINES
C HSPLVN AND COSET.
C-----
C-----
C PDECOL IS THE DRIVER ROUTINE FOR A SOPHISTICATED PACKAGE OF
C SUBROUTINES WHICH IS DESIGNED TO SOLVE THE GENERAL SYSTEM OF
C NPDE NONLINEAR PARTIAL DIFFERENTIAL EQUATIONS OF AT MOST SECOND
C ORDER ON THE INTERVAL (XLEFT,XRIGHT) FOR T,GT, TO WHICH IS OF THE
C FORM....
C DDZ/DT = F( T, X, U, UX, UXX )
C WHERE
C U = ( U(1), U(2), ... , U(NPDE) )
C UX = ( UX(1), UX(2), ... , UX(NPDE) )
C UXX = ( UXX(1), UXX(2), ... , UXX(NPDE) ) .
C EACH U(K) IS A FUNCTION OF THE SCALAR QUANTITIES T AND X.
C UX(K) REPRESENTS THE FIRST PARTIAL DERIVATIVE OF U(K) WITH RESPECT
C TO THE VARIABLE X. UXX(K) REPRESENTS THE SECOND PARTIAL DERIVATIVE
C OF U(K) WITH RESPECT TO THE VARIABLE X. AND DDZ/DT IS THE VECTOR OF
C PARTIAL DERIVATIVES OF U WITH RESPECT TO THE TIME VARIABLE T.
C F REPRESENTS AN ARBITRARY VECTOR VALUED FUNCTION WHOSE NPDE
C COMPONENTS DEFINE THE RESPECTIVE PARTIAL DIFFERENTIAL EQUATIONS OF
C THE PDE SYSTEM. SEE SUBROUTINE F OF DESCRIPTION BELOW.
C THE PROGRAM IS MODIFIED TO SOLVE THE SYSTEM USING
C SUCCESSIVE CALCULATION.
C THAT IS, THE PDE'S ARE UNCOUPLED AND SOLVED IN ORDER.
C THE EQUATIONS THAT VARY THE MOST (RADICALS) SHOULD BE LISTED FIRST.
C BOUNDARY CONDITIONS
C DEPENDING ON THE TYPE OF PDE(S). 0, 1, OR 2 BOUNDARY CONDITIONS
C ARE PREPARED FOR EACH PDE IN THE SYSTEM. THESE ARE IMPOSED AT XLEFT
C AND/OR XRIGHT AND EACH MUST BE OF THE FORM....
C 4(U,UX) = Z(T)

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C WHERE H AND Z ARE ARBITRARY VECTOR VALUED FUNCTIONS WITH
C NPDE COMPONENTS AND U, UX, AND T ARE AS ABOVE. THESE BOUNDARY
C CONDITIONS MUST BE CONSISTENT WITH THE INITIAL CONDITIONS WHICH ARE
C DESCRIBED NEXT.
C INITIAL CONDITIONS
C EACH SOLUTION COMPONENT U(K) IS ASSUMED TO BE A KNOWN (USER
C PROVIDED) FUNCTION OF X AT THE INITIAL TIME T = T0. THE
C INITIAL CONDITION FUNCTIONS MUST BE CONSISTENT WITH THE BOUNDARY
C CONDITIONS ABOVE. I.E. THE INITIAL CONDITION FUNCTIONS MUST
C SATISFY THE BOUNDARY CONDITIONS FOR T = T0. SEE SUBROUTINE UINIT
C DESCRIPTION BELOW.
-----
C REQUIRED USER SUPPLIED SUBROUTINES
C THE USER IS REQUIRED TO CONSTRUCT THREE SUBPROGRAMS AND A MAIN
C PROGRAM WHICH DEFINE THE PDE PROBLEM WHOSE SOLUTION IS TO BE
C ATTEMPTED. THE THREE SUBPROGRAMS ARE...
C C 1) SUBROUTINE F(T,X,U,UX,UXX,FVAL,NPDE,KNPDE,IC,KSH,T,KSKH)
C DIMENSION U(NPDE),UX(NPDE),UXX(NPDE)
C THIS ROUTINE DEFINES THE DESIRED PARTIAL DIFFERENTIAL
C EQUATIONS TO BE SOLVED. THE PACKAGE PROVIDES VALUES OF THE
C INPUT SCALARS T AND X AND INPUT ARRAYS (LENGTH NPDE) U, UX,
C AND UXX, AND THE USER MUST CONSTRUCT THIS ROUTINE TO COMPUTE
C THE OUTPUT ARRAY FVAL (LENGTH NPDE) WHICH CONTAINS THE
C CORRESPONDING VALUES OF THE RIGHT HAND SIDES OF THE DESIRED
C PARTIAL DIFFERENTIAL EQUATIONS, I.E.
C FVAL(K) = THE VALUE OF THE RIGHT HAND SIDE OF THE K-TH PDE IN
C THE PDE SYSTEM ABOVE, FOR K = 1 TO NPDE.
C THE INCOMING VALUE OF THE SCALAR QUANTITY X WILL BE A
C COLLOCATION POINT VALUE (SEE INITIAL AND CULPNT) AND THE
C INCOMING VALUES IN THE ARRAYS U, UX AND UXX CORRESPOND TO THIS
C POINT X AND TIME T.
C RETURN
C END
C C 2) SUBROUTINE BNDRY( T, X, U, UX, UXX, DBDYA, DBDXT, NPDE )
C DIMENSION U(NPDE), UX(NPDE), UXX(NPDE)
C DIMENSION DBDYA(NPDE),DBDXT(NPDE)
C THIS ROUTINE IS USED TO PROVIDE THE PDE PACKAGE WITH NEEDED
C INFORMATION ABOUT THE BOUNDARY CONDITION FUNCTIONS H AND Z
C ABOVE. THE PACKAGE PROVIDES VALUES OF THE INPUT VARIABLES
C T, X, U, AND UX, AND THE USER IS TO DEFINE THE CORRESPONDING
C OUTPUT VALUES OF THE DERIVATIVES OF THE FUNCTIONS H AND Z
C WHEN NECESSARY.
C DBDYA(K)=PARTIAL DERIVATIVE OF THE K-TH COMPONENT OF H
C WITH RESPECT TO U(X).
C DBDXT(K)=PARTIAL DERIVATIVE OF THE K-TH COMPONENT OF Z
C WITH RESPECT TO U(X).
C DDT(K)=DERIVATIVE OF THE K-TH COMPONENT OF Z
C WITH RESPECT TO T.

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115 CC      NOTE...ALL PARTIAL DERIVATIVES OF THE I-TH COMPONENT OF H
116 CC      WITH RESPECT TO U(I) OR U(J), I NOT EQUAL TO J,
117 CC      ARE ASSUMED TO BE NEGLIGIBLE.
118 CC      NOTE...THE INCOMING VALUE OF X WILL BE EITHER XLEFT OR XRIGHT.
119 CC      IF NO BOUNDARY CONDITION IS DESIRED FOR SAY THE K-TH PDE AT
120 CC      ONE OR BOTH OF THE ENDPOINTS XLEFT OR XRIGHT, THEN DHDUX(K,K)
121 CC      AND DHDUX(K,K) SHOULD BOTH BE SET TO ZERO WHEN BNDRY IS
122 CC      CALLED FOR THAT POINT. WE REFER TO THIS AS A NULL BOUNDARY
123 CC      CONDITION. THIS ROUTINE CAN BE STRUCTURED AS FOLLOWS...
124 CC      THE COMMON BLOCK /ENDPT/ IS NOT A PART OF PDECOL AND
125 CC      MUST BE SUPPLIED AND DEFINED BY THE USER.
126 CC      COMMON /ENDPT/ XLEFT
127 CC      IF (X.NE.XLEFT) GO TO 10
128 CC      HERE DEFINE AND SET PROPER VALUES FOR DHDUX(K,J), DHDUX(K,J),
129 CC      AND DZDT(K) FOR K,J = 1 TO NPDE FOR THE LEFT BOUNDARY POINT
130 CC      X = XLEFT.
131 CC      RETURN
132 CC      10 CONTINUE
133 CC      HERE DEFINE AND SET PROPER VALUES FOR DHDUX(K,J), DHDUX(K,J),
134 CC      AND DZDT(K) FOR K,J = 1 TO NPDE FOR THE RIGHT BOUNDARY POINT
135 CC      X = XRIGHT.
136 CC      RETURN
137 CC      END
138 CC
139 CC      3) SUBROUTINE UINIT(X, U, NPDE)
140 CC      DIMENSION U(NPDE)
141 CC      THIS ROUTINE IS USED TO PROVIDE THE PDE PACKAGE WITH THE
142 CC      NEEDED INITIAL CONDITION FUNCTION VALUES. THE PACKAGE
143 CC      PROVIDES A VALUE OF THE INPUT VARIABLE X, AND THE USER IS TO
144 CC      DEFINE THE PROPER INITIAL VALUES (AT I = 10) FOR ALL OF THE
145 CC      PDE COMPONENTS, I.E.
146 CC      U(I) = DESIRED INITIAL VALUE OF PDE COMPONENT U(I) AT
147 CC      X AND I = 10 FOR K = 1 TO NPDE.
148 CC      NOTE...THE INCOMING VALUE OF X WILL BE A COLLOCATION POINT
149 CC      VALUE. THE INITIAL CONDITIONS AND BOUNDARY CONDITIONS
150 CC      MUST BE CONSISTENT (SEE ABOVE).
151 CC      RETURN
152 CC      END
153 CC
154 CC      -----
155 CC      OPTIONAL USER SUPPLIED SUBROUTINE
156 CC
157 CC      IF THE USER DESIRES TO USE THE MF = 11 OR 21 OPTION IN ORDER TO SAVE
158 CC      ABOUT 10-20 PERCENT IN EXECUTION TIME (SEE BELOW), THEN THE USER MUST
159 CC      PROVIDE THE FOLLOWING SUBROUTINE WHICH PROVIDES INFORMATION ABOUT THE
160 CC      DERIVATIVES OF THE FUNCTION F ABOVE. THIS PROVIDES FOR MORE EFFICIENT
161 CC      JACOBIAN MATRIX GENERATION. ON MOST COMPUTER SYSTEMS, THE USER WILL
162 CC      BE REQUIRED TO SUPPLY THIS SUBROUTINE AS A DUMMY SUBROUTINE IF THE
163 CC      OPTIONS MF = 12 OR 22 ARE USED (SEE BELOW).
164 CC      THIS OPTION IS NOT USED BY THE FLAME CODE. THE MESSAGE
165 CC      SUBROUTINE IS VERY DIFFICULT TO CONSTRUCT.
166 CC
167 CC      1) SUBROUTINE DERIVF(X, U, UA, UXA, OFDU, OFDUX, OFDUA, NPDE)
168 CC      DIMENSION U(NPDE), UA(NPDE), UXA(NPDE)
169 CC      DIMENSION OFDU(NPDE,NPDE), OFDUA(NPDE,NPDE), OFDUX(NPDE,NPDE)
170 CC      THE PACKAGE PROVIDES VALUES OF THE INPUT VARIABLES X, U, UA,
171 CC      AND UXA, AND THE USER SHOULD CONSTRUCT THIS ROUTINE TO PROVIDE
172 CC

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04/15/80 11.10.37

FTN 4.8+49H

76/76 OPT=1 MOUND=\*\*/ I-ACE

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C THE FOLLOWING CORRESPONDING VALUES OF THE OUTPUT ARRAYS
C DFUDX, DFUDXX, AND DFUDXX FOR K,J = 1 TO NPUR...
C DFUD(K,J) = PARTIAL DERIVATIVE OF THE K-TH COMPONENT OF THE
C PUE DEFINING FUNCTION F WITH RESPECT TO THE
C VARIABLE U(J).
C DFUDX(K,J) = PARTIAL DERIVATIVE OF THE K-TH COMPONENT OF THE
C PUE DEFINING FUNCTION F WITH RESPECT TO THE
C VARIABLE U(J).
C DFUDXX(K,J) = PARTIAL DERIVATIVE OF THE K-TH COMPONENT OF THE
C PUE DEFINING FUNCTION F WITH RESPECT TO THE
C VARIABLE U(K,J).
C NOTE... THE INCOMING VALUE OF X WILL BE A COLLOCATION POINT
C VALUE.
C RETURN
C END
C-----
C METHODS USED
C THE PACKAGE PDECOL IS BASED ON THE METHOD OF LINES AND USES A
C FINITE ELEMENT COLLOCATION PROCEDURE WITH PIECEWISE POLYNOMIALS
C AS THE TRIAL SPACE FOR THE DISCRETIZATION OF THE SPATIAL VARIABLE
C X. THE COLLOCATION PROCEDURE REDUCES THE PUE SYSTEM TO A SEMI-
C DISCRETE SYSTEM WHICH THEN DEPENDS ONLY ON THE TIME VARIABLE T.
C THE TIME INTEGRATION IS THEN ACCOMPLISHED BY USE OF SLIGHTLY
C MODIFIED STANDARD TECHNIQUES (SEE REFS. 1,2).
C
C PIECEWISE POLYNOMIALS
C THE USER IS REQUIRED TO SELECT THE PIECEWISE POLYNOMIAL SPACE
C WHICH IS TO BE USED TO COMPUTE HIS APPROXIMATE SOLUTION. FIRST, THE
C ORDER, KORD, OF THE POLYNOMIALS TO BE USED MUST BE SPECIFIED
C (KORD = POLYNOMIAL DEGREE + 1). NEXT, THE NUMBER OF PIECES
C (INTERVALS), NINT, INTO WHICH THE SPATIAL DOMAIN (LEFT,RIGHT) IS
C TO BE DIVIDED, IS CHOSEN. THE NINT + 1 DISTINCT BREAKPOINTS OF
C THE DOMAIN MUST BE DEFINED AND SET INTO THE ARRAY XBRKT IN
C STRICTLY INCREASING ORDER, I.E.
C XLEFT=XBRKT(1) .LT. XBRKT(2) .LT. ... .LT. XBRKT(NINT+1)=XRIGHT.
C THE APPROXIMATE SOLUTION AT ANY TIME T WILL BE A POLYNOMIAL OF
C ORDER KORD OVER EACH SUBINTERVAL (XBRKT(I)-XBRKT(I+1)). THE
C NUMBER OF CONTINUITY CONDITIONS, NCC, TO BE IMPOSED ACROSS ALL OF
C THE BREAKPOINTS IS THE LAST PIECE OF USER SUPPLIED DATA WHICH IS
C REQUIRED TO UNIQUELY DETERMINE THE DESIRED PIECEWISE POLYNOMIAL
C SPACE. FOR EXAMPLE, NCC = 2 WOULD REQUIRE THAT THE APPROXIMATE
C SOLUTION (MADE UP OF THE SEPARATE POLYNOMIAL PIECES) AND ITS FIRST
C SPATIAL DERIVATIVE BE CONTINUOUS AT THE BREAKPOINTS AND MATCH ON
C THE ENTIRE DOMAIN (XLEFT,XRIGHT). NCC = 3 WOULD REQUIRE THAT THE
C APPROXIMATE SOLUTION AND ITS FIRST AND SECOND SPATIAL DERIVATIVES
C BE CONTINUOUS AT THE BREAKPOINTS, ETC. THE DIMENSION OF THIS LINEAR
C SPACE IS KNOWN AND FINITE AND IS LCPTS = KORD*NINT - NCC*(NINT-1).
C THE WELL-KNOWN H-SPLINE BASIS (SEE REF. 3) FOR THIS SPACE IS USED
C BY PDECOL AND IT CONSISTS OF NCPIS KNOWN PIECEWISE POLYNOMIAL
C FUNCTIONS HF(I,X), FOR I=1 TO NCPIS, WHICH DO NOT DEPEND ON THE
C TIME VARIABLE T. WE WISH TO EMPHASIZE THAT THE PIECEWISE POLYNOMIAL
C SPACE USED IN PDECOL (WHICH IS SELECTED BY THE USER) WILL DETERMINE
C THE MAGNITUDE OF THE SPATIAL DISCRETIZATION ERRORS IN THE COMPUTED
C APPROXIMATE SOLUTION. THE PACKAGE HAS NO CONTROL OVER ERRORS

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04/15/80 11.10.37

FTN 4.4+498

76/76 OUT=1 MOUND=0.0/ 1-ACE

230 C INTRODUCED BY THE USER'S CHOICE OF THIS SPACE. SEE INPUT PARAMETERS  
231 C BELOW.  
232 C  
233 C COLLOCATION OVER PIECEWISE POLYNOMIALS  
234 C  
235 C THE BASIC ASSUMPTION MADE IS THAT THE APPROXIMATE SOLUTION  
236 C SATISFIES  
237 C  
238 C 
$$U(T,X) = \sum_{i=1}^{NCPTS} C(i,1) * HF(I,X)$$
  
239 C  
240 C WHEN THE UNKNOWN COEFFICIENTS C DEPEND ONLY ON THE TIME T AND  
241 C THE KNOWN BASIS FUNCTIONS DEPEND ONLY ON X (WE HAVE ASSUMED THAT  
242 C  $NPUE = 1$  FOR CONVENIENCE). SO, AT ANY GIVEN TIME T THE APPROX-  
243 C IMATE SOLUTION IS A PIECEWISE POLYNOMIAL IN THE USER CHOSEN SPACE.  
244 C THE SEMI-DISCRETE EQUATIONS (ACTUALLY ORDINARY DIFFERENTIAL  
245 C EQUATIONS) WHICH DETERMINE THE COEFFICIENTS C ARE OBTAINED BY  
246 C REQUIRING THAT THE ABOVE APPROXIMATE  $U(T,X)$  SATISFY THE PDE AND  
247 C BOUNDARY CONDITIONS EXACTLY AT A SET OF NCPTS COLLOCATION POINTS  
248 C (SEE COLPNT). THUS, PUECOL ACTUALLY COMPUTES THE BASIS FUNCTION  
249 C COEFFICIENTS RATHER THAN SPECIFIC APPROXIMATE SOLUTION VALUES.  
250 C  
251 C REFERENCES  
252 C  
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265 C  
266 C  
267 C  
268 C  
269 C  
270 C  
271 C  
272 C  
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274 C  
275 C  
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04/15/80 11.10.37

FIN 4.8.64H

76/76 OPT=1 MOUNDD=.../ PHACE

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C NINT = ENOUGH SO THAT ANY FINE STRUCTURE OF THE PROBLEM MAY BE
C RESOLVED.
C KORD = 4
C NCC = 2
C MF = 22
C INDX = 1 (ON FIRST CALL ONLY, THEN 0 THEREAFTER).

C THE INPUT PARAMETERS ARE..
C TU = THE INITIAL VALUE OF T, THE INDEPENDENT VARIABLE
C      (USED ONLY ON FIRST CALL).
C TOUT = THE VALUE OF T AT WHICH OUTPUT IS DESIRED NEXT. SINCE
C      THE PACKAGE CHOOSES ITS OWN TIME STEP SIZES, THE
C      INTEGRATION WILL NORMALLY GO SLIGHTLY BEYOND TOUT
C      AND THE PACKAGE WILL INTERPOLATE TO T = TOUT.
C UT = THE INITIAL STEP SIZE IN T. IF INDEX = 1, OR, THE
C      MAXIMUM STEP SIZE ALLOWED (MUST BE .GT. 0). IF INDEX = 3,
C      USED FOR INPUT ONLY WHEN INDEX = 1 OR 3. SEE BELOW.
C ARKPT = THE ARRAY OF PIECEWISE POLYNOMIAL BREAKPOINTS.
C      THE NINT+1 VALUES MUST BE STRICTLY INCREASING WITH
C      ARKPT(1) = LEFT AND ARKPT(NINT+1) = RIGHT (USED ONLY
C      ON FIRST CALL).
C EPS = THE RELATIVE TIME ERROR TOL (USED ONLY ON THE
C      FIRST CALL, UNLESS INDEX = 4). SINGLE STEP ERROR
C      ESTIMATES DIVIDED BY CMAX(1) WILL BE KEPT LESS THAN
C      EPS IN ROOT-MEAN-SQUARE NORM. THE VECTOR CMAX OF WEIGHTS
C      IS COMPUTED IN PDECOL. INITIALLY CMAX(1) IS SET TO
C      ABS(C(1)), WITH A DEFAULT VALUE OF 1 IF ABS(C(1)) .LT. 1.
C      THEREAFTER, CMAX(1) IS THE LARGEST VALUE
C      OF ABS(C(I)) SEEN SO FAR, OR THE INITIAL CMAX(1) IF
C      THAT IS LARGER. TO ALTER EITHER OF THESE, CHANGE THE
C      APPROPRIATE STATEMENTS IN THE DO-LOOPS ENDING AT
C      STATEMENTS 50 AND 130 BELOW. THE USER SHOULD EXERCISE
C      SOME DISCRETION IN CHOOSING EPS. IN GENERAL, THE
C      OVERALL RUNNING TIME FOR A PROBLEM WILL BE GREATER IF
C      EPS IS CHOSEN SMALLER. THERE IS USUALLY LITTLE REASON TO
C      CHOOSE EPS MUCH SMALLER THAN THE ERRORS WHICH ARE BEING
C      INTRODUCED BY THE USER'S CHOICE OF THE POLYNOMIAL SPACE.
C      SEE RELATED COMMENTS CONCERNING CMAX BELOW STATEMENT 40.
C NINT = THE NUMBER OF SUBINTERVALS INTO WHICH THE SPATIAL DOMAIN
C      (XLEFT-XRIGHT) IS TO BE DIVIDED (MUST BE .GE. 1)
C      (USED ONLY ON FIRST CALL).
C KORD = THE ORDER OF THE PIECEWISE POLYNOMIAL SPACE TO BE USED.
C      ITS VALUE MUST BE GREATER THAN 2 AND LESS THAN 21. FOR
C      FIRST ATTEMPTS WE RECOMMEND KORD = 4. IF THE SOLUTION
C      IS SMOOTH AND MUCH ACCURACY IS DESIRED, HIGHER VALUES
C      MAY PROVE TO BE MORE EFFICIENT. WE HAVE SELDOM USED
C      VALUES OF KORD IN EXCESS OF 4 OR 9, THOUGH THEY ARE
C      AVAILABLE FOR USE IN PDECOL (USED ONLY ON FIRST CALL).
C NCC = THE NUMBER OF CONTINUITY CONDITIONS TO BE IMPOSED ON THE
C      APPROXIMATE SOLUTION AT THE BREAKPOINTS IN ARKPT.
C      NCC MUST BE GREATER THAN 1 AND LESS THAN KORD. WE
C      RECOMMEND THE USE OF NCC = 2 (WITH INDEGS = 0, SEE
C      BELOW). SINCE THEORY PREDICTS THAT DRAMATICALLY MORE
C      ACCURATE RESULTS CAN OBTAIN BE OBTAINED USING THIS CHOICE
C      (USED ONLY ON FIRST CALL).
C NPHIE = THE NUMBER OF PARTIAL DIFFERENTIAL EQUATIONS IN THE SYSTEM

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C MF = THE METHOD FLAG (USED ONLY ON FIRST CALL).
C
C 365 INDEX = 4). ALLOWED VALUES ARE 1, 12, 21, 22.
C FOR FIRST ATTEMPTS WE RECOMMEND THE USE OF MF = 22.
C MF HAS TWO DECIMAL DIGITS, METH AND MITER
C (MF = 100METH + MITER).
C
C 370 METH IS THE BASIC METHOD INDICATOR..
C METH = 1 MEANS THE 44ANS METHODS (GENERALIZATIONS OF
C CRANK-NICOLSON).
C
C 375 METH = 2 MEANS THE BACKWARD DIFFERENTIATION
C FORMULAS (DIFF) OR STIFF METHODS OF GEAR.
C MITER IS THE ITERATION METHOD INDICATOR
C AND DETERMINES HOW THE JACOBIAN MATRIX IS
C TO BE COMPUTED..
C MITER = 1 MEANS CHOUD METHOD WITH ANALYTIC JACOBIAN.
C FOR THIS USER SUPPLIES SUBROUTINE DLMVIF.
C SEE DESCRIPTION ABOVE.
C
C 380 MITER = 2 MEANS CHOUD METHOD WITH JACOBIAN CALCULATED
C INTERNALLY BY FINITE DIFFERENCES. SEE
C SUBROUTINE PSETH AND DIFF.
C
C INDEX = INTEGER USED ON INPUT TO INDICATE TYPE OF CALL.
C WITH THE FOLLOWING VALUES AND MEANINGS..
C
C 385 1 THIS IS THE FIRST CALL FOR THIS PROBLEM.
C 0 THIS IS NOT THE FIRST CALL FOR THIS PROBLEM,
C AND INTEGRATION IS TO CONTINUE.
C
C 390 2 SAME AS 0 EXCEPT THAT TOUT IS TO BE HIT
C EXACTLY (NO INTERPOLATION IS DONE). SEE NOTE
C BELOW. ASSUMES TOUT .GE. THE CURRENT T.
C IF TOUT IS .LT. THE CURRENT TIME, THEN TOUT IS
C RESET TO THE CURRENT TIME AND CONTROL IS
C RETURNED TO THE USER. A CALL TO VALUES WILL
C PRODUCE ANSWERS FOR THE NEW VALUE OF TOUT.
C
C 395 3 SAME AS 0 EXCEPT CONTROL RETURNS TO CALLING
C PROGRAM AFTER ONE STEP. TOUT IS IGNORED AND
C IT MUST BE SET .GT. 0 TO A MAXIMUM ALLOWED
C DT VALUE. SEE ABOVE.
C
C 400 4 THIS IS NOT THE FIRST CALL FOR THE PROBLEM.
C AND THE USER HAS RESET EPS AND/OR MF.
C SINCE THE NORMAL OUTPUT VALUE OF INDEX IS 0,
C IT NEED NOT BE RESET FOR NORMAL CONTINUATION.
C
C NOTE.. THE PACKAGE MUST HAVE TAKEN AT LEAST ONE SUCCESSFUL TIME
C STEP BEFORE A CALL WITH INDEX = 2 OR 4 IS ALLOWED.
C AFTER THE INITIAL CALL, IF A NORMAL RETURN OCCURRED AND A NORMAL
C CONTINUATION IS DESIRED, SIMPLY RESET TOUT AND CALL AGAIN.
C ALL OTHER PARAMETERS WILL BE READY FOR THE NEXT CALL.
C A CHANGE OF PARAMETERS WITH INDEX = 4 CAN BE MADE AFTER
C EITHER A SUCCESSFUL OR AN UNSUCCESSFUL RETURN PROVIDED AT LEAST
C ONE SUCCESSFUL TIME STEP HAS BEEN MADE.
C
C 405 WORK = FLOATING POINT WORKING ARRAY FOR PDECOL. WE RECOMMEND
C THAT IT BE INITIALIZED TO ZERO BEFORE THE FIRST CALL
C TO PDECOL. ITS TOTAL LENGTH MUST BE AT LEAST
C
C 410 NDEP + 130000 + NDEP*(1000000) +
C NDEP*NCPTS*(100000000)

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400 C
401 C WHERE ML AND MAXDEM ARE DEFINED BELOW (SEE STORAGE
402 C ALLOCATION).
403 C IWORK = INTEGER WORKING ARRAY FOR PDECOL. THE FIRST TWO
404 C LOCATIONS MUST BE DEFINED AS FOLLOWS...
405 C IWORK(1) = LENGTH OF USER'S ARRAY WORK
406 C IWORK(2) = LENGTH OF USER'S ARRAY IWORK
407 C THE TOTAL LENGTH OF IWORK MUST BE AT LEAST
408 C NCPTS*(NPUZ + 1).
409 C
410 C C OUTPUT
411 C
412 C THE SOLUTION VALUES ARE NOT RETURNED DIRECTLY TO THE USER BY PDECOL.
413 C THE METHODS USED IN PDECOL COMPUTE COEFFICIENTS, SO
414 C THE USER (AFTER A RETURN FROM PDECOL) MUST CALL THE PACKAGE ROUTINE
415 C VALUES TO OBTAIN HIS APPROXIMATE SOLUTION VALUES AT ANY DESIRED SPACE
416 C POINTS X AT THE TIME T = TOUT. SEE THE COMMENTS IN SUBROUTINE VALUES
417 C FOR DETAILS ON HOW TO PROPERLY MAKE THE CALL.
418 C
419 C THE COMMON BLOCK /GEARD/ CAN BE ACCESSED EXTERNALLY BY THE USER
420 C IF DESIRED. IT CONTAINS THE STEP SIZE LAST USED (SUCCESSFULLY),
421 C THE ORDER LAST USED (SUCCESSFULLY), THE NUMBER OF STEPS TAKEN
422 C SO FAR, THE NUMBER OF RESIDUAL EVALUATIONS (NLS CALLS) SO FAR,
423 C AND THE NUMBER OF MATRIX EVALUATIONS (PSETM CALLS) SO FAR.
424 C DIFFUN CALLS ARE COUNTED IN WITH RESIDUAL EVALUATIONS.
425 C
426 C THE OUTPUT PARAMETERS ARE...
427 C DT = THE STEP SIZE USED LAST, WHETHER SUCCESSFULLY OR NOT.
428 C TOUT = THE OUTPUT VALUE OF T. IF INTEGRATION WAS SUCCESSFUL,
429 C AND THE INPUT VALUE OF INDEX WAS NOT 3, TOUT IS
430 C UNCHANGED FROM ITS INPUT VALUE. OTHERWISE, TOUT
431 C IS THE CURRENT VALUE OF T TO WHICH THE INTEGRATION
432 C HAS BEEN COMPLETED.
433 C INDEX = INTEGER USED ON OUTPUT TO INDICATE RESULTS,
434 C WITH THE FOLLOWING VALUES AND MEANINGS...
435 C 0 INTEGRATION WAS COMPLETED TO TOUT OR BEYOND.
436 C -1 THE INTEGRATION WAS HALTED AFTER FAILING TO PASS THE
437 C ERROR TEST EVEN AFTER REDUCING DT BY A FACTOR OF
438 C 1.E10 FROM ITS INITIAL VALUE.
439 C -2 AFTER SOME INITIAL SUCCESS, THE INTEGRATION WAS
440 C HALTED EITHER BY REPEATED ERROR TEST FAILURES OR BY
441 C A TEST ON EPS. TOO MUCH ACCURACY HAS BEEN REQUESTED.
442 C THE INTEGRATION WAS HALTED AFTER FAILING TO ACHIEVE
443 C CORRECTOR CONVERGENCE EVEN AFTER REDUCING DT BY A
444 C FACTOR OF 1.E10 FROM ITS INITIAL VALUE.
445 C -3 SINGULAR MATRIX ENCOUNTERED. PROBABLY DUE TO STORAGE
446 C OVERWRITES.
447 C -4 INDEX WAS 4 ON INPUT, BUT THE DESIRED CHANGES OF
448 C PARAMETERS WERE NOT IMPLEMENTED BECAUSE TOUT
449 C WAS NOT BEYOND T. INTERPOLATION TO T = TOUT WAS
450 C PERFORMED AS ON A NORMAL RETURN. TO TRY AGAIN,
451 C SIMPLY CALL AGAIN WITH INDEX = 4 AND A NEW TOUT.
452 C ILLEGAL INDEX VALUE.
453 C -5 ILLEGAL EPS VALUE.
454 C -6 AN ATTEMPT TO INTEGRATE IN THE WRONG DIRECTION. THE
455 C SIGN OF DT IS WRONG RELATIVE TO DT AND TOUT.
456 C -7 DT .EQ. 0.0.
457 C -8 ILLEGAL NLS VALUE.
458 C -9 ILLEGAL NPUZ VALUE.
459 C -10
460 C -11

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C      -12  ILLEGAL NCC VALUE.          PDECOL 458
C      -13  ILLEGAL NPDE VALUE.         PDECOL 459
C      -14  ILLEGAL MF VALUE.           PDECOL 460
C      -15  ILLEGAL BREAKPOINTS - NOT STRICTLY INCREASING. PDECOL 461
C      -16  INSUFFICIENT STORAGE FOR WORK OR IROHR. PDECOL 462
C      ----- PDECOL 463
C      PDECOL 464
C      PDECOL 465
C      PDECOL 466
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C      PDECOL 513
C      PDECOL 514

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C DIFF - PERFORMS SAME ROLE AS THE USER ROUTINE DERIVE. COMPUTES
C DERIVATIVE APPROXIMATIONS BY USE OF FINITE DIFFERENCES.
C
C DECH - PERFORM AN LU DECOMPOSITION AND FORWARD AND BACKWARD
C SOLV SUBSTITUTION FOR SOLVING Banded SYSTEMS OF LINEAR EQUATIONS.
C
C-----
C
C STORAGE ALLOCATION
C
C SINCE PDECOL IS A DYNAMICALLY DIMENSIONED PROGRAM, MOST OF ITS
C WORKING STORAGE IS PROVIDED BY THE USER IN THE ARRAYS WORK AND IWORK.
C THE FOLLOWING GIVES A LIST OF THE ARRAYS WHICH MAKE UP THE CONTENTS
C WORK AND IWORK, THEIR LENGTHS, AND THEIR USES. WHEN MORE THAN ONE
C NAME IS GIVEN, IT INDICATES THAT DIFFERENT NAMES ARE USED FOR THE
C SAME ARRAY IN DIFFERENT PARTS OF THE PROGRAM. THE DIFFERENT NAMES
C OCCUR BECAUSE PDECOL IS AN AMALGAMATION OF SEVERAL OTHER CODES
C WRITTEN BY DIFFERENT PEOPLE AND WE HAVE TRIED TO LEAVE THE SEPARATE
C PARTS AS UNCHANGED FROM THEIR ORIGINAL VERSIONS AS POSSIBLE.
C
C NAMES LENGTH USE
C-----
C HC 4*NPDE BOUNDARY CONDITION INFORMATION
C WORK
C
C A 3*KORD*NCPTS BASIS FUNCTION VALUES AT COLLOCATION POINT
C WORK(IW1)
C
C XT NCPTS + KORD BREAKPOINT SEQUENCE FOR GENERATION OF BASIS
C WORK(IW2) FUNCTION VALUES.
C
C XC NCPTS COLLOCATION POINTS.
C WORK(IW3)
C
C CMAX NPDE*NCPTS VALUES USED IN ESTIMATING TIME
C YMAX INTEGRATION ERRORS.
C WORK(IW4)
C
C FWORK NPDE*NCPTS TIME INTEGRATION ERRORS.
C WORK(IW5)
C
C SAVE1 NPDE*NCPTS WORKING STORAGE FOR THE TIME INTEGRATION
C WORK(IW6) METHOD.
C
C SAVE2 NPDE*NCPTS WORKING STORAGE FOR THE TIME INTEGRATION
C WORK(IW7) METHOD.
C
C SAVE3 NPDE*NCPTS WORKING STORAGE FOR THE TIME INTEGRATION
C WORK(IW8) METHOD.
C
C UVAL 3*NPDE WORKING STORAGE FOR VALUES OF U, UX, AND
C WORK(IW9) UX AT THE POINT.
C
C C NPDE*NCPTS CURRENT ASIS FORATION COEFFICIENT VALUES
C

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C AS INDICATED BELOW.

C PACKAGE ROUTINES CALLED.. EVAL,INITIAL,INTERP,STIFIN

C USER ROUTINES CALLED.. HNDHY

C CALLED BY.. USERS MAIN PROGRAM

C FORTRAN FUNCTIONS USED.. AMS,AMAX1,FLUAT,SURT

C-----

      DIMENSION WORK(1),IWORK(1),XHKPT(1)

      COMMON /GEAR0/ DTUSED,NUUSED,NSIFP,NFE,NJE

      COMMON /GEAR1/ T,UTC,DTM,N,DTM,N,DTM,N,MFC,KFLAG,JSTART

      COMMON /GEAR2/ EPSJ,RU,ML,MU,MW,N,NUML,JU,NOWI

      COMMON /OPTION/ HOGAUS,MAXDER

      COMMON /SIZES/ NIT,KUR,NC,NPD,NCTS,NEQN,IQUAD

      COMMON /ISTART/ IWI,IW2,IW3,IW4,IW5,IW6,IW7,IW8,IW9,IW10,

      IWI1,IWI2,IWI3,IWI4,IWI5,IWI6,IWI7,IWI8

      COMMON /IQUANT/ LOUT

      IF (INDEX .EQ. 0) GO TO 60

      IF (INDEX .EQ. 2) GO TO 70

      IF (INDEX .EQ. 4) GO TO 80

      IF (INDEX .EQ. 3) GO TO 90

      C-----

C SEVERAL CHECKS ARE MADE HERE TO DETERMINE IF THE INPUT PARAMETERS

C HAVE LEGAL VALUES. ERROR CHECKS ARE MADE ON INDEX, EPS, (TU-TOUT)\*DI,

C DI, NINT, KORD, NCC, NPD, MF, WHETHER THE BREAKPOINT SEQUENCE IS

C STRICTLY INCREASING, AND WHETHER THERE IS SUFFICIENT STORAGE

C PROVIDED FOR WORK AND IWORK. PROBLEM DEPENDENT PARAMETERS ARE

C CALCULATED AND PLACED IN COMMON.

C-----

      IERID = -6

      IF (INDEX .NE. 1) GO TO 320

      IERID = IERID - 1

      IF (EPS .LE. 0.) GO TO 320

      IERID = IERID - 1

      IF ((TU-TOUT)\*DI .GT. 0.) GO TO 320

      IERID = IERID - 1

      IF (DT .EQ. 0.0) GO TO 320

      IERID = IERID - 1

      NIN = NINT

      IF (NIN .LT. 1) GO TO 320

      IERID = IERID - 1

      KUR = KORD

      IF (KOR .LT. 3 .OR. KOR .GT. 20) GO TO 320

      IERID = IERID - 1

      NC = NCC

      IF (NCC .LT. 2 .OR. NCC .GT. 100) GO TO 320

      IERID = IERID - 1

      NPD = NPD

      NPD2 = NPD\*NPD

      IF (NPD2 .LT. 1) GO TO 320

      IERID = IERID - 1

      IF (MF .NE. 22 .AND. MF .NE. 21 .AND. MF .NE. 12 .AND. MF .NE. 11) GO TO 320

629 PDECOL

630 PDECOL

631 PDECOL

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645 ML=KOR-2
    MU = ML
    MW = ML * ML * 1
    NU=NEUN*(3*ML*1)
    NUL=NCPTS*(3*ML*1)
    IWSAVE = IWORK(1)
    IISAVE = IWORK(2)
    IW1=4*NPDE*1
    IW2 = IW1 * 3*KORU*NCPTS
    IW3 = IW2 * NCPTS * KORD
    IW4 = IW3 * NCPTS
    IW5 = IW4 * NEUN
    IW6 = IW5 * NEON
    IW7 = IW6 * NEON
    IW8 = IW7 * NEON
    IW9 = IW8 * NEON
    IW10 = IW9 * 3*NPDE
    IW11 = IW10 * NEUN*(MAXDER*1)
    IW12=IW11*NPDE
    IW13=IW12*NPDE
    IW14=IW13*NPDE
    IW15=IW14*NPDE
    IW16=IW15*NPDE
    IW17 = IW16 * NPDE
    IW18 = NCPTS * 1
    IEPID = IEPID - 1
    IINSTOR = IW17 * NEUN*(3*ML*1) - 1
    IINSTOR = IW18 * NEUN - 1
    IF ( IWSAVE .LT. IINSTOR .OR. IISAVE .LT. IINSTOR ) GO TO 335
-----
C PERFORM INITIALIZATION TASKS. IF KORD .EQ. 3 THEN CALCULATE THE HAND-
C WIDTH OF THE ASSOCIATED MATRIX PROBLEM BY DETERMINING THE TYPE OF
C BOUNDARY CONDITIONS. THEN CHECK FOR SUFFICIENT STORAGE AGAIN.
C-----
    CALL INITIAL(KOR,WORK(IW1),WORK(IW6),XKRPT,WORK(IW2),WORK(IW3),
    * WORK(IW17),IWORK(IW14),IWORK)
    IF (IQUAD .NE. 0) GO TO 280
    IINSTOR = IW17 * NEUN*(3*ML*1) - 1
    IF ( IWSAVE .LT. IINSTOR ) GO TO 335
-----
C IF INITIAL VALUES OF CMAX OTHER THAN THOSE SET BELOW ARE DESIRED,
C THEY SHOULD BE SET HERE. ALL CMAX(1) MUST BE POSITIVE.
C HAVING PROPER VALUES OF CMAX FOR THE PROBLEM BEING SOLVED IS AS
C IMPORTANT AS CHOOSING EPS (SEE ABOVE). SINCE ERRORS ARE
C MEASURED RELATIVE TO CMAX. IF VALUES FOR UMIN OR UTMX, THE
C ROUNDS ON ABS(UT), OTHER THAN THOSE BELOW ARE DESIRED, THEY
C SHOULD BE SET BELOW.
C-----
CC SPEC MODIFIES THE ERROR CONTROL.
CC AN ABSOLUTE ERROR CRITERION IS USED FOR QUANTITIES LESS THAN SPEC.
    DO 50 I = 1,NEUN
        II = I - 1
        WORK(IW4*11)=AMAX1(WORK(IW5*11),SPEC)
50 WORK(IW10*11) = WORK(IW4*11)
    I = NEUN
    I = 70
    ITC = 07

```

04/15/80 11.10.37

FIN 4.8.44M

SUBROUTINE PDLCOL 76/76 OPT=1 MOUNO=--0/ TRACE

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DTMX = ABS(DT)  
 DTUSED = 0.  
 EPSC = EPS  
 MFC = MF  
 JSTART = 0  
 EPSJ = SORT(UKOUNO)  
 NM1 = NEON - 1  
 NM1 = NEON \* ML  
 NMOUT = 0  
 KFLAG = 0  
 TOUTP = T0  
 IF ( T0 .EQ. TOUT ) GO TO 360  
 60 DTMX = ABS(TOUT-TOUTP)\*10.  
 GO TO 140

70 DTMX = ABS(TOUT-TOUTP)\*10.  
 IF ((T-TOUT)\*DTX .GE. 0.) GO TO 340  
 GO TO 150

80 IF ((T-TOUT)\*DTX .GE. 0.) GO TO 300  
 JSTART = -1  
 EPSC = EPS  
 MFC = MF  
 GO TO 100

90 DTMX = DT  
 100 IF ((T-DTC) .EQ. T) WRITE(LOUT,110)  
 110 FORMAT(36H WARNING.. T \* DT = 1 ON NEXT STEP.)  
 C TAKE A TIME STEP BY CALLING THE INTEGRATOR.  
 CALL STIFIR(NCON,TOUT,WORK(1M10),WORK(1M2),WORK(1M3),WORK(1M6),  
 \* WORK(1M7),WORK(1M9),WORK(1M17),WORK(1M1H),WORK,WORK)  
 KGO = 1 - KFLAG  
 GO TO (120, 160, 220, 260, 240), KGO  
 KFLAG = 0, -1, -2, -3, -4  
 120 CONTINUE  
 C NORMAL RETURN FROM INTEGRATOR.  
 C THE WEIGHTS CMAX(1) ARE UPDATED. IF DIFFERENT VALUES ARE DESIRED,  
 C THEY SHOULD BE SET HERE. A TEST IS MADE FOR EPS BEING TOO SMALL  
 C FOR THE MACHINE PRECISION.  
 C ANY OTHER TESTS OR CALCULATIONS THAT ARE REQUIRED AFTER EVERY  
 C STEP SHOULD BE INSERTED HERE.  
 C IF INDEX = 3, SAVE1 IS SET TO THE CURRENT C VALUES ON RETURN.  
 C IF INDEX = 2, DT IS CONTROLLED TO HIT TOUT (WITHIN MOUNDOFF  
 C ERROR). AND THEN THE CURRENT C VALUES ARE PUT IN SAVE1 ON RETURN.  
 C FOR ANY OTHER VALUE OF INDEX, CONTROL RETURNS TO THE INTEGRATOR.  
 C UNLESS TOUT HAS BEEN REACHED. THEN INTERPOLATED VALUES OF C ARE  
 C COMPUTED AND STORED IN SAVE1 ON RETURN.  
 C IF INTERPOLATION IS NOT DESIRED, THE CALL TO INTTAP SHOULD BE  
 C REMOVED AND CONTROL TRANSFERRED TO STATEMENT 340 INSTEAD OF 360.



04/15/80 11.10.37

FIN 4.8.49H

SUBROUTINE PDECOL 76/76 OPT=1 MOUND=\*\*/ I=ACE

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      * 45H CONVECTION CONVERGENCE COULD NOT BE ACHIEVED//
      GO TO 180
C
      240 WHITE (LOUT,290)
      290 FORMAT(/24H SINGULAR MATRIX ENCOUNTERED,
      * 35H PROBABLY DUE TO STORAGE OVERWRITES//)
      KFLAG = -4
      GO TO 340
C
      300 WHITE (LOUT,310) T,TOUT,DTC
      310 FORMAT(/45H INDEX = -1 ON INPUT WITH (1-TOUT)*DT,GE. 0./
      * 44H T = E16.8,9H TOUT = E16.8,8H DTC = E16.8,7
      * 44H INTERPOLATION WAS DONE AS ON NORMAL RETURN./
      * 44H DESIRED PARAMETER CHANGES WERE NOT MADE.)
      CALL INTERP(TOUT,WORK(1M10),NEUT,WORK(1M6))
      INDEX = -5
      RETURN
C
      320 WHITE (LOUT,330) ILMID
      330 FORMAT(/24H ILLEGAL INPUT...IN+X= .13//)
      INDEX = IERID
      RETURN
C
      335 WHITE (LOUT,336) IWSIOR,IWSAVE,IJ,TOM,IJSAVE
      336 FORMAT(/21H INSUFFICIENT STORAGE/24H WORK MUST BE OF LENGTH,
      * 110.5X,12HYOU PROVIDED,110/24H WORK MUST BE OF LENGTH,110.5X,
      * 12HYOU PROVIDED,110//)
      INDEX = IERID
      RETURN
C
      340 TOUT = T
      DO 350 I = 1,NEUN
      350 I = I - 1
      350 WORK(1M6,11) = WORK(1M10,11)
      360 INDEX = KFLAG
      TOUTP = TOUT
      DT = DTUSED
      IF (KFLAG .NE. 0) DT = DTC
      RETURN
      END
      460
      465
      470
      475
      480
      485
      490
      495

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PDECOL H57  
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 PDECOL H96

CARD NR. SEVERITY DETAILS DIAGNOSIS OF PROBLEM  
 691 I IWORK ARRAY REFERENCE OUTSIDE DIMENSION HOUNDS.

```

1      SUBROUTINE VALUES(X,USOL,SC1CH,NDIM1,NDIM2,NPTS,NDERV,NWORK)
2      VALUES
3      C SUBROUTINE VALUES COMPUTES THE SOLUTION U AND THE FIRST NDERV
4      C DERIVATIVES OF U AT THE NPTS POINTS A AND AT THE FIRST NDERV
5      C THEM IN THE ARRAY USOL. THIS ROUTINE MUST BE USED TO OBTAIN
6      C SOLUTION VALUES SINCE PUECOL DOES NOT RETURN ANY SOLUTION VALUES
7      C TO THE USER. SEE PUECOL.
8      VALUES
9      C THE CALLING PARAMETERS ARE...
10     VALUES
11     C X = AN ARBITRARY VECTOR OF SPATIAL POINTS OF LENGTH NPTS AT
12     C WHICH THE SOLUTION AND THE FIRST NDERV DERIVATIVE VALUES
13     C ARE TO BE CALCULATED. IF X.LT. XLEFT (X.GT. XRIGHT)
14     C THEN THE PIECEWISE POLYNOMIAL OVER THE LEFTMOST (RIGHT-
15     C MOST) INTERVAL IS EVALUATED TO CALCULATE THE SOLUTION
16     C VALUES AT THIS UNUSUAL VALUE OF X. SEE PUECOL.
17     VALUES
18     C USOL = AN ARRAY WHICH CONTAINS THE SOLUTION AND THE FIRST
19     C NDERV DERIVATIVES OF THE SOLUTION AT ALL THE POINTS IN
20     C THE INPUT VECTOR X. IN PARTICULAR, USOL(J,I,K) CONTAINS
21     C THE VALUE OF THE (K-1)-ST DERIVATIVE OF THE J-TH PDE
22     C COMPONENT AT THE I-TH POINT OF THE X VECTOR FOR
23     C J = 1 TO NPUT, I = 1 TO NPTS, AND K = 1 TO NDERV+1.
24     VALUES
25     C SC1CH = A USER SUPPLIED WORKING STORAGE ARRAY OF LENGTH AT LEAST
26     C KORD*(NDERV+1). SEE HELPO AND PUECOL FOR DEFINITIONS OF
27     C THESE PARAMETERS.
28     VALUES
29     C NDIM1 = THE FIRST DIMENSION OF THE OUTPUT ARRAY USOL IN THE CALLING
30     C PROGRAM. NDIM1 MUST BE GE. NPTS.
31     VALUES
32     C NDIM2 = THE SECOND DIMENSION OF THE OUTPUT ARRAY USOL IN THE
33     C CALLING PROGRAM. NDIM2 MUST BE GE. NPTS.
34     VALUES
35     C NPTS = THE NUMBER OF POINTS IN THE X VECTOR.
36     VALUES
37     C NDERV = THE NUMBER OF DERIVATIVE VALUES OF THE SOLUTION THAT ARE
38     C TO BE CALCULATED. NDERV SHOULD BE LESS THAN KORD SINCE
39     C THE KORD-TH DERIVATIVE OF A POLYNOMIAL OF DEGREE KORD-1
40     C IS EQUAL TO ZERO. SEE PUECOL.
41     VALUES
42     C NWORK = THE USERS WORKING STORAGE ARRAY WHICH IS USED IN THIS CASE
43     C TO PROVIDE THE CURRENT HASTIS FUNCTION COEFFICIENTS AND THE
44     C PIECEWISE POLYNOMIAL AREAPPOINT SEQUENCE.
45     VALUES
46     C PACKAGE ROUTINES CALLED... HSPLVD,INTERV
47     C USER ROUTINES CALLED... NONE
48     C CALLED BY... USERS MAIN PROGRAM
49     C FORTHAN FUNCTIONS USED... NONE
50     VALUES
51     C-----
52     C DIMENSION USOL (NDIM1,NDIM2,NDERV),X (NPTS),SC1CH(1),WORK(1)
53     C COMMON /SIZES/ NINT,KORD,ACC,NPAC,NCPIS,NEUW,INUAD
54     C DATA ILEFT/0, MFLAU/0
55     C NDERV = NDERV + 1
56     C DO 20 IPTS=1,NPTS
57     C CALL INTERV(WORK(1,2),NCPIS,X(IPTS),ILEFT,MFLAU)
58     VALUES

```

SUBROUTINE	VALUES	76/76	OPT=1	ROUND=0.00/	1-ACE	FIN 4.0-4.9K	04/15/80	11-10-37	PAGE	1K
60	CALL HSPLVD(WORK(IW2),KORD,X(I-1TS),ILEFT,SCATCH,NDERVI)						VALUES	59		
	IK = ILEFT - KORD						VALUES	60		
	DO 10 M=1,NDERVI						VALUES	61		
	II = (M-1)*KORD						VALUES	62		
	DO 10 K=1,NPDE						VALUES	63		
	USOL(K,IPTS,M) = 0.						VALUES	64		
	IS=IW6-1+(K-1)*ICPTS*IK						VALUES	65		
	DO 10 I=1,KORD						VALUES	66		
	USOL(K,IPTS,M)=USOL(K,IPTS,M)+WII*IK(IS*I)*SCATCH(I,II)						VALUES	67		
65	10 CONTINUE						VALUES	68		
	20 CONTINUE						VALUES	69		
	RETURN						VALUES	70		
70	END						VALUES	71		

```

1 SUBROUTINE INITIAL(K,A,MHS,X,XI,AC,PW,IPIV,ILEFT)
2
3 C INITIAL IS CALLED ONLY ONCE BY PDECOL TO PERFORM INITIALIZATION TASKS.
4 C THESE TASKS INCLUDE - 1) DEFINING THE PIECEWISE POLYNOMIAL SPACE
5 C HREAPPOINT SEQUENCE, 2) CALLING THE SUBROUTINE CULPNT TO DEFINE THE
6 C REQUIRED COLLOCATION POINTS, 3) DEFINING THE PIECEWISE POLYNOMIAL SPACE
7 C HREAPPOINT SEQUENCE, 4) DEFINING THE INITIAL HASTIS FUNCTION
8 C HASTIS FUNCTION VALUES (PLUS FIRST AND SECOND DERIVATIVE VALUES) AT
9 C THE COLLOCATION POINTS, AND 4) DEFINING THE INITIAL HASTIS FUNCTION
10 C COEFFICIENTS WHICH DETERMINE THE PIECEWISE POLYNOMIAL WHICH
11 C INTERPOLATES THE USER SUPPLIED (UINIT) INITIAL CONDITION FUNCTION(S)
12 C AT THE COLLOCATION POINTS.
13
14 C K = ORDER OF PIECEWISE POLYNOMIAL SPACE.
15 C A = BASIS FUNCTION VALUES GENERATED BY INITIAL.
16 C MHS = TEMPORARY STORAGE USED TO RETURN INITIAL CONDITION COEFFICIENT
17 C VALUES.
18 C X = USER DEFINED PIECEWISE POLYNOMIAL HREAPPOINTS.
19 C XI = PIECEWISE POLYNOMIAL HREAPPOINT SEQUENCE GENERATED BY INITIAL.
20 C XC = COLLOCATION POINTS GENERATED BY INITIAL.
21 C PW = STORAGE FOR BAND MATRIX USED TO GENERATE INITIAL
22 C COEFFICIENT VALUES.
23 C IPIV = PIVOT INFORMATION FOR LINEAR EQUATION SOLVER DECH-SOLH.
24 C ILEFT = POINTERS TO HREAPPOINT SEQUENCE GENERATED BY INITIAL.
25
26 C PACKAGE ROUTINES CALLED.. HSPLOY,CULPNT,DECH,INTERV,SOLH
27 C USER ROUTINES CALLED.. UINIT
28 C CALLED BY.. PDECOL
29 C FOURTHAN FUNCTIONS USED.. MAX0,MIN0
30
31 C-----
32 C DIMENSION A(K,3,1),MHS(1),X(1),XI(1),XC(1),PW(1),IPIV(1),ILEFT(1)
33 C COMMON /SIZES/ NINT,NORD,NCC,NPI,NPCPTS,NREU,NIER
34 C COMMON/GEAR9/EPJ,RO,ML,MU,IOUM(3),NOM,NOMU
35 C FOR HURNER STABILIZED FLAMES,
36 C THE MASS FLUX FRACTION THROUGH THE LEFT BOUNDARY IS SET EQUAL
37 C TO THE INITIAL COLD BOUNDARY MASS FRACTIONS.
38 C COMMON/TAHCT/RL,CPMX,HO(20),H2O(1-0),H2D(1-20)
39 C COMMON/TAHFN/FM
40 C COMMON/START/NSTART,NHURN,NTHAND,NHD
41 C MFLAG = -2
42 C IEM = 0
43
44 C-----
45 C SET UP THE PIECEWISE POLYNOMIAL SPACE HREAPPOINT SEQUENCE.
46
47 C-----
48 C KPNT = KORD - NCC
49 C DO 10 I=1,KORD
50 C XI(KPNT+I) = X(NINT+I)
51
52 C DO 20 J=1,NINT
53 C XI(1+J) = X(1)
54
55 C-----
56 C SET UP COLLOCATION POINTS ARRAY XC.
57 C-----
58 C CALL CULPNT(K, XC, XI)
59
60 C-----
61 C GENERATE THE ILEFT ARRAY. STORE THE HASTIS FUNCTION VALUES IN THE

```

```

C ARRAY A. THE ARRAY A IS DIMENSIONED (AKORD,3*NCPTS) AND A(K,J,1)
C CONTAINS THE VALUE OF THE (J-1)-ST DERIVATIVE (J = 1,2,3) OF THE K-TH
C NONZERO BASIS FUNCTION (K = 1, ..., AKORD) AT THE 1-TH COLLOCATION
C POINT (I = 1, ..., NCPTS). SET UP A'S FOR INTERPOLATING THE INITIAL
C CONDITIONS AT THE COLLOCATION POINTS. SET THE INTERPOLATION MATRIX
C INTO THE HANDED MATRIX PW.
C-----
DO 35 I=1,NCPTS
  CALL INTERV(XI,NCPTS,XC(I),ILPPT(I),MFLAU)
  CALL BSPLVD(XI,KORD,XC(I),LEP(I),A(I,1,1),3)
35 CONTINUE
CC IF INSTANT = 1.
CC THE STARTING VALUES FOR THE C'S ARE JUST THE INCOMING Y VALUES.
CC RATHER THAN SOLVING THE SYSTEM OF EQUATIONS.
CC THIS PREVENTS THE SYSTEM FROM OVERWORKING TRYING TO GET AROUND)
CC SWAMP CORNERS OR GENERATING NEGATIVE VALUES.
NM=NCPTS-1
DO 50 KPUE=1,NPUE
  DO 40 I=1,NCPTS
    IP=1-(KPUE-1)*NCPTS
    CALL UNITXC(I),HHS(IP),NPUE,KPUE)
    IF INSTANT.EQ.1 GO TO 50
    IF NMURN = 1. WE NEED THE CORRECT VALUE FOR H2O AT THE LEFT
    C BOUNDARY TO GET THE PROPER MASS FLUX FRACTION.
    IF (INTRAN.EQ.1) GO TO 230
    IF (KPUE.GT.1) GO TO 230
    HEAD(1,220)(H2O(L),L=1,NPUE)
    FORMAT(1PRE14.6)
    WRITE(3,225)(R20(L),L=1,NPUE)
    225 FORMAT(2X,5HR20) =,1E6E14.6/
    230 CONTINUE
    DO 30 I=1,NOW1
      PW(I)=0.0
      DO 34 J=1,NM
        I1=I-1
        ICUL=1LEFT(I)-1-1
        JL=HAKO(1,1+2-NCPTS)
        JU=MINO(KORD,KORD+1-2)
        DO 34 J=JL,JU
          J1=1+NCPTS*(ICUL+J-1)
          PW(I+J)=A(J,1,1)
          IF (NMURN.EQ.0) GO TO 34
          IF (I.EQ.1.AND.KPUE.LT.NPUE) PW(I+J1)=
            * A(J,1,1)-R20(KPUE)*A(J,2,1)/FM
      *
    34 CONTINUE
    CC WE WANT THE SPACE DERIVATIVE AT THE RIGHT TO BE ZERO
    ICUL=1LEFT(NCPTS)-NCPTS-1
    DO 39 J=2,KORD
      J1=NM+NCPTS*(ICUL+J-1)
      PW(I+J1)=A(J,2,1)
    39 JP=NCPTS*KPUE
    HHS(IP)=0.0
    CALL DECH(NCPTS,NCPTS,ML,MU,P,0,1,IV,IER)
    IF (IER.NE.0) RETURN
    15=(KPUE-1)*NCPTS+1
    CALL SOLM(NCPTS,NCPTS,ML,MU,P,0,1,15,1,1)*P(IV)
    50 CONTINUE

```

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FIN 4.8.49H

OPT=1 FOUND=\*\*\*// IN ACE

76/76

SUBROUTINE INITIAL

INITIAL 116  
INITIAL 117

RETURN  
END

115

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1      SUBROUTINE COLPNT(X, XC, XT)
2
3      C-----
4      C COLPNT IS CALLED ONLY ONCE BY INITIAL TO DEFINE THE REQUIRED COLLOCA-
5      C TION POINTS WHICH ARE TO BE USED WITH THE USER SELECTED PIECEWISE
6      C POLYNOMIAL SPACE. THE COLLOCATION POINTS ARE CHOSEN SUCH THAT THEY
7      C ARE EITHER THE POINTS AT WHICH THE PIECEWISE POLYNOMIAL SPACE HAS ITS
8      C FUNCTIONS ATTAIN THEIR UNIQUE MAXIMUM VALUES, OR, THE GAUSS-LEGENDRE
9      C QUADRATURE POINTS WITHIN EACH PIECEWISE POLYNOMIAL SPACE SUBINTERVAL,
10     C DEPENDING UPON THE SPACE BEING USED AND THE DESIRE OF THE USER.
11
12     C X = USER DEFINED PIECEWISE POLYNOMIAL BREAKPOINTS.
13     C XC = COLLOCATION POINTS DEFINED BY COLPNT.
14     C XT = PIECEWISE POLYNOMIAL BREAKPOINT SEQUENCE.
15
16     C
17     C PACKAGE ROUTINES CALLED.. HSPLOD, I, IERV
18     C USER ROUTINES CALLED.. NONE
19     C CALLED BY.. INITIAL
20     C FORTRAN FUNCTIONS USED.. NONE
21
22     C-----
23     C DIMENSION RHO(40), XI(1), XC(1), AT(1)
24     C COMMON /SIZES/ NINT,KORD,NCC,NPNT,NCPPTS,NEUN,IQUAD
25     C COMMON /OPTION/ NOGAUS,MAXDER
26     C DATA ILEFT/0/
27
28     C-----
29     C IF THE VARIABLE NOGAUS IN THE COMMON BLOCK /OPTION/ IS SET .EQ. 1,
30     C THE USE OF THE GAUSS-LEGENDRE POINTS IS PROHIBITED FOR ALL CASES.
31     C NOGAUS IS CURRENTLY SET .EQ. 0 BY A DATA STATEMENT IN THE BLOCK DATA.
32     C THE USER MAY CHANGE THIS AS DESIRED.
33
34     C-----
35     C IF ( NCC .NE. 2 .OR. NOGAUS .EQ. 1 ) GO TO 200
36
37     C-----
38     C COMPUTE THE COLLOCATION POINTS TO BE AT THE GAUSS-LEGENDRE POINTS IN
39     C EACH PIECEWISE POLYNOMIAL SPACE SUBINTERVAL. THE ARRAY RHO IS SET TO
40     C CONTAIN THE GAUSS-LEGENDRE POINTS FOR THE STANDARD INTERVAL (-1,1).
41
42     C-----
43     IPTS = KORD - 2
44     GO TO (10,20,30,40,50,60,70,80,90,100,110,120,130,140,150,160,170,
45     * 180),IPTS
46     10 RHO(1) = 0.
47     GO TO 190
48     20 RHO(2) = .577350269189626E-00
49     RHO(1) = - RHO(2)
50     GO TO 190
51     30 RHO(3) = .774596669241483E-00
52     RHO(1) = - RHO(3)
53     RHO(2) = 0.
54     GO TO 190
55     40 RHO(13) = .339981043584456E-00
56     RHO(12) = - RHO(13)
57     RHO(14) = .861136311594053E-00
58     RHO(11) = - RHO(14)
59     GO TO 190
60     50 RHO(4) = .53846941015683E-00
61     RHO(2) = - RHO(4)
62     RHO(5) = .906179845938664E-00
63     RHO(1) = - RHO(5)
64     RHO(3) = 0.

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60  MMU(4) = .238619186083197E-00
   MMU(3) = - MMU(4)
   MMU(5) = .651209386466265E-00
   MMU(2) = - MMU(5)
   MMU(6) = .932469514203152E-00
   MMU(1) = - MMU(6)
60 TO 190
70  MMU(5) = .405845151377397E-00
   MMU(3) = - MMU(5)
   MMU(6) = .741531185599394E-00
   MMU(2) = - MMU(6)
   MMU(7) = .949107912342759E-00
   MMU(1) = - MMU(7)
   MMU(4) = 0.
60 TO 190
80  MMU(5) = .183434642495650E-00
   MMU(4) = - MMU(5)
   MMU(6) = .525532409916329E-00
   MMU(3) = - MMU(6)
   MMU(7) = .79666477413627E-00
   MMU(2) = - MMU(7)
   MMU(8) = .960289856497536E-00
   MMU(1) = - MMU(8)
60 TO 190
90  MMU(5) = .0
   MMU(6) = .324253423403809E-00
   MMU(7) = .613371432700590E-00
   MMU(8) = .836031107326636E-00
   MMU(9) = .968160239507626E-00
   MMU(1) = 1.4
95  MMU(1) = -MMU(10-1)
60 TO 190
100 MMU(6) = .148874338981631E-00
   MMU(7) = .433395394129247E-00
   MMU(8) = .679409568299024E-00
   MMU(9) = .865063366888984E-00
   MMU(10) = .973906528517172E-00
   MMU(1) = 1.5
105 MMU(1) = -MMU(11-1)
60 TO 190
110 MMU(6) = .0
   MMU(7) = .269543155952345E-00
   MMU(8) = .519096129206812E-00
   MMU(9) = .730152005574049E-00
   MMU(10) = .86706259764095E-00
   MMU(11) = .9782246554146057E-00
   MMU(1) = 1.6
115 MMU(1) = -MMU(12-1)
60 TO 190
120 MMU(7) = .125233404511469E-00
   MMU(8) = .347831498998180E-00
   MMU(9) = .547317954284617E-00
   MMU(10) = .769902674194304E-00
   MMU(11) = .904117259370475E-00
   MMU(12) = .981560834246719E-00
   MMU(1) = 1.6

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115      125      WHO(1) = -WHO(13-1)
      GO TO 190
130      WHO( 7) = .0
      WHO( 8) = .23045831555135E-00
      WHO( 9) = .448492751036447E-00
      WHO(10) = .642349339440340E-00
      WHO(11) = .80157409073310E-00
      WHO(12) = .917598395222974E-00
      WHO(13) = .984183054718584E-00
      DO 135 I=1,6
      WHO(1) = -WHO(14-1)
      GO TO 190
135      140      WHO( 8) = .108054948707344E-00
      WHO( 9) = .319112368927890E-00
      WHO(10) = .515244636358154E-00
      WHO(11) = .687292904811685E-00
      WHO(12) = .827201315069765E-00
      WHO(13) = .928434883663574E-00
      WHO(14) = .986283806696812E-00
      DO 145 I=1,7
      WHO(1) = -WHO(15-1)
      GO TO 190
145      150      WHO( 8) = .0
      WHO( 9) = .201194093997435E-00
      WHO(10) = .394151347077563E-00
      WHO(11) = .570972172608539E-00
      WHO(12) = .724417731360170E-00
      WHO(13) = .848206583410427E-00
      WHO(14) = .937273392440706E-00
      WHO(15) = .987992518020485E-00
      DO 155 I=1,7
      WHO(1) = -WHO(16-1)
      GO TO 190
155      160      WHO( 9) = .950125098376374E-01
      WHO(10) = .291603550779259E-00
      WHO(11) = .458016777657227E-00
      WHO(12) = .617876244402644E-00
      WHO(13) = .755404408355003E-00
      WHO(14) = .865631202387832E-00
      WHO(15) = .944575023073233E-00
      WHO(16) = .989400934991650E-00
      DO 165 I=1,8
      WHO(1) = -WHO(17-1)
      GO TO 190
165      170      WHO( 9) = .0
      WHO(10) = .178444141495844E-00
      WHO(11) = .351231763453876E-00
      WHO(12) = .51269053086477E-00
      WHO(13) = .657671154216691E-00
      WHO(14) = .781514003896801E-00
      WHO(15) = .880239153726946E-00
      WHO(16) = .95067542168764E-00
      WHO(17) = .990575675314417E-00
      DO 175 I=1,8
      WHO(1) = -WHO(18-1)
      GO TO 190
175      180      WHO(10) = .867750130417353E-01

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175      RHO(11) = .2518446225691504E-00
      RHO(12) = .411751161462843E-00
      RHO(13) = .55977031073944E-00
      RHO(14) = .691687043060353E-00
      RHO(15) = .803704558972523E-00
      RHO(16) = .892602466497554E-00
      RHO(17) = .955823944571394E-00
      RHO(18) = .991565168420931E-00
      DO 145 I=1,9
140      RHO(I) = -RHO(19-I)
      C-----
      C COMPUTE THE GAUSS-LEGENDRE COLLOCATION POINTS IN EACH SUBINTERVAL.
      C-----
145      DO 195 I=1,NINT
      FAC = ( X(I+1) - X(I) ) * .5
      DO 195 J = 1,IPTS
      KNOT = IPTS * (I-1) + J + 1
195      XC(KNOT) = X(I) + FAC * ( RHO(J) + 1. )
      XC(I) = X(I)
      XC(INCPTS) = X(INIT+1)
      RETURN
      C-----
      C COMPUTE THE COLLOCATION POINTS TO BE AT THE POINTS WHERE THE BASIS
      C FUNCTIONS ATTAIN THEIR MAXIMA. A BISECTION METHOD IS USED TO FIND
      C THE POINTS TO MACHINE PRECISION. THIS PROCESS COULD BE SPEEDED UP
      C BY USING A SECANT METHOD IF DESIRED.
      C-----
200      ITOP = NCPTS - 1
      MFLAG = -2
      XC(1) = X(1)
      XC(INCPTS) = X(INIT+1)
      DO 240 I=2,ITOP
      XOLD = 1.E+20
      XL = XT(I)
      XR = XT(I+KORD)
      XNEW = .5 * (XL + XR)
      IF ( XOLD -EO. XNEW ) GO TO 240
      CALL INTERV(XT,NCPTS,XNEW,ILEFT,MFLAG)
      CALL HSPLVD(XT,KORD,XNEW,ILEFT,RHO,2)
      DO 220 J=1,KORD
      IF ( 1.EO. J + ILEFT - KORD ) GO TO 230
220      CONTINUE
      XVAL = RHO(KORD+J)
      IF ( XVAL -EO. 0.0 ) XR = XNEW
      IF ( XVAL -GT. 0.0 ) XL = XNEW
      IF ( XVAL -LT. 0.0 ) XR = XNEW
      XOLD = XNEW
      GO TO 210
240      XC(I) = XR
      RETURN
      END

```

## SUBROUTINE COLPNT

### CLARK NR. SEVERITY DETAILS DIAGNOSIS OF PROBLEM

[illegible]

04/15/80 11.10.37

FIN 4.0049H

76/76 OPT=1 MURD=00/ 1-ACE

SUBROUTINE CULPNT

CARD NR.	SEVERITY	DETAILS	DIAGNOSIS OF PROBLEM	HIGH	UPPER DIGITS	RETAINED	HUT	SOME	PRECISION	LOST
155	1	CONSTANT TOO LONG.	HIGH	UPPER DIGITS	RETAINED	HUT	SOME	PRECISION	LOST	
161	1	CONSTANT TOO LONG.	HIGH	UPPER DIGITS	RETAINED	HUT	SOME	PRECISION	LOST	
162	1	CONSTANT TOO LONG.	HIGH	UPPER DIGITS	RETAINED	HUT	SOME	PRECISION	LOST	
163	1	CONSTANT TOO LONG.	HIGH	UPPER DIGITS	RETAINED	HUT	SOME	PRECISION	LOST	
164	1	CONSTANT TOO LONG.	HIGH	UPPER DIGITS	RETAINED	HUT	SOME	PRECISION	LOST	
165	1	CONSTANT TOO LONG.	HIGH	UPPER DIGITS	RETAINED	HUT	SOME	PRECISION	LOST	
166	1	CONSTANT TOO LONG.	HIGH	UPPER DIGITS	RETAINED	HUT	SOME	PRECISION	LOST	
167	1	CONSTANT TOO LONG.	HIGH	UPPER DIGITS	RETAINED	HUT	SOME	PRECISION	LOST	
171	1	CONSTANT TOO LONG.	HIGH	UPPER DIGITS	RETAINED	HUT	SOME	PRECISION	LOST	
173	1	CONSTANT TOO LONG.	HIGH	UPPER DIGITS	RETAINED	HUT	SOME	PRECISION	LOST	
174	1	CONSTANT TOO LONG.	HIGH	UPPER DIGITS	RETAINED	HUT	SOME	PRECISION	LOST	
175	1	CONSTANT TOO LONG.	HIGH	UPPER DIGITS	RETAINED	HUT	SOME	PRECISION	LOST	
176	1	CONSTANT TOO LONG.	HIGH	UPPER DIGITS	RETAINED	HUT	SOME	PRECISION	LOST	
177	1	CONSTANT TOO LONG.	HIGH	UPPER DIGITS	RETAINED	HUT	SOME	PRECISION	LOST	
178	1	CONSTANT TOO LONG.	HIGH	UPPER DIGITS	RETAINED	HUT	SOME	PRECISION	LOST	
179	1	CONSTANT TOO LONG.	HIGH	UPPER DIGITS	RETAINED	HUT	SOME	PRECISION	LOST	

```

1 SUBROUTINE HSPLVD ( AT, K, X, ILEFT, VNIXK, NUERIV )
2
3 C THIS SUBROUTINE IS PART OF THE H-SPLINE PACKAGE FOR THE STABLE
4 C EVALUATION OF ANY H-SPLINE BASIS EQUATION ON DERIVATIVE VALUE.
5 C SEE REFERENCE BELOW.
6
7 C CALCULATES THE VALUE AND THE FIRST DERIV-1 DERIVATIVES OF ALL
8 C H-SPLINES WHICH DO NOT VANISH AT X. THE ROUTINE FILLS THE TWO-
9 C DIMENSIONAL ARRAY VNIXK(J,IDERIV), J=1,NUERIV, ... K WITH NONZERO
10 C VALUES OF H-SPLINES OF ORDER K+1-IDERIV, 1, ... 1, BY
11 C REPEATED CALLS TO HSPLVN.
12
13 C XT = PIECEWISE POLYNOMIAL BREAKPOINT SEQUENCE.
14 C K = ORDER OF THE PIECEWISE POLYNOMIAL SPACE.
15 C X = POINT AT WHICH THE H-SPLINE IS TO BE EVALUATED.
16 C ILEFT = POINTER TO THE BREAKPOINT SEQUENCE.
17 C VNIXK = TABLE OF H-SPLINE VALUES AND DERIVATIVES.
18 C NUERIV = DETERMINES NUMBER OF DERIVATIVES TO BE GENERATED.
19
20 C REFERENCE
21
22 C UEMOR, C., PACKAGE FOR CALCULATING WITH H-SPLINES, SIAM J.
23 C NUMER. ANAL., VOL. 14, NO. 3, JUNE 1977, PP. 441-472.
24
25 C PACKAGE ROUTINES CALLED.. HSPLVN
26 C USER ROUTINES CALLED.. NONE
27 C CALLED BY.. COLPNT,INITAL,VALUES
28 C FORTHAN FUNCTIONS USED.. FLOAT,MAX
29
30 DIMENSION XT(1),VNIXK(K,NUERIV)
31 DIMENSION A(20,20)
32 K0 = K + 1 - NUERIV
33 CALL HSPLVN(XT,K0,1,X,ILEFT,VNIXK,NUERIV,NUERIV)
34 IF (NUERIV.LE. 1) GO TO 120
35 IUEIV = NUERIV
36 DO 20 I=2,NUERIV
37 IDEIVM = IUEIV-1
38 DO 10 J=1,NUERIV-K
39 VNIXK(J-1,IDERIVM) = VNIXK(J,IDERIV)
40 IDEIV = IUEIVM
41 CALL HSPLVN(XT,0,X,ILEFT,VNIXK(1,IDERIV),NUERIV)
42 CONTINUE
43 DO 40 J=1,K
44 DO 30 J=1,K
45 A(1,J) = 0.
46 A(1,1) = 1.
47 KMD = K
48 DO 110 M=2,NUERIV
49 KMD = KMD - 1
50 FRMD = FLOAT(KMD)
51 I = ILEFT
52 J = K
53 JPJ = J-1
54 IPRND = 1 + K MD
55 DIFF = XT(IPRND) - AT(1)
56 IF (J.MI.EQ. 0) GO TO 110
57 IF (DIFF.EQ. 0.) GO TO 70

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279

04/15/80 11.10.37

FIN 4.8.498

1-ACE

76/76

SUBROUTINE HSPLVD

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1 SUBROUTINE HSPLVN ( XT, JHIGH, IDEX, X, ILEFT, VNIXX )
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C THIS SUBROUTINE IS PART OF THE H-SPLINE PACKAGE FOR THE STABLE  
C EVALUATION OF ANY R-SPLINE BASIS FUNCTION OR DERIVATIVE VALUE.  
C SEE REFERENCE BELOW.

C CALCULATES THE VALUE OF ALL POSSIBLY NONZERO R-SPLINES AT THE  
C POINT X OF ORDER MAX(JHIGH, (J+1)\*(INDEX-1)) FROM THE BREAKPOINT SE-  
C QUENCE XT. ASSUMING THAT XT(ILEFT) .LE. X .LE. XT(ILEFT+1). THE ROUT-  
C LINE RETURNS THE R-SPLINE VALUES IN THE ONE DIMENSIONAL ARRAY VNIXX.

C FOR DEFINITIONS OF CALLING ARGUMENTS SEE ABOVE AND HSPLVD.

C REFERENCE

C DEHOOP, C. PACKAGE FOR CALCULATING WITH H-SPLINES. SIAM J.  
C NUMER. ANAL. VOL. 14, NO. 3, JUNE 1977, PP. 661-672.

C PACKAGE ROUTINES CALLED.. NONE  
C USER ROUTINES CALLED.. NONE  
C CALLED BY.. HSPLVD  
C FORTRAN FUNCTIONS USED.. NONE

-----

10 J = 1  
VNIXX(1) = 1.  
IF (J .GE. JHIGH) GO TO 40  
20 IPJ = ILEFT+J  
DELTA(IJ) = XT(IPJ) - X  
JMPJ = ILEFT-J+1  
DELTA(J) = X - XT(JMPJ)  
VMPREV = 0.  
JPI = J+1  
30 L=1, J  
JPIML = JPI-L  
VM = VNIXX(L)/(DELTA(IJ) + DELTA(JPIML))  
VNIXX(L) = VM\*DELTA(IJ) + VMPREV  
VMPREV = VM\*DELTA(JPIML)  
VNIXX(JPI) = VMPREV  
J = JPI  
40 RETURN  
END

11.10.37

04/15/80

FTN 4.8.49H

1WACL

76/76

OPT=1

MOUNO=

```

1 SURROUTINE INTERV ( XT, LXT, X, ILEFT, MFLAG )
2 INTERV
3
4 C THIS SURROUTINE IS PART OF THE H-SPLINE PACKAGE FOR THE STAYL
5 C EVALUATION OF ANY H-SPLINE HASITS FUNCTION OR DERIVATIVE VALUE.
6 C SEE REFERENCE BELOW.
7
8 C COMPUTES LARGEST ILEFT IN (1,LXT) SUCH THAT AT(ILEFT) .LE. X. THE
9 C PROGRAM STARTS THE SEARCH FOR ILEFT WITH THE VALUE OF ILEFT THAT WAS
10 C RETURNED AT THE PREVIOUS CALL (AND WAS SAVED IN THE LOCAL VARIABLE
11 C ILO) TO MINIMIZE THE WORK IN THE CASE THAT THE VALUE OF X OR
12 C THIS CALL IS CLOSE TO THE VALUE OF X ON THE PREVIOUS CALL. SHOULD
13 C THIS ASSUMPTION NOT BE VALID, THEN THE PROGRAM LOCATES ILO AND IHI
14 C SUCH THAT AT(ILO) .LE. X .LT. AT(IHI) AND, SINCE THEY ARE FOUND USES
15 C BISECTION TO FIND THE CORRECT VALUE FOR ILEFT. MFLAG IS AN ERROR FLAG.
16 INTERV
17
18 C FOR DEFINITIONS OF CALLING ARGUMENTS SEE ABOVE AND HSPLVD.
19 INTERV
20
21 C REFERENCE
22
23 C DEMODK C. PACKAGE FOR CALCULATING WITH H-SPLINES. SIAM J.
24 NUMER. ANAL., VOL. 14, NO. 3, JUNE 1977, PP. 441-472.
25
26 C PACKAGE ROUTINES CALLED.. NONE
27 C USER ROUTINES CALLED.. NONE
28 C CALLED BY.. COLPNT=INITIAL VALUES
29 C FOURTRAN FUNCTIONS USED.. NONE
30
31 DIMENSION XT(LXT)
32 IF (MFLAG.EQ.-2) ILO = 1
33 IHI = ILO + 1
34 IF (IHI .LT. LXT) GO TO 20
35 IF (X .GE. AT(LXT)) GO TO 110
36 IF (X .LE. AT(1)) GO TO 90
37 ILO = LXT - 1
38 GO TO 21
39
40 20 IF (X .GE. AT(IHI)) GO TO 40.
41 21 IF (X .GE. AT(ILO)) GO TO 100
42
43 C NOW X .LT. AT(IHI). FIND LOWER BOUND.
44
45 30 ISTEP = 1
46 31 IHI = ILO
47 ILO = IHI - ISTEP
48 IF (ILO .LE. 1) GO TO 35
49 IF (X .GE. AT(ILO)) GO TO 50
50 ISTEP = ISTEP*2
51 GO TO 31
52
53 35 ILO = 1
54 IF (X .LT. AT(1)) GO TO 90
55 GO TO 50
56
57 C NOW X .GE. AT(ILO). FIND UPPER BOUND.
58
59 40 ISTEP = 1
60 41 ILO = IHI
61 IHI = ILO + ISTEP
62 IF (IHI .GE. LXT) GO TO 45

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11.10.37

04/15/80

FTN 4.8.49H

SUBROUTINE INTERV 76/76 OPT=1 MOUND=\*\*/ TRACE

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        IF (X .LT. XT(MHI)) GO TO 50
        ISTEP = ISTEP+2
        GO TO 41
    45 IF (X .GE. AT(LXI)) GO TO 110
        IMI = LXT
    C-----
    C NOW AT(ILO) .LE. X .LT. AT(MHI). NAME THE INTERVAL.
    C-----
    50 MIDDLE = (ILO + IMI)/2
        IF (MIDDLE .EQ. ILO) GO TO 100
    C-----
    C NOTE.. IT IS ASSUMED THAT MIDDLE = ILO IN CASE IMI = ILO+1.
    C-----
    70 IF (X .LT. XT(MIDDLE)) GO TO 53
        ILO = MIDDLE
        GO TO 50
    53 IMI = MIDDLE
        GO TO 50
    C-----
    C SET OUTPUT AND RETURN.
    C-----
    90 MFLAG = -1
        ILEFT = 1
        RETURN
    100 MFLAG = 0
        ILEFT = ILO
        RETURN
    110 MFLAG = 1
        ILEFT = LXT
        RETURN
        END
    
```



```

C ERRORS IN Y ARE COMPARED..
C AN ARRAY OF N ELEMENTS. P-RON(1)/10(2) IS THE ESTIMATED
C ONE-STEP ERROR IN Y(1).
C SAVE1,SAVE2,SAVE3 THREE WORKING STORAGE ARRAYS, EACH OF LENGTH 11.
C PW A BLOCK OF LOCATIONS USED FOR THE CHORD ITERATION
C MATRIX. SEE DESCRIPTION IN PDECOL.
C IPIV AN INTEGER ARRAY OF LENGTH N FOR PIVOT INFORMATION.
C ML,MU THE LOWER AND UPPER HALF BANDWIDTHS, RESPECTIVELY, OF
C THE CHORD ITERATION MATRIX. SEE DESCRIPTION IN PDECOL.
C WORK,WORKK WORKING ARRAYS WHICH ARE USED TO PASS APPROPRIATE
C ARRAYS TO OTHER SUBROUTINES.

C PACKAGE ROUTINES CALLED.. COSET,DIFFUN,PSETIM,PES,SOLN
C USER ROUTINES CALLED.. NONE
C CALLED BY.. PDECOL
C FORTRAN FUNCTIONS USED.. ABS,AMAX1,AMINI,FLUAT

C-----
C DIMENSION Y(N0,1),YMAX(N0),ERROR(N0),SAVE1(N0),SAVE2(N0),
C * SAVE3(N0),PW(1),PIV(1),WORK(1),WORKK(1)
C COMMON /SIZES/ NINT,KOH0,NCC,NPIC,NCPIS,NEUN,IQUAD
C COMMON /ISTART/ Iw1,Iw2,Iw3,Iw4,Iw5,Iw6,Iw7,Iw8,Iw9,Iw10,Iw11,
C * Iw12,Iw13,Iw14,Iw15,Iw16,Iw17,Iw18
C COMMON /GEAR1/ T,H,HMIN,HMAX,EPS,KOUNO,N,MF,KFLAG,JSTART
C COMMON /GEAR9/EPSJ,H0,ML,MU,MW,NM1,NOML,NUM,NOWI
C COMMON /GEAR0/ HUSED,NOUSED,NSTEP,NFE,NJE
C COMMON /GEARSS/HMX
C COMMON /OPTION/ NOGAUS,MAXDEX
C COMMON /TARSH/SMALL
C DIMENSION EL(13),TU(4)
C DATA EL(2)/1.0, OLDLU(1.0, TU(1)/0.0, IER/0/
C KFLAG = 0
C TOLD = T
C IF (JSTART.GT. 0) GO TO 200
C IF (JSTART.NE. 0) GO TO 120
C-----
C ON THE FIRST CALL, THE ORDER IS SET TO 1 AND THE INITIAL YDOT IS
C CALCULATED. RMAX IS THE MAXIMUM RATIO BY WHICH H CAN BE INCREASED
C IN A SINGLE STEP. IT IS INITIALLY 1.E4 TO COMPENSATE FOR THE SMALL
C INITIAL H, BUT THEN IS NORMALLY EQUAL TO 10. IF A FAILURE
C OCCURS (IN CORRECTOR CONVERGENCE OR LKOR TEST), RMAX IS SET AT 2
C FOR THE NEXT INCREASE.
C-----
C NO = 1
C IEP = 0
C CALL DIFFUN(T,Y,SAVE1,IER,PW,PIV,WORK,WORKK)
C IF (IER.NE. 0) GO TO 6H5
C DO 110 I = 1,N
C Y(I,2) = H*SAVE1(I)
C METH = MF/10
C L = 2
C MITER = MF - 10*MTH
C L = 2
C LKORH = 3
C RMAX = 1.E+04
C JC = 0.
C CHATE = 1.
C PSOLD = FPS
C-----

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11.10.37

FIN 4.4.44M

PLACE

76/76

SUBROUTINE STIFIN

OPT=1 ROUNDO=\*\*/

```

115 MFOLD = MF
    NOLD = N
    NSTEP = 0
    NSTEPJ = 0
    MFE = 0
    NJE = 1
    IMFT = 3
    GO TO 130

120 -----
    C IF THE CALLER HAS CHANGED METH, CUSET IS CALLED TO SET
    C THE COEFFICIENTS OF THE METHOD. IF THE CALLER HAS CHANGED
    C N, EPS, OR METH, THE CONSTANTS E, EUP, EUP, AND HND MUST BE RESET.
    C E IS A COMPARISON FOR ERRORS OF THE CURRENT ORDER NO. EUP IS
    C TO TEST FOR INCREASING THE ORDER. EUP FOR DECREASING THE ORDER.
    C HND IS USED TO TEST FOR CONVERGENCE OF THE CORRECTOR ITERATES.
    C IF THE CALLER HAS CHANGED M, Y MUST BE RESCALED.
    C IF H ON METH HAS BEEN CHANGED, IDOON IS RESET TO L + 1 TO PREVENT
    C FURTHER CHANGES IN M FOR THAT MANY STEPS.
    -----
120 IF (MF .EQ. MFOLD) GO TO 150
    MEU = METH
    MIO = MITF
    METH = MF/10
    MITER = MF - 10*METH
    MFOLD = MF
    IF (MITER .NE. MIO) IMEVAL = MITER
    IF (METH .EQ. MEU) GO TO 150
    IDOON = L + 1
    IMET = 1
130 CALL CUSET (METH, NU, EL, TU)
    LMAX = MAXDER + 1
    MC = RC*EL(1)/OLDLO
    OLDLO = EL(1)
140 FN = FLOAT(N)
    EUN = FN*(TU(1)*EPS)**2
    E = FN*(TU(2)*EPS)**2
    EUP = FN*(TU(3)*EPS)**2
    HND = FN*(TU(4)*EPS)**2
    GO TO (160, 170, 200), IMET
150 IF ((EPS .EQ. EPSOLD) .AND. (N .EQ. NOLD)) GO TO 160
    EPSOLD = EPS
    NOLD = N
    IMFT = 1
    GO TO 140
160 IF (H .EQ. HOLD) GO TO 200
    H = H/HOLD
    H = HOLD
    IMEDN = 3
    GO TO 175
170 HH = AMAX1(RH, HMIN/ AHS(H))
175 HH = AMIN1(RH, HMAX/ AHS(H)*HMAX)
    HH = AMIN1(RH, HMX/ AHS(H))
    H1 = 1.
    DO 180 J = 2, L
        H1 = H1*HH
    180 180 I = 1, J
    190 Y(I, J) = Y(I, J)*H1
190

```

04/15/80 11.10.37

FIN 4.4449H

SUBROUTINE STIFIN 76/76 OPT=1 MOUND=0.0/ 1-ACE

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175      H = H*RH
      MC = MC*RH
      IDURB = L + 1
      IF (I*EDU .EQ. 0) GO TO 690
      C-----
      C THIS SECTION COMPUTES THE PREDICTED VALUES BY EFFECTIVELY
      C MULTIPLYING THE Y ARRAY BY THE PASCAL TRIANGLE MATRIX.
      C RC IS THE RATIO OF NEW TO OLD VALUES OF THE COEFFICIENT H*EL(1).
      C WHEN MC DIFFERS FROM 1 BY MORE THAN 30 PERCENT, OR THE CALLER HAS
      C CHANGED MITER, IWEAL IS SET TO MITER TO FORCE PW TO BE UPDATED.
      C IN ANY CASE, PW IS UPDATED AT LEAST EVERY 40-TH STEP.
      C PW IS THE CHORD ITERATION MATRIX A = H*EL(1)*(DG/DY).
      C-----
185      200 IF (ABS(RC-1.) .GT. 0.3) IWEAL = MITER
      IF (NSTEP .GE. NSTEPJ*40) IWEAL = MITER
      T = T + H
      DO 210 J1 = 1,N0
      DO 210 J2 = J1,N0
        J = (N0 + J1) - J2
        DO 210 I = 1,N
          210 Y(I,J) = Y(I,J) + Y(I,J+1)
        220 CONTINUE
      IF (IWEAL .LE. 0) GO TO 350
      C IF INDICATED, THE MATRIX PW IS REEVALUATED BEFORE STARTING THE
      C CORRECTOR ITERATION. IWEAL IS SET TO 0 AS AN INDICATOR
      C THAT THIS HAS BEEN DONE. PW IS COMPUTED AND PROCESSED IN PSETIM.
      C-----
200      IWEAL = 0
      MC = 1.
      NJE = NJE + 1
      NSTEPJ = NSTEP
      CUN = -H*EL(1)
      * CALL PSETIM (Y, PW, N0, CON, MITER, IER, WORK(IW1), WORK(IW2),
      * WORK(IW3), WORK(IW4), SAVE2, PIV, YMAX, WORK(IW11), WORK(IW12),
      * WORK(IW13), WORK(IW14), WORK(IW15), WORK, NPITE)
      IF (IER .NE. 0) GO TO 420
      350 CONTINUE
      C-----
      C UP TO 3 CORRECTOR ITERATIONS ARE TAKEN. A CONVERGENCE TEST IS
      C MADE ON THE P. M. S. SUM OF EACH CONNECTION, USING HND, WHICH
      C IS DEPENDENT ON EPS. THE SUM OF THE CONNECTIONS IS ACCUMULATED IN
      C THE VECTOR ERROR(1). THE Y ARRAY IS NOT ALTERED IN THE CORRECTOR
      C LOOP. THE UPDATED Y VECTOR IS STORED TEMPORARILY IN SAVE1. THE
      C UPDATED H*YDOT IS STORED IN SAVE2.
      C-----
      DO 230 I=1,N
        SAVE2(I)=Y(I,2)
        SAVE1(I)=Y(I,1)
      230      ERROR(1)=0.0
      M=0
      N=0
      DO 345 K=1,NPITE
        IST=(K-1)*NCPTS+1
        IEND=K*NCPTS
        HST=(IEND-I)*H0*1.1
        CALL HST(ITEM,SAVE1,SAVE2,SAVE3,DEVID(PITS,WORK(IW1)),WORK,

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230      *      WORK(14)=WORK(15)+WORK(16)+WORK(17)
231      *      WORK(18)=K
232      *      WORK(19)=K
233      *      COMPUTE THE CORRECTION ERROR, K SUM, AND SOLVE THE LINEAR SYSTEM
234      *      WITH THAT AS RIGHT-HAND SIDE AND PW AS COEFFICIENT MATRIX,
235      *      USING THE LU DECOMPOSITION OF PW.
236      *      CALL SOLVINGPTS(NPTS,ML,MU,P,CONST),SAVE(1ST),IPIV(1ST))
237      *      DO 360 I=1ST,IEND
238      *      ERROR(I) = ERROR(I) + SAVE(1)
239      *      D = D + (SAVE(1)/YMAX(I))**2
240      *      SAVE(1) = Y(I,1) + EL(I)*ERROR(I)
241      *      SAVE(2) = Y(I,2) + ERROR(I)
242      *      CONTINUE
243      *      NFE=NFE+1
244      *      IF (M.EQ. 3) GO TO 410
245      *      TEST FOR CONVERGENCE. IF M.GT.0, AN ESTIMATE OF THE CONVERGENCE
246      *      RATE CONSTANT IS STORED IN CRATE, AND THIS IS USED IN THE TEST.
247      *      IF (M.NE. 0) CRATE = AMAX1(.9*CRATE,D/UL)
248      *      IF (ID*AMIN1(1.0E0,2.*CRATE)).LE..ND)GO TO 450
249      *      D1 = D
250      *      M = M + 1
251      *      IF (M.EQ. 3) GO TO 410
252      *      GO TO 360
253      *      THE CORRECTOR ITERATION FAILED TO CONVERGE IN J TRIES.
254      *      THE Y ARRAY IS RETRACTED TO ITS VALUES
255      *      BEFORE PREDICTION, AND H IS REDUCED, IF POSSIBLE. IF NOT, A
256      *      NO-CONVERGENCE EXIT IS TAKEN.
257      *      CONTINUE
258      *      IFEVAL=IFEVAL+1
259      *      T = TOLD
260      *      RMAX = 2.
261      *      DO 430 J1 = 1,N
262      *      J = (N1 + J1) - J2
263      *      DO 430 I = 1,N
264      *      Y(I,J) = Y(I,J) - Y(I,J+1)
265      *      IF (ABS(H) .LE. HMIN*.10001) GO TO 640
266      *      H=0.60
267      *      IFEVAL = 1
268      *      GO TO 170
269      *      THE CORRECTOR HAS CONVERGED. IFEVAL IS SET TO -1 TO SIGNAL
270      *      THAT PW MAY NEED UPDATING ON SUBSEQUENT STEPS. THE ERROR TEST
271      *      IS MADE AND CONTROL PASSES TO STATEMENT 500 IF IT FAILS.
272      *      IFEVAL = -1
273      *      D = 0.
274      *      DO 460 I = 1,M
275      *      D = D + (ERROR(I)/YMAX(I))**2
276      *      IF (D.GT. E) GO TO 500
277      *      AFTER A SUCCESSFUL STEP, UPDATE THE Y ARRAY.
278      *      CONSIDER CHANGING H IF IFEVAL = 1. OTHERWISE DECREASE IFEVAL BY 1.

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```

290      C IF IDOUB IS THEN 1 AND NO .LT. MAADP, THEN ERROR IS SAVED FOR
291      C USE IN A POSSIBLE ORDER INCREASE ON THE NEXT STEP.
292      C IF A CHANGE IN H IS CONSIDERED, AN INCREASE OR DECREASE IN ORDER
293      C TRY ONE IS CONSIDERED ALSO. A CHANGE IN H IS MADE ONLY IF IT IS BY A
294      C FACTOR OF AT LEAST 1.1. IF NOT, IDOUB IS SET GO 3 TO PREVENT
295      C TESTING FOR THAT MANY STEPS.
296      C H IS NOT INCREASED NEAR TOUT.
297      -----
298      KFLAG = 0
299      IMED0 = 0
300      NSTEP = NSTEP + 1
301      HUSED = H
302      NUUSED = NO
303      DO 470 J = 1, L
304      470      Y(I,J) = Y(I,J) + EL(I,J)*ERROR(I)
305      IDOUB=IDOUB*M
306      IF (IDOUB .EQ. 1) GO TO 520
307      IDOUB = IDOUB - 1
308      IF (IDOUB .GT. 1) GO TO 700
309      IF (L .EQ. LMAX) GO TO 700
310      DO 490 I = 1, N
311      490      Y(I,LMAX) = ERROR(I)
312      GO TO 700
313      -----
314      C THE ERROR TEST FAILED. KFLAG KEEPS TRACK OF MULTIPLE FAILURES.
315      C RESTORE T AND THE Y ARRAY TO THEIR PREVIOUS VALUES, AND PREPARE
316      C TO TRY THE STEP AGAIN. COMPUTE THE OPTIMUM STEP SIZE FOR THIS OR
317      C ONE LOWER ORDER.
318      -----
319      500 KFLAG = KFLAG - 1
320      T = TOLD
321      DO 510 J1 = 1, N1
322      510      DO 510 J2 = J1, N1
323      J = (N1 + J1) - J2
324      DO 510 I = 1, N
325      510      Y(I,J) = Y(I,J) - Y(I,J+1)
326      RMAX = 2.
327      IF (ABS(H) .LE. HMIN*1.00001) GO TO 660
328      IF (KFLAG .LE. -3) GO TO 640
329      IMED0 = 2
330      EN02=.5/FLOAT(L)
331      PR2=((O/E)*EN02)*2.0+2.0E-06
332      RH=1.0/PR2
333      GO TO 170
334      -----
335      C REGARDLESS OF THE SUCCESS OR FAILURE OF THE STEP, FACTORS
336      C PH1, PH2, AND PH3 ARE COMPUTED, BY WHICH H COULD BE DIVIDED
337      C AT ORDER NO - 1, ORDER NO, OR ORDER NO + 1, RESPECTIVELY.
338      C IN THE CASE OF FAILURE, PH3 = 1.620 TO AVOID AN ORDER INCREASE.
339      C THE SMALLEST OF THESE IS DETERMINED AND THE NEW ORDER CHOSEN,
340      C ACCORDINGLY. IF THE ORDER IS TO BE INCREASED, WE COMPUTE ONE
341      C ADDITIONAL SCALED DERIVATIVE.
342      -----
343      520 PH3 = 1.6E+20
344      IF (L .EQ. LMAX) GO TO 540

```

```

345      U1 = 0.
      DO 530 I = 1,N
        U1 = 01 + ((FPOW(I) - Y(I,LMAX))/YMAX(I))**2
        ENQ3 = .5/ FLOAT(L+1)
        PM3 = (D1/EUP)**ENQ3*.14 + 1.4*-06
        540 FNO2 = .5/ FLOAT(L)
        PR2 = (D1/E)**ENQ2*.12 + 1.2E-06
        PM1 = 1.E+20
        IF (NO.EQ. 1) GO TO 560
        U = 0.
        DO 550 I = 1,N
          U = 0 + (Y(I,L)/YMAX(I))**2
          FNO1 = .5/ FLOAT(NU)
          PM1 = ((D1/FUN)**ENQ1)*1.3 + 1.3*-06
          560 IF (PR2.LE. PM3) GO TO 570
          IF (PR3.LT. PM1) GO TO 540
          GO TO 540
          570 IF (PR2.GT. PM1) GO TO 540
          NEWO = NU
          RM = 1./PR2
          GO TO 620
          580 NEWO = NC - 1
          RM = 1./PM1
          GO TO 620
          590 NEWO = L
          RM = 1./PR3
          IF (RM.LT. 1.1) GO TO 610
          DO 600 I = 1,N
            600 Y(I,NEWO+1) = FNOH(I)*EL(L)/ FLOAT(L)
            GO TO 630
          610 IOUNT=3
          GO TO 700
          620 IF ((KFLAG.EQ. 0) .AND. (RM.LT. 1.1)) GO TO 610
          C-----
          C IF THERE IS A CHANGE OF ORDER, RESET NO, L, AND THE COEFFICIENTS.
          C IN ANY CASE H IS RESET ACCORDING TO RM AND THE Y ARRAY IS RESCALED.
          C THEN EXIT FROM 690 IF THE STEP WAS OK, OR REDO THE STEP OTHERWISE.
          C-----
          IF (NEWO.EQ. NU) GO TO 170
          630 NU = NEWO
          L = NO + 1
          PMT = 2
          GO TO 130
          C-----
          C CONTROL REACHES THIS SECTION IF 3 OR MORE FAILURES HAVE OCCURRED.
          C IT IS ASSUMED THAT THE DERIVATIVES THAT HAVE ACCUMULATED IN THE
          C Y ARRAY HAVE ERRORS OF THE WRONG ORDER. REDUCE THE FIRST
          C DERIVATIVE IS RECOMPUTED, AND THE ORDER IS SET TO 1. THEN
          C H IS REDUCED BY A FACTOR OF 10, AND THE STEP IS RETRIED.
          C AFTER A TOTAL OF 7 FAILURES, AN EXIT IS TAKEN WITH KFLAG = -2.
          C-----
          640 IF (KFLAG.EQ. -7) GO TO 570
          RM = .1
          PM = AMAX1 (RMIN/ ABS(H),RM)
          H = PM*H
          PM = 0
          CALL DIFFUN (NO, I, Y, SAVF1, IPR, PM, PMT, WERR, IWORK)
          650 IF (KFLAG.EQ. -2) GO TO 570
          660 IF (KFLAG.EQ. -1) GO TO 570
          670 IF (KFLAG.EQ. -3) GO TO 570
          680 IF (KFLAG.EQ. -4) GO TO 570
          690 IF (KFLAG.EQ. -5) GO TO 570
          700 IF (KFLAG.EQ. -6) GO TO 570
          710 IF (KFLAG.EQ. -7) GO TO 570
          720 IF (KFLAG.EQ. -8) GO TO 570
          730 IF (KFLAG.EQ. -9) GO TO 570
          740 IF (KFLAG.EQ. -10) GO TO 570
          750 IF (KFLAG.EQ. -11) GO TO 570
          760 IF (KFLAG.EQ. -12) GO TO 570
          770 IF (KFLAG.EQ. -13) GO TO 570
          780 IF (KFLAG.EQ. -14) GO TO 570
          790 IF (KFLAG.EQ. -15) GO TO 570
          800 IF (KFLAG.EQ. -16) GO TO 570
          810 IF (KFLAG.EQ. -17) GO TO 570
          820 IF (KFLAG.EQ. -18) GO TO 570
          830 IF (KFLAG.EQ. -19) GO TO 570
          840 IF (KFLAG.EQ. -20) GO TO 570
          850 IF (KFLAG.EQ. -21) GO TO 570
          860 IF (KFLAG.EQ. -22) GO TO 570
          870 IF (KFLAG.EQ. -23) GO TO 570
          880 IF (KFLAG.EQ. -24) GO TO 570
          890 IF (KFLAG.EQ. -25) GO TO 570
          900 IF (KFLAG.EQ. -26) GO TO 570
          910 IF (KFLAG.EQ. -27) GO TO 570
          920 IF (KFLAG.EQ. -28) GO TO 570
          930 IF (KFLAG.EQ. -29) GO TO 570
          940 IF (KFLAG.EQ. -30) GO TO 570
          950 IF (KFLAG.EQ. -31) GO TO 570
          960 IF (KFLAG.EQ. -32) GO TO 570
          970 IF (KFLAG.EQ. -33) GO TO 570
          980 IF (KFLAG.EQ. -34) GO TO 570
          990 IF (KFLAG.EQ. -35) GO TO 570
          1000 IF (KFLAG.EQ. -36) GO TO 570
          1010 IF (KFLAG.EQ. -37) GO TO 570
          1020 IF (KFLAG.EQ. -38) GO TO 570
          1030 IF (KFLAG.EQ. -39) GO TO 570
          1040 IF (KFLAG.EQ. -40) GO TO 570
          1050 IF (KFLAG.EQ. -41) GO TO 570
          1060 IF (KFLAG.EQ. -42) GO TO 570
          1070 IF (KFLAG.EQ. -43) GO TO 570
          1080 IF (KFLAG.EQ. -44) GO TO 570
          1090 IF (KFLAG.EQ. -45) GO TO 570
          1100 IF (KFLAG.EQ. -46) GO TO 570
          1110 IF (KFLAG.EQ. -47) GO TO 570
          1120 IF (KFLAG.EQ. -48) GO TO 570
          1130 IF (KFLAG.EQ. -49) GO TO 570
          1140 IF (KFLAG.EQ. -50) GO TO 570
          1150 IF (KFLAG.EQ. -51) GO TO 570
          1160 IF (KFLAG.EQ. -52) GO TO 570
          1170 IF (KFLAG.EQ. -53) GO TO 570
          1180 IF (KFLAG.EQ. -54) GO TO 570
          1190 IF (KFLAG.EQ. -55) GO TO 570
          1200 IF (KFLAG.EQ. -56) GO TO 570
          1210 IF (KFLAG.EQ. -57) GO TO 570
          1220 IF (KFLAG.EQ. -58) GO TO 570
          1230 IF (KFLAG.EQ. -59) GO TO 570
          1240 IF (KFLAG.EQ. -60) GO TO 570
          1250 IF (KFLAG.EQ. -61) GO TO 570
          1260 IF (KFLAG.EQ. -62) GO TO 570
          1270 IF (KFLAG.EQ. -63) GO TO 570
          1280 IF (KFLAG.EQ. -64) GO TO 570
          1290 IF (KFLAG.EQ. -65) GO TO 570
          1300 IF (KFLAG.EQ. -66) GO TO 570
          1310 IF (KFLAG.EQ. -67) GO TO 570
          1320 IF (KFLAG.EQ. -68) GO TO 570
          1330 IF (KFLAG.EQ. -69) GO TO 570
          1340 IF (KFLAG.EQ. -70) GO TO 570
          1350 IF (KFLAG.EQ. -71) GO TO 570
          1360 IF (KFLAG.EQ. -72) GO TO 570
          1370 IF (KFLAG.EQ. -73) GO TO 570
          1380 IF (KFLAG.EQ. -74) GO TO 570
          1390 IF (KFLAG.EQ. -75) GO TO 570
          1400 IF (KFLAG.EQ. -76) GO TO 570
          1410 IF (KFLAG.EQ. -77) GO TO 570
          1420 IF (KFLAG.EQ. -78) GO TO 570
          1430 IF (KFLAG.EQ. -79) GO TO 570
          1440 IF (KFLAG.EQ. -80) GO TO 570
          1450 IF (KFLAG.EQ. -81) GO TO 570
          1460 IF (KFLAG.EQ. -82) GO TO 570
          1470 IF (KFLAG.EQ. -83) GO TO 570
          1480 IF (KFLAG.EQ. -84) GO TO 570
          1490 IF (KFLAG.EQ. -85) GO TO 570
          1500 IF (KFLAG.EQ. -86) GO TO 570
          1510 IF (KFLAG.EQ. -87) GO TO 570
          1520 IF (KFLAG.EQ. -88) GO TO 570
          1530 IF (KFLAG.EQ. -89) GO TO 570
          1540 IF (KFLAG.EQ. -90) GO TO 570
          1550 IF (KFLAG.EQ. -91) GO TO 570
          1560 IF (KFLAG.EQ. -92) GO TO 570
          1570 IF (KFLAG.EQ. -93) GO TO 570
          1580 IF (KFLAG.EQ. -94) GO TO 570
          1590 IF (KFLAG.EQ. -95) GO TO 570
          1600 IF (KFLAG.EQ. -96) GO TO 570
          1610 IF (KFLAG.EQ. -97) GO TO 570
          1620 IF (KFLAG.EQ. -98) GO TO 570
          1630 IF (KFLAG.EQ. -99) GO TO 570
          1640 IF (KFLAG.EQ. -100) GO TO 570
          1650 IF (KFLAG.EQ. -101) GO TO 570
          1660 IF (KFLAG.EQ. -102) GO TO 570
          1670 IF (KFLAG.EQ. -103) GO TO 570
          1680 IF (KFLAG.EQ. -104) GO TO 570
          1690 IF (KFLAG.EQ. -105) GO TO 570
          1700 IF (KFLAG.EQ. -106) GO TO 570
          1710 IF (KFLAG.EQ. -107) GO TO 570
          1720 IF (KFLAG.EQ. -108) GO TO 570
          1730 IF (KFLAG.EQ. -109) GO TO 570
          1740 IF (KFLAG.EQ. -110) GO TO 570
          1750 IF (KFLAG.EQ. -111) GO TO 570
          1760 IF (KFLAG.EQ. -112) GO TO 570
          1770 IF (KFLAG.EQ. -113) GO TO 570
          1780 IF (KFLAG.EQ. -114) GO TO 570
          1790 IF (KFLAG.EQ. -115) GO TO 570
          1800 IF (KFLAG.EQ. -116) GO TO 570
          1810 IF (KFLAG.EQ. -117) GO TO 570
          1820 IF (KFLAG.EQ. -118) GO TO 570
          1830 IF (KFLAG.EQ. -119) GO TO 570
          1840 IF (KFLAG.EQ. -120) GO TO 570
          1850 IF (KFLAG.EQ. -121) GO TO 570
          1860 IF (KFLAG.EQ. -122) GO TO 570
          1870 IF (KFLAG.EQ. -123) GO TO 570
          1880 IF (KFLAG.EQ. -124) GO TO 570
          1890 IF (KFLAG.EQ. -125) GO TO 570
          1900 IF (KFLAG.EQ. -126) GO TO 570
          1910 IF (KFLAG.EQ. -127) GO TO 570
          1920 IF (KFLAG.EQ. -128) GO TO 570
          1930 IF (KFLAG.EQ. -129) GO TO 570
          1940 IF (KFLAG.EQ. -130) GO TO 570
          1950 IF (KFLAG.EQ. -131) GO TO 570
          1960 IF (KFLAG.EQ. -132) GO TO 570
          1970 IF (KFLAG.EQ. -133) GO TO 570
          1980 IF (KFLAG.EQ. -134) GO TO 570
          1990 IF (KFLAG.EQ. -135) GO TO 570
          2000 IF (KFLAG.EQ. -136) GO TO 570
          2010 IF (KFLAG.EQ. -137) GO TO 570
          2020 IF (KFLAG.EQ. -138) GO TO 570
          2030 IF (KFLAG.EQ. -139) GO TO 570
          2040 IF (KFLAG.EQ. -140) GO TO 570
          2050 IF (KFLAG.EQ. -141) GO TO 570
          2060 IF (KFLAG.EQ. -142) GO TO 570
          2070 IF (KFLAG.EQ. -143) GO TO 570
          2080 IF (KFLAG.EQ. -144) GO TO 570
          2090 IF (KFLAG.EQ. -145) GO TO 570
          2100 IF (KFLAG.EQ. -146) GO TO 570
          2110 IF (KFLAG.EQ. -147) GO TO 570
          2120 IF (KFLAG.EQ. -148) GO TO 570
          2130 IF (KFLAG.EQ. -149) GO TO 570
          2140 IF (KFLAG.EQ. -150) GO TO 570
          2150 IF (KFLAG.EQ. -151) GO TO 570
          2160 IF (KFLAG.EQ. -152) GO TO 570
          2170 IF (KFLAG.EQ. -153) GO TO 570
          2180 IF (KFLAG.EQ. -154) GO TO 570
          2190 IF (KFLAG.EQ. -155) GO TO 570
          2200 IF (KFLAG.EQ. -156) GO TO 570
          2210 IF (KFLAG.EQ. -157) GO TO 570
          2220 IF (KFLAG.EQ. -158) GO TO 570
          2230 IF (KFLAG.EQ. -159) GO TO 570
          2240 IF (KFLAG.EQ. -160) GO TO 570
          2250 IF (KFLAG.EQ. -161) GO TO 570
          2260 IF (KFLAG.EQ. -162) GO TO 570
          2270 IF (KFLAG.EQ. -163) GO TO 570
          2280 IF (KFLAG.EQ. -164) GO TO 570
          2290 IF (KFLAG.EQ. -165) GO TO 570
          2300 IF (KFLAG.EQ. -166) GO TO 570
          2310 IF (KFLAG.EQ. -167) GO TO 570
          2320 IF (KFLAG.EQ. -168) GO TO 570
          2330 IF (KFLAG.EQ. -169) GO TO 570
          2340 IF (KFLAG.EQ. -170) GO TO 570
          2350 IF (KFLAG.EQ. -171) GO TO 570
          2360 IF (KFLAG.EQ. -172) GO TO 570
          2370 IF (KFLAG.EQ. -173) GO TO 570
          2380 IF (KFLAG.EQ. -174) GO TO 570
          2390 IF (KFLAG.EQ. -175) GO TO 570
          2400 IF (KFLAG.EQ. -176) GO TO 570
          2410 IF (KFLAG.EQ. -177) GO TO 570
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          2440 IF (KFLAG.EQ. -180) GO TO 570
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          2470 IF (KFLAG.EQ. -183) GO TO 570
          2480 IF (KFLAG.EQ. -184) GO TO 570
          2490 IF (KFLAG.EQ. -185) GO TO 570
          2500 IF (KFLAG.EQ. -186) GO TO 570
          2510 IF (KFLAG.EQ. -187) GO TO 570
          2520 IF (KFLAG.EQ. -188) GO TO 570
          2530 IF (KFLAG.EQ. -189) GO TO 570
          2540 IF (KFLAG.EQ. -190) GO TO 570
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          2590 IF (KFLAG.EQ. -195) GO TO 570
          2600 IF (KFLAG.EQ. -196) GO TO 570
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          2640 IF (KFLAG.EQ. -200) GO TO 570
          2650 IF (KFLAG.EQ. -201) GO TO 570
          2660 IF (KFLAG.EQ. -202) GO TO 570
          2670 IF (KFLAG.EQ. -203) GO TO 570
          2680 IF (KFLAG.EQ. -204) GO TO 570
          2690 IF (KFLAG.EQ. -205) GO TO 570
          2700 IF (KFLAG.EQ. -206) GO TO 570
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          2740 IF (KFLAG.EQ. -210) GO TO 570
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          2770 IF (KFLAG.EQ. -213) GO TO 570
          2780 IF (KFLAG.EQ. -214) GO TO 570
          2790 IF (KFLAG.EQ. -215) GO TO 570
          2800 IF (KFLAG.EQ. -216) GO TO 570
          2810 IF (KFLAG.EQ. -217) GO TO 570
          2820 IF (KFLAG.EQ. -218) GO TO 570
          2830 IF (KFLAG.EQ. -219) GO TO 570
          2840 IF (KFLAG.EQ. -220) GO TO 570
          2850 IF (KFLAG.EQ. -221) GO TO 570
          2860 IF (KFLAG.EQ. -222) GO TO 570
          2870 IF (KFLAG.EQ. -223) GO TO 570
          2880 IF (KFLAG.EQ. -224) GO TO 570
          2890 IF (KFLAG.EQ. -225) GO TO 570
          2900 IF (KFLAG.EQ. -226) GO TO 570
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          2920 IF (KFLAG.EQ. -228) GO TO 570
          2930 IF (KFLAG.EQ. -229) GO TO 570
          2940 IF (KFLAG.EQ. -230) GO TO 570
          2950 IF (KFLAG.EQ. -231) GO TO 570
          2960 IF (KFLAG.EQ. -232) GO TO 570
          2970 IF (KFLAG.EQ. -233) GO TO 570
          2980 IF (KFLAG.EQ. -234) GO TO 570
          2990 IF (KFLAG.EQ. -235) GO TO 570
          3000 IF (KFLAG.EQ. -236) GO TO 570
          3010 IF (KFLAG.EQ. -237) GO TO 570
          3020 IF (KFLAG.EQ. -238) GO TO 570
          3030 IF (KFLAG.EQ. -239) GO TO 570
          3040 IF (KFLAG.EQ. -240) GO TO 570
          3050 IF (KFLAG.EQ. -241) GO TO 570
          3060 IF (KFLAG.EQ. -242) GO TO 570
          3070 IF (KFLAG.EQ. -243) GO TO 570
          3080 IF (KFLAG.EQ. -244) GO TO 570
          3090 IF (KFLAG.EQ. -245) GO TO 570
          3100 IF (KFLAG.EQ. -246) GO TO 570
          3110 IF (KFLAG.EQ. -247) GO TO 570
          3120 IF (KFLAG.EQ. -248) GO TO 570
          3130 IF (KFLAG.EQ. -249) GO TO 570
          3140 IF (KFLAG.EQ. -250) GO TO 570
          3150 IF (KFLAG.EQ. -251) GO TO 570
          3160 IF (KFLAG.EQ. -252) GO TO 570
          3170 IF (KFLAG.EQ. -253) GO TO 570
          3180 IF (KFLAG.EQ. -254) GO TO 570
          3190 IF (KFLAG.EQ. -255) GO TO 570
          3200 IF (KFLAG.EQ. -256) GO TO 570
          3210 IF (KFLAG.EQ. -257) GO TO 570
          3220 IF (KFLAG.EQ. -258) GO TO 570
          3230 IF (KFLAG.EQ. -259) GO TO 570
          3240 IF (KFLAG.EQ. -260) GO TO 570
          3250 IF (KFLAG.EQ. -261) GO TO 570
          3260 IF (KFLAG.EQ. -262) GO TO 570
          3270 IF (KFLAG.EQ. -263) GO TO 570
          3280 IF (KFLAG.EQ. -264) GO TO 570
          3290 IF (KFLAG.EQ. -265) GO TO 570
          3300 IF (KFLAG.EQ. -266) GO TO 570
          3310 IF (KFLAG.EQ. -267) GO TO 570
          3320 IF (KFLAG.EQ. -268) GO TO 570
          3330 IF (KFLAG.EQ. -269) GO TO 570
          3340 IF (KFLAG.EQ. -270) GO TO 570
          3350 IF (KFLAG.EQ. -271) GO TO 570
          3360 IF (KFLAG.EQ. -272) GO TO 570
          3370 IF (KFLAG.EQ. -273) GO TO 570
          3380 IF (KFLAG.EQ. -274) GO TO 570
          3390 IF (KFLAG.EQ. -275) GO TO 570
          3400 IF (KFLAG.EQ. -276) GO TO 570
          3410 IF (KFLAG.EQ. -277) GO TO 570
          3420 IF (KFLAG.EQ. -278) GO TO 570
          3430 IF (KFLAG.EQ. -279) GO TO 570
          3440 IF (KFLAG.EQ. -280) GO TO 570
          3450 IF (KFLAG.EQ. -281) GO TO 570
          3460 IF (KFLAG.EQ. -282) GO TO 570
          3470 IF (KFLAG.EQ. -283) GO TO 570
          3480 IF (KFLAG.EQ. -284) GO TO 570
          3490 IF (KFLAG.EQ. -285) GO TO 570
          3500 IF (KFLAG.EQ. -286) GO TO 570
          3510 IF (KFLAG.EQ. -287) GO TO 570
          3520 IF (KFLAG.EQ. -288) GO TO 570
          3530 IF (KFLAG.EQ. -289) GO TO 570
          3540 IF (KFLAG.EQ. -290) GO TO 570
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          3560 IF (KFLAG.EQ. -292) GO TO 570
          3570 IF (KFLAG.EQ. -293) GO TO 570
          3580 IF (KFLAG.EQ. -294) GO TO 570
          3590 IF (KFLAG.EQ. -295) GO TO 570
          3600 IF (KFLAG.EQ. -296) GO TO 570
          3610 IF (KFLAG.EQ. -297) GO TO 570
          3620 IF (KFLAG.EQ. -298) GO TO 570
          3630 IF (KFLAG.EQ. -299) GO TO 570
          3640 IF (KFLAG.EQ. -300) GO TO 570
          3650 IF (KFLAG.EQ. -301) GO TO 570
          3660 IF (KFLAG.EQ. -302) GO TO 570
          3670 IF (KFLAG.EQ. -303) GO TO 570
          3680 IF (KFLAG.EQ. -304) GO TO 570
          3690 IF (KFLAG.EQ. -305) GO TO 570
          3700 IF (KFLAG.EQ. -306) GO TO 570
          3710 IF (KFLAG.EQ. -307) GO TO 570
          3720 IF (KFLAG.EQ. -308) GO TO 570
          3730 IF (KFLAG.EQ. -309) GO TO 570
          3740 IF (KFLAG.EQ. -310) GO TO 570
          3750 IF (KFLAG.EQ. -311) GO TO 570
          3760 IF (KFLAG.EQ. -312) GO TO 570
          3770 IF (KFLAG.EQ. -313) GO TO 570
          3780 IF (KFLAG.EQ. -314) GO TO 570
          3790 IF (KFLAG.EQ. -315) GO TO 570
          3800 IF (KFLAG.EQ. -316) GO TO 570
          3810 IF (KFLAG.EQ. -317) GO TO 570
          3820 IF (KFLAG.EQ. -318) GO TO 570
          3830 IF (KFLAG.EQ. -319) GO TO 570
          3840 IF (KFLAG.EQ. -320) GO TO 570
          3850 IF (KFLAG.EQ. -321) GO TO 570
          3860 IF (KFLAG.EQ. -322) GO TO 570
          3870 IF (KFLAG.EQ. -323) GO TO 570
          3880 IF (KFLAG.EQ. -324) GO TO 570
          3890 IF (KFLAG.EQ. -325) GO TO 570
          3900 IF (KFLAG.EQ. -326) GO TO 570
          3910 IF (KFLAG.EQ. -327) GO TO 570
          3920 IF (KFLAG.EQ. -328) GO TO 570
          3930 IF (KFLAG.EQ. -329) GO TO 570
          3940 IF (KFLAG.EQ. -330) GO TO 570
          3950 IF (KFLAG.EQ. -331) GO TO 570
          3960 IF (KFLAG.EQ. -332) GO TO 570
          3970 IF (KFLAG.EQ. -333) GO TO 570
          3980 IF (KFLAG.EQ. -334) GO TO 570
          3990 IF (KFLAG.EQ. -335) GO TO 570
          4000 IF (KFLAG.EQ. -336) GO TO 570
          4010 IF (KFLAG.EQ. -337) GO TO 570
          4020 IF (KFLAG.EQ. -338) GO TO 570
          4030 IF (KFLAG.EQ. -339) GO TO 570
          4040 IF (KFLAG.EQ. -340) GO TO 570
          4050 IF (KFLAG.EQ. -341) GO TO 570
          4060 IF (KFLAG.EQ. -342) GO TO 570
          4070 IF (KFLAG.EQ. -343) GO TO 570
          4080 IF (KFLAG.EQ. -344) GO TO 570
          4090 IF (KFLAG.EQ. -345) GO TO 570
          4100 IF (KFLAG.EQ. -346) GO TO 570
          4110 IF (KFLAG.EQ. -347) GO TO 570
          4120 IF (KFLAG.EQ. -348) GO TO 570
          4130 IF (KFLAG.EQ. -349) GO TO 570
          4140 IF (KFLAG.EQ. -350) GO TO 570
          4150 IF (KFLAG.EQ. -351) GO TO 570
          4160 IF (KFLAG.EQ. -352) GO TO 570
          4170 IF (KFLAG.EQ. -353) GO TO 570
          4180 IF (KFLAG.EQ. -354) GO TO 570
          4190 IF (KFLAG.EQ. -355) GO TO 570
          4200 IF (KFLAG.EQ. -356) GO TO 570
          4210 IF (KFLAG.EQ. -357) GO TO 570
          4220 IF (KFLAG.EQ. -358) GO TO 570
          4230 IF (KFLAG.EQ. -359) GO TO 570
          4240 IF (KFLAG.EQ. -360) GO TO 570
          4250 IF (KFLAG.EQ. -361) GO TO 570
          4260 IF (KFLAG.EQ. -362) GO TO 570
          4270 IF (KFLAG.EQ. -363) GO TO 570
          4280 IF (KFLAG.EQ. -364) GO TO 570
          4290 IF (KFLAG.EQ. -365) GO TO 570
          4300 IF (KFLAG.EQ. -366) GO TO 570
          4310 IF (KFLAG.EQ. -367) GO TO 570
          4320 IF (KFLAG.EQ. -368) GO TO 570
          4330 IF (KFLAG.EQ. -369) GO TO 570
          4340 IF (KFLAG.EQ. -370) GO TO 570
          4350 IF (KFLAG.EQ. -371) GO TO 570
          4360 IF (KFLAG.EQ. -372) GO TO 570
          4370 IF (KFLAG.EQ. -373) GO TO 570
          4380 IF (KFLAG.EQ. -374) GO TO 570
          4390 IF (KFLAG.EQ. -375) GO TO 570
          4400 IF (KFLAG.EQ. -376) GO TO 570
          4410 IF (KFLAG.EQ. -377) GO TO 570
          4420 IF (KFLAG.EQ. -378) GO TO 570
          4430 IF (KFLAG.EQ. -379) GO TO 570
          4440 IF (KFLAG.EQ. -380) GO TO 570
          4450 IF (KFLAG.EQ. -381) GO TO 570
          4460 IF (KFLAG.EQ. -382) GO TO 570
          4470 IF (KFLAG.EQ. -383) GO TO 570
          4480 IF (KFLAG.EQ. -384) GO TO 570
          4490 IF (KFLAG.EQ. -385) GO TO 570
          4500 IF (KFLAG.EQ. -386) GO TO 570
          4510 IF (KFLAG.EQ. -387) GO TO 570
          4520 IF (KFLAG.EQ. -388) GO TO 570
          4530 IF (KFLAG.EQ. -389) GO TO 570
          4540 IF (KFLAG.EQ. -390) GO TO 570
          4550 IF (KFLAG.EQ. -391) GO TO 570
          4560 IF (KFLAG.EQ. -392) GO TO 570
          4570 IF (KFLAG.EQ. -393) GO TO 570
          4580 IF (KFLAG.EQ. -394) GO TO 570
          4590 IF (KFLAG.EQ. -395) GO TO 570
          4600 IF (KFLAG.EQ. -396) GO TO 570
          4610 IF (KFLAG.EQ. -397) GO TO 570
          4620 IF (KFLAG.EQ. -398) GO TO 570
          4630 IF (KFLAG.EQ. -399) GO TO 570
          4640 IF (KFLAG.EQ. -400) GO TO 570
          4650 IF (KFLAG.EQ. -401) GO TO 570
          4660 IF (KFLAG.EQ. -402) GO TO 570
          4670 IF (KFLAG.EQ. -403) GO TO 570
          4680 IF (KFLAG.EQ. -404) GO TO 570
          4690 IF (KFLAG.EQ. -405) GO TO 570
          4700 IF (KFLAG.EQ. -406) GO TO 570
          4710 IF (KFLAG.EQ. -407) GO TO 570
          4720 IF (KFLAG.EQ. -408) GO TO 570
          4730 IF (KFLAG.EQ. -409) GO TO 570
          4740 IF (KFLAG.EQ. -410) GO TO 570
          4750 IF (KFLAG.EQ. -411) GO TO 570
          4760 IF (KFLAG.EQ. -412) GO TO 570
          4770 IF (KFLAG.EQ. -413) GO TO 570
          4780 IF (KFLAG.EQ. -414) GO TO 570
          4790 IF (KFLAG.EQ. -415) GO TO 570
          4800 IF (KFLAG.EQ. -416) GO TO 570
          4810 IF (KFLAG.EQ. -417) GO TO 570
          4820 IF (KFLAG.EQ. -418) GO TO 570
          4830 IF (KFLAG.EQ. -419) GO TO 570
          4840 IF (KFLAG.EQ. -420) GO TO 570
          4850 IF (KFLAG.EQ. -421) GO TO 570
          4860 IF (KFLAG.EQ. -422) GO TO 570
          4870 IF (KFLAG.EQ. -423) GO TO 570
          4880 IF (KFLAG.EQ. -424) GO TO 570
          4890 IF (KFLAG.EQ. -425) GO TO 570
          4900 IF (KFLAG.EQ. -426) GO TO 570
          4910 IF (KFLAG.EQ. -427) GO TO 570
          4920 IF (KFLAG.EQ. -428) GO TO 570
          4930 IF (KFLAG.EQ. -429) GO TO 570
          4940 IF (KFLAG.EQ. -430) GO TO 570
          4950 IF (KFLAG.EQ. -431) GO TO 570
          4960 IF (KFLAG.EQ. -432) GO TO 570
          4970 IF (KFLAG.EQ. -433) GO TO 570
          4980 IF (KFLAG.EQ. -434) GO TO 570
          4990 IF (KFLAG.EQ. -435) GO TO 570
          5000 IF (KFLAG.EQ. -436) GO TO 570
          5010 IF (KFLAG.EQ. -437) GO TO 570
          5020 IF (KFLAG.EQ. -438) GO TO 570
          5030 IF (KFLAG.EQ. -439) GO TO 570
          5040 IF (KFLAG.EQ. -440) GO TO 570
          5050 IF (KFLAG.EQ. -441) GO TO 570
          5060 IF (KFLAG.EQ. -442) GO TO 570
          5070 IF (KFLAG.EQ. -443) GO TO 570
          5080 IF (KFLAG.EQ. -444) GO TO 570
          5090 IF (KFLAG.EQ. -445) GO TO 570
          5100 IF (KFLAG.EQ. -446) GO TO 570
          5110 IF (KFLAG.EQ. -447) GO TO 570
          5120 IF (KFLAG.EQ. -448) GO TO 570
          5130 IF (KFLAG.EQ. -449) GO TO 570
          5140 IF (KFLAG.EQ. -450) GO TO 570
          5150 IF (KFLAG.EQ. -451) GO TO 570
          5160 IF (KFLAG.EQ. -452) GO TO 570
          5170 IF (KFLAG.EQ. -453) GO TO 570
          5180 IF (KFLAG.EQ. -454) GO TO 570
          5190 IF (KFLAG.EQ. -455) GO TO 570
          5200 IF (KFLAG.EQ. -456) GO TO 570
          5210 IF (KFLAG.EQ. -457) GO TO 570
          5220 IF (KFLAG.EQ. -458
```

04/15/90 11.10.37

FTN 4.8+498

SUBROUTINE STIFIN 76/76 OPT=1 ROUNDO=0.0/ TRACE

```

400      IF (IER.NE.0) GO TO 685
        NJE = NJE + 1
        DO 650 I = 1,N
          650 Y(I,2) = H*SAVEI(1)
          IMEVAL = MITER
          IDOUR=3
          IF (NU.EQ.1) GO TO 200
          NU = 1
          L = 2
          IMET = 3
          GO TO 130
        C-----
        C ALL RETURNS ARE MADE THROUGH THIS SECTION. H IS SAVED IN HOLD
        C TO ALLOW THE CALLER TO CHANGE H ON THE NEXT STEP.
        C-----
        660 KFLAG = -1
          GO TO 700
        670 KFLAG = -2
          GO TO 700
        680 KFLAG = -3
          GO TO 700
        685 KFLAG = -4
          GO TO 700
        690 HMAX=10.0
          700 HOLD = H
          JSTART = NO
        C WE DO NOT WANT ANY NEGATIVE CONCENTRATIONS.
          RETURN
        END

```

99

CARD NR. SEVERITY DETAILS DIAGNOSIS OF PHOULEM

153 I AN IF STATEMENT MAY BE MORE EFFICIENT THAN A 2 OR 3 BRANCH COMPUTED GO TO STATEMENT.



04/15/80 11.10.37

FIN 4.8.498

11-ACE

76/76 OPT=1

SUBROUTINE GFUN

```

CALL HNDRYT(XC(NCPTS),UVAL,UVAL(1:2),UMJ,U*(IMJUA+UZDT,NPUE)
ILIM = NCPTS * J * KORD - KORD - 1
HC(KPDE+4)=1.0
IF (ORDU(KPDE).EQ.0.0.AND.UHUA(KPDE).EQ.0.0)GO TO 60
UDOT(KS)=DZOT(KPDE)
HC(KPDE+3)=AILIM)*UHUA(KPDE)
HC(KPDE+4)=ORDU(KPDE)-HC(KPDE+3)
60 CONTINUE
RETURN
END

```

GFUN 59  
GFUN 60  
GFUN 61  
GFUN 62  
GFUN 63  
GFUN 64  
GFUN 65  
GFUN 66  
GFUN 67  
GFUN 68

60

65

04/15/80 11.10.37

FTN 4.4-49M

OPT=1 MOUNU=0/ (ACE

76/76

SUMROUTINE EVAL

```

1      SUMROUTINE EVAL(ICPT,NPDE,C,UVAL,A,ILEFT)
2      CALLING ARGUMENTS ARE DEFINED BELOW AND IN PDECOL.
3
4      C
5      C SURROUTINE EVAL EVALUATES U(K), UX(K), AND UXX(K), K=1 TO NPDE,
6      C AT THE COLLOCATION POINT WITH INDEX ICPT USING THE VALUES OF
7      C THE BASIS FUNCTION COEFFICIENTS IN C AND THE BASIS FUNCTION VALUES
8      C STORED IN A. THE RESULTS ARE STORED IN UVAL AS FOLLOWS..
9      C UVAL(K,1) = U(K), UVAL(K,2) = UX(K), AND UVAL(K,3) = UXX(K).
10     C
11     C PACKAGE ROUTINES CALLED.. NONE
12     C USER ROUTINES CALLED.. NONE
13     C CALLED BY.. GFUN,PDECOL,PSETFH
14     C FUNCTION FUNCTIONS USED.. NONE
15     C
16     C
17     DIMENSION C(1),UVAL(NPDE,3),A(1),ILEFT(1)
18     COMMON/SIZES/NINT,KORD,IDUM(2),LPTS,NEQN,IUVAL
19     IK = ILEFT(ICPT) - KORD
20     IC = 3*KORD*(ICPT-1)
21     DO 10 M=1,3
22         ICC = IC + KORD*(M-1)
23         DO 10 J=1,NPDE
24             JC=(J-1)*NCPTS
25             UVAL(J,M) = 0.
26             DO 10 I=1,KORD
27                 UVAL(J,M)=UVAL(J,M)+C(JC+I)*IK**A(I+ICC)
28     10 CONTINUE
29     RETURN
30     END

```

11.10.37

04/15/80

FTN 4.8.49H

PAGE

76/76

SUBROUTINE DIFFUN

OPT=1

MOUND=

/

FACE

```

1      SUBROUTINE DIFFUN (N, T, Y, YDOT, IER, PW, PIV, WORK, IWORK)
2      C-----
3      C CALLING ARGUMENTS ARE DEFINED BELOW AND IN PDECOL.
4      C-----
5      C THIS ROUTINE COMPUTES YDOT = A(Y,T) = G(Y,T) BY USE OF
6      C THE ROUTINES GFUN, ADUA, DECB, AND SOLR.
7      C-----
8      C PACKAGE ROUTINES CALLED.. ADUA, DECB, GFUN, SOLR
9      C USER ROUTINES CALLED.. NONE
10     C CALLED BY.. STIFM
11     C FURTHER FUNCTIONS USED.. NONE
12     C-----
13     DIMENSION Y(N), YDOT(N), PW(1), PIV(1), WORK(1), IWORK(1)
14     COMMON/GEAR9/ EPSJ, MU, ML, MU, MW, NM1, NOML, NOM, NUM
15     COMMON /SIZES/ NINT, KORD, NCC, NP1, NCPTS, NEUN, IQUAD
16     COMMON /ISTART/ IW1, IW2, IW3, IQUAD(S), IW9, IW10, IW11, IW12, IW13, IW14,
17     * IW15, IW16, IW17, IW18
18     DO 10 I = 1, NOM
19     PW(I) = 0.
20     DO 20 K = 1, NPDE
21     CALL GFUN(T, Y, YDOT, NPDE, NCPTS, MU, K(IW1), WORK, WORK(IW14),
22     * WORK(IW15), WORK(IW16), WORK(IW13), WORK(IW9), IWORK(K,K,K))
23     NST = (K-1)*NOM + 1
24     IST = (K-1)*NCPTS + 1
25     CALL ADUA(PW(NST), NCPTS, WORK(IST), IWORK, WORK, NPDE, K)
26     CALL DECB(NCPTS, NCPTS, ML, MU, PW(NST), PIV(IST), IER)
27     IF (IER.NE.0) RETURN
28     CALL SOLR(NCPTS, NCPTS, ML, MU, PW(NST), YDOT(IST), PIV(IST))
29     CONTINUE
30     RETURN
31     END
32

```

```

1      SUBROUTINE ADDA(PW,NO,A,ILEFT,B,INPDE,KPIE)
2      ADDA
3      ADDA
4      ADDA
5      ADDA
6      ADDA
7      ADDA
8      ADDA
9      ADDA
10     ADDA
11     ADDA
12     ADDA
13     ADDA
14     ADDA
15     ADDA
16     ADDA
17     ADDA
18     ADDA
19     ADDA
20     ADDA
21     ADDA
22     ADDA
23     ADDA
24     ADDA
25     ADDA
26     ADDA
27     ADDA
28     ADDA
29     ADDA
30     ADDA
31     ADDA
32     ADDA
33     ADDA
34     ADDA
35     ADDA
36     ADDA
37     ADDA
38     ADDA
39     ADDA
40     ADDA
41     ADDA

C-----
C CALLING ARGUMENTS ARE DEFINED BELOW AND IN PUECOL AND STIFL.
C-----
C SUBROUTINE ADDA ADDS THE MATRIX A TO THE MATRIX STORED IN PW IN
C HAND FORM. PW IS STORED BY DIAGONALS WITH THE LOWERMOST DIAGONAL
C STORED IN THE FIRST COLUMN OF THE ARRAY.
C ONLY THE PART OF PW CORRESPONDING TO THE KPIE-TH PUE IS
C WORKED WITH.
C-----
C PACKAGE ROUTINES CALLED.. NONE
C USER ROUTINES CALLED.. NONE
C CALLED BY.. DIFFUN,PSTTH
C FURTHER FUNCTIONS USED.. NONE
C-----
C DIMENSION PW(N0+1),A(1),ILEFT(1),HC(INPDE,4)
C COMMON /SIZES/ NINT,KORD,NCC,NPI,NCPTS,NEUN,IQUAD
C-----
C ADD THE BOUNDARY CONDITION PORTIONS OF THE A MATRIX TO PW ( THE FIRST
C AND LAST ROWS).
C-----
C ICOL=ILEFT(1)+IQUAD-1
C PW(1,ICOL)=PW(1,ICOL)+HC(KPDE,1)
C PW(1,ICOL)=PW(1,ICOL)+HC(KPIE,2)
C PW(NCPTS,ICOL)=PW(NCPTS,ICOL)+HC(KPDE,3)
C PW(NCPTS,ICOL)=PW(NCPTS,ICOL)+HC(KPDE,4)
C-----
C UPDATE THE REMAINING ROWS OF PW BY ADDING THE APPROPRIATE VALUES
C IN A TO PW.
C-----
C IND = NCPTS - 1
C DO 20 I=2,IND
C   I2 = (I-1) * KORD + 3
C   ICOL = ILEFT(I) - 1 + IQUAD - 1
C   DO 20 J=1,KORD
C     J1=ICOL+J
C     J2 = I2 + J
C     PW(I,J1)=PW(I,J1)+A(J2)
C   20  WETURN
C     END

```



04/15/80 11.10.37

FTN 4.N.49H

76/76 OPT=1 MOUND=0.0/ 1/ACE

```

1      SUBROUTINE PSETIH (C, PW, NO, CO, MITER, IEM, A, ILEFT, XC, UVAL,
5      SAVE2, IPIV, CMAX, OFDU, OFDUX, OFDUA, DZDT, DRDU, DRDUX, HC, NPDE)
C      CALLING ARGUMENTS ARE DEFINED BELOW AND IN PDECOL AND STIFIH.
C
C      PSETIH IS CALLED BY STIFIH TO COMPUTE AND PROCESS THE MATRIX
C      PW = A - H*EL(1)*(DG/DC), WHERE A AND DG/DC ARE TREATED IN HAND
C      FORM. DG/DC IS COMPUTED, EITHER WITH THE AID OF THE USER-SUPPLIED
C      ROUTINE DERIVC IF MITER = 1, OR BY FINITE DIFFERENCING WITH THE AID
C      OF THE PACKAGE-SUPPLIED ROUTINE DIFFP IF MITER = 2. FINALLY,
C      PW IS SUBJECTED TO LU DECOMPOSITION IN PREPARATION FOR LATER
C      SOLUTION OF LINEAR SYSTEMS WITH PW AS COEFFICIENT MATRIX.
C      SEE SUBROUTINES DECH AND SOLB.
C      CC FOR THE FLAME CODE, MITER ALWAYS EQUALS 2.
C
C      IN ADDITION TO VARIABLES DESCRIBED PREVIOUSLY, COMMUNICATION
C      WITH PSETIH USES THE FOLLOWING..
C      EPSJ = SORT(MOUND), USED IN THE NUMERICAL JACOBIAN INCREMENTS.
C      MW = ML + MU + 1.
C      NMI = NO - 1.
C      NOML = NO*ML.
C      NOMU = NO*MU.
C
C      PACKAGE ROUTINES CALLED.. ADDA, DECH, DIFFP, EVAL, GFUN
C      USER ROUTINES CALLED.. BNDRY, DERIVC
C      CALLED BY.. STIFIH
C      FORTRAN FUNCTIONS USED.. ABS, FLOAT, MAX0, MIN0, SORT
C
C      DIMENSION PW(1), C(1), CMAX(1)
C      DIMENSION A(1), ILEFT(1), BC(1), AC(1), UVAL(NPDE,3), SAVE2(1), IPIV(1)
C      DIMENSION OFDU(NPDE), OFDUX(NPDE), OFDUA(NPDE), DRDUX(NPDE)
C      DIMENSION DZDT(NPDE), DRDU(NPDE), DRDUA(NPDE)
C      COMMON /SIZES/ NINT, KORD, NCC, NPI, NCPTS, NEUN, IUQAU
C      COMMON /GEAR1/ T, H, DUMMY(3), UKORD, D, N, IDUMMY(3)
C      COMMON /GEAR9/ EPSJ, H0, ML, MU, MW, NM1, NOML, NOMU, NOW1
C      DO 10 I=1, NO*1
C      PW(I)=0.0
C      D=0.0
C      DO 100 K=1, NPDE
C      IST=(K-1)*NCPTS+1
C      IEND=K*NCPTS
C      CALL GFUN(T, C, SAVE2, NPDE, NCPTS, A, BC, DRDU, DRDUX, DZDT, XC, UVAL,
C      ILEFT, K, K, K)
C      DO 20 I=IST, IEND
C      U=D+SAVE2(I)*0.2
C      CONTINUE
C      H0=ABS(H)*SORT(D)/FLOAT(N0))*1.0+0.3*UROUND
C      DO 30 I=1, NCPTS
C      I2 = (I-1)*KORD+3
C      CALL EVAL(I, NPDE, C, UVAL, A, ILEFT)
C      CALL DIFFP(I, XC(1), I, UVAL, UVAL(1,2), UVAL(1,3),
C      OFDU, OFDUX, OFDUA, NPDE, CMAX, SAVE2)
C      ICOL = ILEFT(I) - 1 + IUQAU - 1
C      KLOW = MAX0(1, K-2-NCPTS)
C      KUP = MIN0(KORD, KORD+1-2)
C      DO 30 K=1, NPI*JF
C      NST=(K-1)*N0+1+1

```

11.10.37

04/15/80

FTN 4.H.49H

I-ACE

76/76

SUBROUTINE PSETIH

OPT=1

HOUND=0

```

      NEND=K*NOWI
      DO 30 K=K+1,KUP
        J1=ICOL*KBLK
        J2 = J1 + KBLK
        J3 = J2 + KBLK
        J4 = J3 + KBLK
        PW(NST-1,J1-1)*NCPTS=DFDUX(K)*A(J2)+DFDUX(K)*A(J3)
        *DFDUX(K)*A(J4)
      30 CONTINUE
      C-----
      C MODIFY THE LAST AND THE FIRST BLOCK PWS FOR THE BOUNDARY CONDITIONS.
      C CURRENT INFORMATION FOR THE RIGHT BOUNDARY CONDITION IS ALREADY IN
      C THE ARRAY DRDU, DRDUX AS A RESULT OF A PREVIOUS CALL TO GFUN.
      C-----
      DO 50 K=1,NPUE
        NST=(K-1)*NOWI+1
        NEND=K*NOWI
        IF (DRDU(K).EQ.0.0.AND.DRDUX(K).EQ.0.0)GO TO 50
        DO 40 J=1,MW
          PW(NST-1,J)*NCPTS=0.0
        40 CONTINUE
      50 CONTINUE
      CALL EVAL(1,NPDE,C,UVAL,A,ILEFT)
      CALL WNDY(T,XC(1),UVAL,UVAL(1,2),DRDU,DRDUX,DZUT,NPDE)
      DO 70 K=1,NPDE
        NST=(K-1)*NOWI+1
        NEND=K*NOWI
        IF (DRDU(K).EQ.0.0.AND.DRDUX(K).EQ.0.0)GO TO 70
        DO 60 J=1,MW
          PW(NST,(J-1)*NCPTS)=0.0
        60 CONTINUE
      70 CONTINUE
      DO 80 I=1,NOW
        PW(I)=PW(I)*CON
      80 CONTINUE
      C ADD MATRIX A(C,T) TO PW.
      C-----
      DO 90 K=1,NPUE
        NST=(K-1)*NOWI+1
        NEND=K*NOWI
        IST=(K-1)*NCPTS+1
        CALL ADDA(PW(NST),NCPTS,A,ILEFT,C*NPDE,K)
      90 CONTINUE
      C DO LU DECOMPOSITION ON PW.
      C-----
      CALL DECH(NCPTS,NCPTS,ML,MU,PW(NST),IPIV(IST),IER)
      CONTINUE
      RETURN
      END

```

11.10.37

04/15/80

FIN 4.8.49H

I-ACE

76/76

SUBROUTINE DIFFF

OPT=1

HOUNO=

```

1      SUBROUTINE DIFFF(I,A,IPT,U,UXX,UAA,DFDU,DFDUX,DFDUXX,NPDE,CMAX,
2      SAVE2)
3
4      C CALLING ARGUMENTS ARE DEFINED BELOW AND IN PUELCL.
5
6      C
7      C SUBROUTINE DIFFF IS USED IF MITER=2 TO PROVIDE FINITE DIFFERENCE
8      C APPROXIMATIONS FOR THE PARTIAL DERIVATIVES OF THE K-TH USER DEFINED
9      C FUNCTION IN THE F ROUTINE WITH RESPECT TO THE VARIABLES U, UAA, AND
10     C UXX. THESE PARTIALS WITH RESPECT TO U, UAA, AND UXX ARE COMPUTED,
11     C STORED, AND RETURNED IN THE NPDE VECTORS DFDU,DFDUX,AND DFDUXX
12     C RESPECTIVELY. AT COLLUCATION POINT NUMBER IPT. AGAIN, PARTIALS
13     C OF THE K-TH FUNCTION WITH RESPECT TO THE JTH VARIABLE, K NOT
14     C EQUAL TO J, ARE ASSUMED TO BE NEGLIGABLE.
15
16     C PACKAGE ROUTINES CALLED.. NONE
17     C USER ROUTINES CALLED.. F
18     C CALLED BY.. PSETIH
19     C FORTRAN FUNCTIONS USED.. AMAX1
20
21     DIMENSION U(NPDE),UA(NPDE),UXX(NPDE),DFDU(NPDE)
22     DIMENSION DFDUX(NPDE),DFDUXX(NPDE),CMAX(1),SAVE2(1)
23     COMMON/GEAR9/EPJS,RU,ML,MU,MW,MH,1,NOML,NUM,NUM1
24     COMMON/SIZES/NINT,KOND,NCC,NPD,NCP,NPTS,NEL,NOLUAD
25     KSKR=2
26     KSKT=2
27     DO 40 K=1,NPDE
28       IF (K.EQ.NPDE) KSKH=1
29       ID=(K-1)*NCP,NPTS,IPT
30       UK=U(K)
31       R=EPJS*CMAX(ID)
32       H = AMAX1(R,R0)
33       U(K)=U(K)+P
34       MINV = 1. / R
35       CALL F(T,X,U,UXX,DFDU (K),NPDE,K,IPT,1,KSKH)
36       DFDU(K)=(DFDU(K)-SAVE2(ID))*MINV
37       U(K)=UK
38       UK=U(K)
39       UXX(K)=UXX(K)+H
40       CALL F(T,X,U,UXX,DFDUX (K),NPDE,K,IPT,1,KSKH)
41       DFDUX(K)=(DFDUX(K)-SAVE2(ID))*MINV
42       UK=U(K)
43       UXX(K)=UK
44       UXX(K)=UXX(K)+H
45       CALL F(T,X,U,UXX,DFDUXX(K),NPDE,K,IPT,1,KSKH)
46       DFDUXX(K)=(DFDUXX(K)-SAVE2(ID))*MINV
47       UXX(K)=UK
48       UXX(K)=UXX(K)+H
49       UXX(K)=UXX(K)+H
50
51     40 CONTINUE
52     RETURN
53     END

```

11.10.37

04/15/80

FTN 4.M.49H

IMAGE

76/76

SUBROUTINE INTERP

76/76

OPT=1 MOUND=0.0/

```

1      SUBROUTINE INTERP (IOUT, Y, NU, Y0)
2      C CALLING ARGUMENTS ARE DEFINED BELOW (NU) IN STIF.H
3      C
4      C
5      C SUBROUTINE INTERP COMPUTES INTERPOLATED VALUES OF THE DEPENDENT
6      C VARIABLE Y AND STORES THEM IN Y0. THE INTERPOLATION IS TO THE
7      C POINT T = TOUT, AND USES THE NONDSTRUCT HISTORY ARRAY Y, AS FOLLOWS..
8      C
9      C
10     C      NU
11     C      Y0(I) = SUM Y(I,J+1)*S**J
12     C      J=0
13     C
14     C WHERE S = -(T-TOUT)/H.
15     C
16     C PACKAGE ROUTINES CALLED.. NONE
17     C USER ROUTINES CALLED.. NONE
18     C CALLED BY.. PDECOL
19     C FORTRAN FUNCTIONS USED.. NONE
20     C
21     C
22     C
23     C
24     C
25     C
26     C
27     C
28     C
29     C
30     C
31     C
32     C

```





```

115 EL(8) = 1.9345238095238E-03
    EL(9) = 1.1160714285714E-04
    EL(10) = 2.7557319223986E-06
    GO TO 900
110 EL(11) = 2.869754442857E-01
    EL(13) = 1.4144841269841E-00
    EL(14) = 1.0772156084656E-00
    EL(15) = 4.9856701940035E-01
    EL(16) = 0.1484375E-00
    EL(17) = 2.9060570987654E-02
    EL(18) = 3.7202380952381E-03
    EL(19) = 2.9968544650085E-04
    EL(110) = 1.3778659611993E-05
    EL(111) = 2.7557319223986E-07
    GO TO 900
111 EL(11) = 2.8018959644394E-01
    EL(13) = 1.4644841269841E-00
    EL(14) = 1.1715145502646E-00
    EL(15) = 5.7935819003527E-01
    EL(16) = 1.8832286155203E-01
    EL(17) = 4.1430362654321E-02
    EL(18) = 6.2111441798942E-03
    EL(19) = 6.2520667989418E-04
    EL(110) = 4.0417401528513E-05
    EL(111) = 1.5156525573192E-06
    EL(112) = 2.5052108365442E-08
    GO TO 900
112 EL(11) = 2.7426554003160E-01
    EL(13) = 1.5099386724387E-00
    EL(14) = 1.2602711640212E-00
    EL(15) = 6.5923418209877E-01
    EL(16) = 2.3045800264550E-01
    EL(17) = 5.5697246105232E-02
    EL(18) = 9.4394841269841E-03
    EL(19) = 1.1192749664312E-03
    EL(110) = 9.0939153439153E-05
    EL(111) = 4.8225304641975E-06
    EL(112) = 1.5031265031265E-07
    EL(113) = 2.087675698768E-09
    GO TO 900
113 EL(11) = 1.0E-00
    GO TO 900
202 EL(11) = 6.6666666666667E-01
    EL(13) = 3.3333333333333E-01
    GO TO 900
203 EL(12) = 5.4545454545455E-01
    EL(13) = EL(11)
    EL(14) = 9.0909090909091E-02
    GO TO 900
204 EL(11) = 0.48E-00
    EL(13) = 0.7E-00
    EL(14) = 0.2E-00
    EL(15) = 0.02E-00
    GO TO 900
205 EL(11) = 4.3795420437954E-01
    EL(13) = 8.2116784321168E-01
    EL(14) = 3.1021497410219E-01

```

SUBROUTINE COSET 76/76 OPT=1 KOUNU=---/ TRACE FIN 4.M-49K 04/15/80 11.10.37 PAGE 54

EL(5) = 5.4744525547445E-02  
EL(6) = 3.6496350364964E-03

C

900 DU 910 K = 1.3

910 TDIR) = PERTST(IND-METH(K)

TU(4) = .5E-00+TU(2)/ FLOAT(NU+)

RETURN

END

COSET 173  
COSET 174  
COSET 175  
COSET 176  
COSET 177  
COSET 178  
COSET 179  
COSET 180

CARD NR. SEVERITY DETAILS DIAGNOSIS OF PROBLEM

46 1 AN IF STATEMENT MAY BE MORE EFFICIENT THAN A 2 OR 3 BRANCH COMPUTED GO TO STATEMENT.

```

1      SUBROUTINE DECB (NDIM, N, ML, MU, B, IPIV, IER)
2      DECH
3      DECH
4      DECH
5      C SUBROUTINES DECB AND SOLB FORM A TWO SUBROUTINE PACKAGE FOR THE
6      C DIRECT SOLUTION OF A SYSTEM OF LINEAR EQUATIONS IN WHICH THE
7      C COEFFICIENT MATRIX IS REAL AND BANDWIDTH
8      C
9      C LU DECOMPOSITION OF BAND MATRIX A...  $L \cdot U = P \cdot A$ , WHERE P IS A
10     C PERMUTATION MATRIX, L IS A UNIT LOWER TRIANGULAR MATRIX,
11     C AND U IS AN UPPER TRIANGULAR MATRIX.
12     C N = ORDER OF MATRIX.
13     C B = N BY (2*ML+MU+1) ARRAY CONTAINING THE MATRIX A ON INPUT
14     C AND ITS FACTORED FORM ON OUTPUT.
15     C ON INPUT, B(1,K) (1.LE.I.LE.N) CONTAINS THE K-TH
16     C DIAGONAL OF A, OR A(I,J) IS STORED IN B(I,J+1+ML+1).
17     C ON OUTPUT, B CONTAINS THE L AND U FACTORS, WITH
18     C U IN COLUMNS 1 TO ML+MU+1, AND L IN COLUMNS
19     C ML+MU+2 TO 2*ML+MU+1.
20     C ML+MU = WIDTHS OF THE LOWER AND UPPER PARTS OF THE BAND, NOT
21     C COUNTING THE MAIN DIAGONAL. TOTAL BANDWIDTH IS ML+MU+1.
22     C NDIM = THE FIRST DIMENSION (COLUMN LENGTH) OF THE ARRAY B.
23     C IPIV = ARRAY OF LENGTH N CONTAINING PIVOT INFORMATION.
24     C IER = ERROR INDICATOR..
25     C = 0 IF NO ERROR.
26     C = K IF THE K-TH PIVOT CHOSEN WAS ZERO (A IS SINGULAR).
27     C THE INPUT ARGUMENTS ARE NDIM, N, ML, MU, B.
28     C THE OUTPUT ARGUMENTS ARE B, IPIV, IER.
29     C
30     C PACKAGE ROUTINES CALLED.. NONE
31     C USER ROUTINES CALLED.. NONE
32     C CALLED BY.. DIFFUN,INITAL,PSETIB
33     C FORTRAN FUNCTIONS USED.. ABS,MINO
34     C
35     C-----
36     C DIMENSION R (NDIM,1),IPIV(N)
37     C IER = 0
38     C IF (N.EQ.1) GO TO 92
39     C LL = ML + MU + 1
40     C N1 = N - 1
41     C IF (ML.EQ.0) GO TO 32
42     C DO 30 I = 1,ML
43     C   II = MU + 1
44     C   K = ML + 1 - I
45     C   DO 10 J = 1,II
46     C     H(I,J) = B(I,J+K)
47     C   K = II + 1
48     C   DO 20 J = K,LL
49     C     H(I,J) = 0.
50     C   CONTINUE
51     C   LH = ML
52     C   DO 90 NR = 1,N1
53     C     NP = NR + 1
54     C     IF (LP.NE.N) LW = LR + 1
55     C     MX = LR
56     C     XM = ABS(R(NP,1))
57     C     IF (ML.FU.0) GO TO 42
58     C     DO 40 I = NP,LW
59     C       IF (ABS(H(I,1)) .LE. XM) GO TO 40
60     C

```

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FIN 4.M.49M

SUBROUTINE DECH 76/76 OPT=1 MOUND=\*\*/ IPACE

```

40      MX = 1
41      XM = ABS(B(I,1))
42      CONTINUE
      IPIV(NR) = MX
      IF (MX .EQ. NH) GO TO 69
      DO 50 I = 1, LL
        XX = R(NR,I)
        R(NR,I) = R(MX,I)
        R(MX,I) = XX
        XM = H(NR,I)
        IF (XM .EQ. 0.) GO TO 100
        H(NH,I) = 1./XM
        IF (ML .EQ. 0) GO TO 90
        XM = -R(NH,I)
        KK = MIND(N-NH,LL-1)
        DO 40 I = NP, LH
          J = LL + I - NR
          XX = R(I,I)*XM
          R(NH,J) = XX
          DO 70 II = 1, KK
            H(I,II) = H(I,II+1) + XX*(NR,II+1)
            H(LL,II) = 0.
          70      CONTINUE
        80      NR = N
        90      IF (H(N,1) .EQ. 0.) GO TO 100
            H(N,1) = 1./H(N,1)
            RETURN
        100      IER = NR
            RETURN
            END

```

DECH 59  
 DECH 60  
 DECH 61  
 DECH 62  
 DECH 63  
 DECH 64  
 DECH 65  
 DECH 66  
 DECH 67  
 DECH 68  
 DECH 69  
 DECH 70  
 DECH 71  
 DECH 72  
 DECH 73  
 DECH 74  
 DECH 75  
 DECH 76  
 DECH 77  
 DECH 78  
 DECH 79  
 DECH 80  
 DECH 81  
 DECH 82  
 DECH 83  
 DECH 84  
 DECH 85  
 DECH 86  
 DECH 87  
 DECH 88

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FIN 4.8.49H

76/76 OPT=1 ROUNDED=\*\*/ 1-ACE

SUBROUTINE SOLB

```

1      SUBROUTINE SOLB (NDIM, N, ML, MU, R, Y, IPIV)
2      SOLB
3      SOLB
4      SOLB
5      SOLB
6      SOLB
7      SOLB
8      SOLB
9      SOLB
10     SOLB
11     SOLB
12     SOLB
13     SOLB
14     SOLB
15     SOLB
16     SOLB
17     SOLB
18     SOLB
19     SOLB
20     SOLB
21     SOLB
22     SOLB
23     SOLB
24     SOLB
25     SOLB
26     SOLB
27     SOLB
28     SOLB
29     SOLB
30     SOLB
31     SOLB
32     SOLB
33     SOLB
34     SOLB
35     SOLB
36     SOLB
37     SOLB
38     SOLB
39     SOLB
40     SOLB
41     SOLB
42     SOLB
43     SOLB
44     SOLB
45     SOLB
46     SOLB
47     SOLB
48     SOLB

C-----
C SUBROUTINES DECH AND SOLB FORM A TWO SUBROUTINE PACKAGE FOR THE
C DIRECT SOLUTION OF A SYSTEM OF LINEAR EQUATIONS IN WHICH THE
C COEFFICIENT MATRIX IS REAL AND Banded.
C
C SOLUTION OF A*X = C GIVEN LU DECOMPOSITION OF A FROM DECH.
C Y = RIGHT-HAND VECTOR C. OF LENGTH N. ON INPUT.
C = SOLUTION VECTOR X ON OUTPUT.
C ALL THE ARGUMENTS ARE INPUT ARGUMENTS.
C THE OUTPUT ARGUMENT IS Y.
C
C PACKAGE ROUTINES CALLED.. NONE
C USER ROUTINES CALLED.. NONE
C CALLED BY.. DIFFUN,INITAL,STIFIB
C FORTAN FUNCTIONS USED.. MINO
C-----
C DIMENSION R(NDIM,1),Y(N),IPIV(N)
C IF (N.EQ. 1) GO TO 60
C N1 = N - 1
C LL = ML + MU + 1
C IF (ML.EQ. 0) GO TO 32
C DO 30 NR = 1,N1
C   IF (IPIV(NR).EQ. NR) GO TO 10
C   J = IPIV(NR)
C   XX = Y(NR)
C   Y(NR) = Y(J)
C   Y(J) = XX
C 10 KK = MIN0(N-NR,ML)
C 20 Y(NR+1) = Y(NR+1) + Y(NR)*B(NR,LL+1)
C 30 CONTINUE
C 32 LL = LL - 1
C 34 Y(N) = Y(N)*B(N,1)
C KK = 0
C DO 50 NR = 1,N1
C   NR = N - NR
C   IF (KK.NE. LL) KK = KK + 1
C   UP = 0.
C   IF (LL.EQ. 0) GO TO 50
C   DO 40 I = 1,KK
C     DP = DP + B(NR,I+1)*Y(NR+1)
C 40 Y(NR) = (Y(NR) - UP)*B(NR,1)
C 50 RETURN
C 60 Y(1) = Y(1)*B(1,1)
C RETURN
C END

```

[illegible]

124

[illegible]H<sub>2</sub>-AIR FLAME. 50 PERCENT H<sub>2</sub>. WARNATZ KINETICS.

PRESSURE = 1.0000E+10 ATM.

NPUE =

	M	U	UC	UM	HU	H2D
1	1.000E+00	6.4851E-14	5.2691E-05	5.2103E+04	4.5634E-07	
2	1.700E+01	1.1025E-10	5.2566E-05	5.5424E+02	1.3530E-07	
3	1.600E+01	1.0374E-10	1.0468E-07	3.7221E+03	1.3746E-07	
4	3.000E+01	2.1401E-10	1.9533E-11	1.5146E+02	9.0944E-08	
5	3.400E+01	2.2043E-10	1.3187E-10	-9.5683E-01	9.0471E-08	
6	2.000E+00	6.6851E-02	3.7564E-02	1.5951E+01	2.3005E-07	
7	3.200E+01	2.1790E-01	4.3495E-04	-3.5684E+02	9.4255E-08	
8	1.600E+01	1.1673E-08	2.4508E-01	-3.2110E+03	1.1438E-07	
9	2.400E+01	7.1725E-01	7.1725E-01	-3.9514E+02	1.3902E-07	

IC = 2.9400E-01      TH = 1.936E+00

$$\begin{aligned} \text{IPN} &= 1.0000\text{E}+03 & \text{MM} &= 5.0000\text{E}-05 & \text{TMN} &= 1.0000\text{E}-03 \end{aligned}$$
$$\begin{aligned} \text{CPM} &= 4.7854 \times 10^4 \\ W &= 7.8440 \times 10^{-4} \\ \text{TPC} &= 2.9890 \times 10^2 \end{aligned}$$

$\mu_{\text{H}} = 0$        $\mu_{\text{H}} = 5$        $\mu_{\text{H}} = 10$        $\mu_{\text{H}} = 15$        $\mu_{\text{H}} = 20$

SALES = 1.000001 SATS = 1.000001

```

INSTANT = 1      NINST = 12      NMIKI = 0      PLUSP = 0.

```

$$T_{H1} = 1 \quad T_{H2} = 5.0000 - 0.1$$

IMAA = 1.60001 + 112

1000 + 40000 = 41000

INITIAL DENSITY = 0.3065E-04

U. 9.801ME-02 1.6733E-01 2.1434E-01 2.5094E-01 2.7550E-01 3.0000E-01 3.2450E-01 3.4901E-01 3.8466E-01  
 4.3267E-01 5.0198E-01 6.0000E-01 7.4505E-02 8.5767E-02 9.801ME-02 1.066ME-01 1.2522E-02 1.3267E-01 1.4134E-01 1.5866E-01 1.7455E-01 1.8571E-01  
 1.11535E-01 1.2801E-01 1.4796E-01 2.0408E-01 2.1021E-01 2.1634E-01 2.2500E-01 2.2743E-01 2.3800E-01 2.4466E-01 2.5099E-01 2.5712E-01 2.6324E-01 2.6937E-01 2.7243E-01  
 2.7550E-01 2.7856E-01 2.8162E-01 2.8468E-01 2.8775E-01 2.9081E-01 2.9387E-01 2.9694E-01 3.0000E-01 3.0306E-01 3.0613E-01 3.0919E-01 3.1225E-01 3.1532E-01 3.1838E-01  
 3.2144E-01 3.2450E-01 3.2757E-01 3.3063E-01 3.3369E-01 3.3676E-01 3.3982E-01 3.4288E-01 3.4595E-01 3.4901E-01 3.5208E-01 3.5514E-01 3.5821E-01 3.6127E-01 3.6434E-01  
 3.6740E-01 3.7047E-01 3.7353E-01 3.7660E-01 3.7967E-01 3.8273E-01 3.8580E-01 3.8887E-01 3.9193E-01 3.9500E-01 3.9807E-01 4.0114E-01 4.0421E-01 4.0728E-01 4.1034E-01  
 4.1341E-01 4.1648E-01 4.1955E-01 4.2262E-01 4.2569E-01 4.2876E-01 4.3183E-01 4.3490E-01 4.3797E-01 4.4104E-01 4.4411E-01 4.4718E-01 4.5025E-01 4.5332E-01 4.5639E-01  
 4.5946E-01 4.6253E-01 4.6560E-01 4.6867E-01 4.7174E-01 4.7481E-01 4.7788E-01 4.8095E-01 4.8402E-01 4.8709E-01 4.9016E-01 4.9323E-01 4.9630E-01 4.9937E-01 5.0244E-01  
 5.0551E-01 5.0858E-01 5.1165E-01 5.1472E-01 5.1779E-01 5.2086E-01 5.2393E-01 5.2700E-01 5.3007E-01 5.3314E-01 5.3621E-01 5.3928E-01 5.4235E-01 5.4542E-01 5.4849E-01  
 5.5156E-01 5.5463E-01 5.5770E-01 5.6077E-01 5.6384E-01 5.6691E-01 5.7000E-01 5.7307E-01 5.7614E-01 5.7921E-01 5.8228E-01 5.8535E-01 5.8842E-01 5.9149E-01 5.9456E-01  
 5.9763E-01 6.0070E-01 6.0377E-01 6.0684E-01 6.0991E-01 6.1298E-01 6.1605E-01 6.1912E-01 6.2219E-01 6.2526E-01 6.2833E-01 6.3140E-01 6.3447E-01 6.3754E-01 6.4061E-01  
 6.4368E-01 6.4675E-01 6.4982E-01 6.5289E-01 6.5596E-01 6.5903E-01 6.6210E-01 6.6517E-01 6.6824E-01 6.7131E-01 6.7438E-01 6.7745E-01 6.8052E-01 6.8359E-01 6.8666E-01  
 6.8973E-01 6.9280E-01 6.9587E-01 6.9894E-01 7.0201E-01 7.0508E-01 7.0815E-01 7.1122E-01 7.1429E-01 7.1736E-01 7.2043E-01 7.2350E-01 7.2657E-01 7.2964E-01 7.3271E-01  
 7.3578E-01 7.3885E-01 7.4192E-01 7.4499E-01 7.4806E-01 7.5113E-01 7.5420E-01 7.5727E-01 7.6034E-01 7.6341E-01 7.6648E-01 7.6955E-01 7.7262E-01 7.7569E-01 7.7876E-01  
 7.8183E-01 7.8490E-01 7.8797E-01 7.9104E-01 7.9411E-01 7.9718E-01 8.0025E-01 8.0332E-01 8.0639E-01 8.0946E-01 8.1253E-01 8.1560E-01 8.1867E-01 8.2174E-01 8.2481E-01  
 8.2788E-01 8.3095E-01 8.3402E-01 8.3709E-01 8.4016E-01 8.4323E-01 8.4630E-01 8.4937E-01 8.5244E-01 8.5551E-01 8.5858E-01 8.6165E-01 8.6472E-01 8.6779E-01 8.7086E-01  
 8.7393E-01 8.7700E-01 8.8007E-01 8.8314E-01 8.8621E-01 8.8928E-01 8.9235E-01 8.9542E-01 8.9849E-01 9.0156E-01 9.0463E-01 9.0770E-01 9.1077E-01 9.1384E-01 9.1691E-01  
 9.1998E-01 9.2305E-01 9.2612E-01 9.2919E-01 9.3226E-01 9.3533E-01 9.3840E-01 9.4147E-01 9.4454E-01 9.4761E-01 9.5068E-01 9.5375E-01 9.5682E-01 9.5989E-01 9.6296E-01  
 9.6603E-01 9.6910E-01 9.7217E-01 9.7524E-01 9.7831E-01 9.8138E-01 9.8445E-01 9.8752E-01 9.9059E-01 9.9366E-01 9.9673E-01 9.9980E-01 10.0287E-01 10.0594E-01 10.0901E-01  
 10.1208E-01 10.1515E-01 10.1822E-01 10.2129E-01 10.2436E-01 10.2743E-01 10.3050E-01 10.3357E-01 10.3664E-01 10.3971E-01 10.4278E-01 10.4585E-01 10.4892E-01 10.5199E-01 10.5506E-01  
 10.5813E-01 10.6120E-01 10.6427E-01 10.6734E-01 10.7041E-01 10.7348E-01 10.7655E-01 10.7962E-01 10.8269E-01 10.8576E-01 10.8883E-01 10.9190E-01 10.9497E-01 10.9804E-01 11.0111E-01  
 11.0418E-01 11.0725E-01 11.1032E-01 11.1339E-01 11.1646E-01 11.1953E-01 11.2260E-01 11.2567E-01 11.2874E-01 11.3181E-01 11.3488E-01 11.3795E-01 11.4102E-01 11.4409E-01 11.4716E-01  
 11.5023E-01 11.5330E-01 11.5637E-01 11.5944E-01 11.6251E-01 11.6558E-01 11.6865E-01 11.7172E-01 11.7479E-01 11.7786E-01 11.8093E-01 11.8400E-01 11.8707E-01 11.9014E-01 11.9321E-01  
 11.9628E-01 11.9935E-01 12.0242E-01 12.0549E-01 12.0856E-01 12.1163E-01 12.1470E-01 12.1777E-01 12.2084E-01 12.2391E-01 12.2698E-01 12.3005E-01 12.3312E-01 12.3619E-01 12.3926E-01  
 12.4233E-01 12.4540E-01 12.4847E-01 12.5154E-01 12.5461E-01 12.5768E-01 12.6075E-01 12.6382E-01 12.6689E-01 12.6996E-01 12.7303E-01 12.7610E-01 12.7917E-01 12.8224E-01 12.8531E-01  
 12.8838E-01 12.9145E-01 12.9452E-01 12.9759E-01 13.0066E-01 13.0373E-01 13.0680E-01 13.0987E-01 13.1294E-01 13.1601E-01 13.1908E-01 13.2215E-01 13.2522E-01 13.2829E-01 13.3136E-01  
 13.3443E-01 13.3750E-01 13.4057E-01 13.4364E-01 13.4671E-01 13.4978E-01 13.5285E-01 13.5592E-01 13.5899E-01 13.6206E-01 13.6513E-01 13.6820E-01 13.7127E-01 13.7434E-01 13.7741E-01  
 13.8048E-01 13.8355E-01 13.8662E-01 13.8969E-01 13.9276E-01 13.9583E-01 13.9890E-01 14.0197E-01 14.0504E-01 14.0811E-01 14.1118E-01 14.1425E-01 14.1732E-01 14.2039E-01 14.2346E-01  
 14.2653E-01 14.2960E-01 14.3267E-01 14.3574E-01 14.3881E-01 14.4188E-01 14.4495E-01 14.4802E-01 14.5109E-01 14.5416E-01 14.5723E-01 14.6030E-01 14.6337E-01 14.6644E-01 14.6951E-01  
 14.7258E-01 14.7565E-01 14.7872E-01 14.8179E-01 14.8486E-01 14.8793E-01 14.9100E-01 14.9407E-01 14.9714E-01 15.0021E-01 15.0328E-01 15.0635E-01 15.0942E-01 15.1249E-01 15.1556E-01  
 15.1863E-01 15.2170E-01 15.2477E-01 15.2784E-01 15.3091E-01 15.3398E-01 15.3705E-01 15.4012E-01 15.4319E-01 15.4626E-01 15.4933E-01 15.5240E-01 15.5547E-01 15.5854E-01 15.6161E-01  
 15.6468E-01 15.6775E-01 15.7082E-01 15.7389E-01 15.7696E-01 15.8003E-01 15.8310E-01 15.8617E-01 15.8924E-01 15.9231E-01 15.9538E-01 15.9845E-01 16.0152E-01 16.0459E-01 16.0766E-01  
 16.1073E-01 16.1380E-01 16.1687E-01 16.1994E-01 16.2301E-01 16.2608E-01 16.2915E-01 16.3222E-01 16.3529E-01 16.3836E-01 16.4143E-01 16.4450E-01 16.4757E-01 16.5064E-01 16.5371E-01  
 16.5678E-01 16.5985E-01 16.6292E-01 16.6599E-01 16.6906E-01 16.7213E-01 16.7520E-01 16.7827E-01 16.8134E-01 16.8441E-01 16.8748E-01 16.9055E-01 16.9362E-01 16.9669E-01 16.9976E-01  
 17.0283E-01 17.0590E-01 17.0897E-01 17.1204E-01 17.1511E-01 17.1818E-01 17.2125E-01 17.2432E-01 17.2739E-01 17.3046E-01 17.3353E-01 17.3660E-01 17.3967E-01 17.4274E-01 17.4581E-01  
 17.4888E-01 17.5195E-01 17.5502E-01 17.5809E-01 17.6116E-01 17.6423E-01 17.6730E-01 17.7037E-01 17.7344E-01 17.7651E-01 17.7958E-01 17.8265E-01 17.8572E-01 17.8879E-01 17.9186E-01  
 17.9493E-01 17.9800E-01 18.0107E-01 18.0414E-01 18.0721E-01 18.1028E-01 18.1335E-01 18.1642E-01 18.1949E-01 18.2256E-01 18.2563E-01 18.2870E-01 18.3177E-01 18.3484E-01 18.3791E-01  
 18.4098E-01 18.4405E-01 18.4712E-01 18.5019E-01 18.5326E-01 18.5633E-01 18.5940E-01 18.6247E-01 18.6554E-01 18.6861E-01 18.7168E-01 18.7475E-01 18.7782E-01 18.8089E-01 18.8396E-01  
 18.8703E-01 18.9010E-01 18.9317E-01 18.9624E-01 18.9931E-01 19.0238E-01 19.0545E-01 19.0852E-01 19.1159E-01 19.1466E-01 19.1773E-01 19.2080E-01 19.2387E-01 19.2694E-01 19.3001E-01  
 19.3308E-01 19.3615E-01 19.3922E-01 19.4229E-01 19.4536E-01 19.4843E-01 19.5150E-01 19.5457E-01 19.5764E-01 19.6071E-01 19.6378E-01 19.6685E-01 19.6992E-01 19.7299E-01 19.7606E-01  
 19.7913E-01 19.8220E-01 19.8527E-01 19.8834E-01 19.9141E-01 19.9448E-01 19.9755E-01 20.0062E-01 20.0369E-01 20.0676E-01 20.0983E-01 20.1290E-01 20.1597E-01 20.1904E-01 20.2211E-01  
 20.2518E-01 20.2825E-01 20.3132E-01 20.3439E-01 20.3746E-01 20.4053E-01 20.4360E-01 20.4667E-01 20.4974E-01 20.5281E-01 20.5588E-01 20.5895E-01 20.6202E-01 20.6509E-01 20.6816E-01  
 20.7123E-01 20.7430E-01 20.7737E-01 20.8044E-01 20.8351E-01 20.8658E-01 20.8965E-01 20.9272E-01 20.9579E-01 20.9886E-01 21.0193E-01 21.0500E-01 21.0807E-01 21.1114E-01 21.1421E-01  
 21.1728E-01 21.2035E-01 21.2342E-01 21.2649E-01 21.2956E-01 21.3263E-01 21.3570E-01 21.3877E-01 21.4184E-01 21.4491E-01 21.4798E-01 21.5105E-01 21.5412E-01 21.5719E-01 21.6026E-01  
 21.6333E-01 21.6640E-01 21.6947E-01 21.7254E-01 21.7561E-01 21.7868E-01 21.8175E-01 21.8482E-01 21.8789E-01 21.9096E-01 21.9403E-01 21.9710E-01 22.0017E-01 22.0324E-01 22.0631E-01  
 22.0938E-01 22.1245E-01 22.1552E-01 22.1859E-01 22.2166E-01 22.2473E-01 22.2780E-01 22.3087E-01 22.3394E-01 22.3701E-01 22.4008E-01 22.4315E-01 22.4622E-01 22.4929E-01 22.5236E-01  
 22.5543E-01 22.5850E-01 22.6157E-01 22.6464E-01 22.6771E-01 22.7078E-01 22.7385E-01 22.7692E-01 22.8000E-01 22.8307E-01 22.8614E-01 22.8921E-01 22.9228E-01 22.9535E-01 22.9842E-01  
 23.0149E-01 23.0456E-01 23.0763E-01 23.1070E-01 23.1377E-01 23.1684E-01 23.1991E-01 23.2298E-01 23.2605E-01 23.2912E-01 23.3219E-01 23.3526E-01 23.3833E-01 23.4140E-01 23.4447E-01  
 23.4754E-01 23.5061E-01 23.5368E-01 23.5675E-01 23.5982E-01 23.6289E-01 23.6596E-01 23.6903E-01 23.7210E-01 23.7517E-01 23.7824E-01 23.8131E-01 23.8438E-01 23.8745E-01 23.9052E-01  
 23.9359E-01 23.9666E-01 23.9973E-01 24.0280E-01 24.0587E-01 24.0894E-01 24.1201E-01 24.1508E-01 24.1815E-01 24.2122E-01 24.2429E-01 24.2736E-01 24.3043E-01 24.3350E-01 24.3657E-01  
 24.3964E-01 24.4271E-01 24.4578E-01 24.4885E-01 24.5192E-01 24.5499E-01 24.5806E-01 24.6113E-01 24.6420E-01 24.6727E-01 24.7034E-01 24.7341E-01 24.7648E-01 24.7955E-01 24.8262E-01  
 24.8569E-01 24.8876E-01 24.9183E-01 24.9490E-01 24.9797E-01 25.0104E-01 25.0411E-01 25.0718E-01 25.1025E-01 25.1332E-01 25.1639E-01 25.1946E-01 25.2253E-01 25.2560E-01 25.2867E-01  
 25.3174E-01 25.3481E-01 25.3788E-01 25.4095E-01 25.4402E-01 25.4709E-01 25.5016E-01 25.5323E-01 25.5630E-01 25.5937E-01 25.6244E-01 25.6551E-01 25.6858E-01 25.7165E-01 25.7472E-01  
 25.7779E-01 25.8086E-01 25.8393E-01 25.8700E-01 25.9007E-01 25.9314E-01 25.9621E-01 25.9928E-01 26.0235E-01 26.0542E-01 26.0849E-01 26.1156E-01 26.1463E-01 26.1770E-01 26.2077E-01  
 26.2384E-01 26.2691E-01 26.2998E-01 26.3305E-01 26.3612E-01 26.3919E-01 26.4226E-01 26.4533E-01 26.4840E-01 26.5147E-01 26.5454E-01 26.5761E-01 26.6068E-01 26.6375E-01 26.6682E-01  
 26.6989E-01 26.7296E-01 26.7603E-01 26.7910E-01 26.8217E-01 26.8524E-01 26.8831E-01 26.9138E-01 26.9445E-01 26.9752E-01 27.0059E-01 27.0366E-01 27.0673E-01 27.0980E-01 27.1287E-01  
 27.1594E-01 27.1901E-01 27.2208E-01 27.2515E-01 27.2822E-01 27.3129E-01 27.3436E-01 27.3743E-01 27.4050E-01 27.4357E-01 27.4664E-01 27.4971E-01 27.5278E-01 27.5585E-01 27.5892E-01  
 27.6199E-01 27.6506E-01 27.6813E-01 27.7120E-01 27.7427E-01 27.7734E-01 27.8041E-01 27.8348E-01 27.8655E-01 27.8962E-01 27.9269E-01 27.9576E-01 27.9883E-01 28.0190E-01 28.0497E-01  
 28.0804E-01 28.1111E-01 28.1418E-01 28.1725E-01 28.2032E-01 28.2339E-01 28.2646E-01 28.2953E-01 28.3260E-01 28.3567E-01 28.3874E-01 28.4181E-01 28.4488E-01 28.4795E-01 28.5102E-01  
 28.5409E-01 28.5716E-01 28.6023E-01 28.6330E-01 28.6637E-01 28.6944E-01 28.7251E-01 28.7558E-01 28.7865E-01 28.8172E-01 28.8479E-01 28.8786E-01 28.9093E-01 28.9400E-01 28.9707E-01  
 29.0014E-01 29.0321E-01 29.0628E-01 29.0935E-01 29.1242E-01 29.1549E-01 29.1856E-01 29.2163E-01 29.2470E-01 29.2777E-01 29.3084E-01 29.3391E-01 29.3698E-01 29.4005E-01 29.4312E-01  
 29.4619E-01 29.4926E-01 29.5233E-01 29.5540E-01 29.5847E-01 29.6154E-01 29.6461E-01 29.6768E-01 29.7075E-01 29.7382E-01 29.7689E-01 29.7996E-01 29.8303E-01 29.8610E-01 29.8917E-01  
 29.9224E-01 29.9531E-01 29.9838E-01 30.0145E-01 30.0452E-01 30.0759E-01 30.1066E-01 30.1373E-01 30.1680E-01 30.1987E-01 30.2294E-01 30.2601E-01 30.2908E-01 30.3215E-01 30.3522E-01  
 30.3829E-01 30.4136E-01 30.4443E-01 30.4750E-01 30.5057E-01 30.5364E-01 30.5671E-01 30.5978E-01 30.6285E-01 30.6592E-01 30.6899E-01 30.7206E-01 30.7513E-01 30.7820E-01 30.8127E-01  
 30.8434E-01 30.8741E-01 30.9048E-01 30.9355E-01 30.9662E-01 30.9969E-01 31.0276E-01 31.0583E-01 31.0890E-01 31.1197E-01 31.1504E-01 31.1811E-01 31.2118E-01 31.2425E-01 31.2732E-01  
 31.3039E-01 31.3346E-01 31.3653E-01 31.3960E-01 31.4267E-01 31.4574E-01 31.4881E-01 31.5188E-01 31.5495E-01 31.5802E-01 31.6109E-01 31.6416E-01 31.6723E-01 31.7030E-01 31.7337E-01  
 31.7644E-01 31.7951E-01 31.8258E-01 31.8565E-01 31.8872E-01 31.9179E-01 31.9486E-01 31.9793E-01 32.0100E-01 32.0407E-01 32.0714E-01 32.1021E-01 32.1328E-01 32.1635E-01 32.1942E-01  
 32.2249E-01 32.2556E-01 32.2863E-01 32.3170E-01 32.3477E-01 32.3784E-01 32.4091E-01 32.4398E-01 32.4705E-01 32.5012E-01 32.5319E-01 32.5626E-01 32.5933E-01 32.6240E-01 32.6547E-01  
 32.6854E-01 32.7161E-01 32.7468E-01 32.7775E-01 32.8082E-01 32.8389E-01 32.8696E-01 32.9003E-01 32.9310E-01 32.9617E-01 32.9924E-01 33.0231E-01 33.0538E-01 33.

MUN TIME = 5.8600E+00

M2D AT LEFT = 9.0467E-03 2.7060E-03 2.7492E-03 1.8197E-03 1.8094E-03 4.6010E-03 1.8849E-03 2.3677E-03 0.  
UPM AT LEFT = 1.3411E-04 -1.7184E-04 5.4405E-10 5.1594E-08 6.2184E-09 -4.3164E-06 -3.1783E-05 1.4065E-05 1.6976E-05 7.6777E-05

UPM2 AT LEFT = -1.8652E-07 -5.3094E-07 -9.6195E-08 -3.6054E-05 -4.9577E-07 1.0648E-03 7.2452E-03 -6.4641E-03 -1.8076E-03 -3.8844E-02

FLAME SPEED = 1.7634E+02 1.8839E+02 4.5225E+02 2.8554E+02 7.3178E+02 2.0676E+02 9.6679E+01 1.1624E+02 6.6176E+02

FLAME SPEED = 5.9300E+02 4.6934E+02 7.5362E+02 6.6555E+03 5.3483E+04 7.3938E+01 1.8262E+01 4.3247E+01 -8.1220E+02

FLAME SPEED = 2.6960E+03 1.9868E+03 6.5519E+05 -4.7572E+09 -1.3298E+08 2.0165E+01 1.6072E+01 1.4593E+01 -3.0986E+02

FLAME FRONT FROM PH1 = 2.4225E-01 TO PH1 = 1.6627E-01

FLAME FRONT FROM X = 2.0247E-02 TO X = 5.2817E-02

FLAME THICKNESS = 3.2565E-02 CM

MU = 3.8372E-02 FLSP = 6.0845E+01 PHNEM = 2.9967E-01

SPN1 = -7.6744E-01 SPN2 = -1.5648E+00 SPEED = -7.6744E-01 US = -6.6600E-01

T = 1.0000E-02 DT = 6.3976E-04 TOTAL STEPS = 62

NF = 84 NJ = 9

MUN TIME = 6.9270E+00

M2D AT LEFT = 9.0467E-03 2.7060E-03 2.7492E-03 1.8197E-03 1.8094E-03 4.6010E-03 1.8849E-03 2.3677E-03 0.

UPM AT LEFT = 4.0121E-04 8.8467E-04 5.6945E-09 1.3841E-07 3.1086E-08 -2.6803E-05 -3.7487E-05 6.0286E-05 3.8157E-06 4.3990E-04

UPM2 AT LEFT = -5.5150E-07 -1.8045E-06 -4.8160E-07 -6.2099E-05 -4.5260E-06 4.9272E-03 4.4479E-03 -1.2060E-02 -1.2443E-03 -1.4961E-01

FLAME SPEED = 2.5450E+02 3.5339E+02 4.0492E+02 3.0447E+02 5.0180E+02 1.7720E+02 9.3857E+01 1.0844E+02 3.5161E+02

FLAME SPEED = 5.4617E+02 4.8131E+02 6.7111E+02 5.4544E+03 4.5531E+04 4.6939E+01 5.1043E+01 5.5297E+01 -4.1936E+02

FLAME SPEED = 5.9544E+03 4.2421E+03 1.7790E+06 5.9311E+08 -2.6225E+08 4.5591E+01 3.5784E+01 3.4471E+01 2.4146E+02

FLAME FRONT FROM PH1 = 2.3797E-01 TO PH1 = 1.7932E-01

FLAME FRONT FROM X = 2.0047E-02 TO X = 5.8427E-02

FLAME THICKNESS = 3.8379E-02 CM

W0 = 2.4700E-02 FLSP = 4.5509E+01 PHMEW = 2.9897E-01

SPN1 = -5.7400E-01 SPN2 = -6.2066E-01 SPEED = -5.7400E-01 OS = 1.9344E-01

T = 1.4548E-02 DT = 7.6771E-04 TOTAL STIPS = 17

NF = 111 NJ = 11

RUN TIME = 9.0630E+00

W2D AT LEFT = 9.0867E-03 2.7060E-03 2.7442E-03 1.4177E-03 1.8099E-03 4.6010E-03 1.4849E-03 2.3677E-03 0.

UPM AT LEFT = 1.8169E-04 6.1181E-08 4.5057E-08 -5.4542E-07 1.7072E-07 -7.1372E-05 -1.0440E-04 1.7900E-04 -2.9770E-06 2.1453E-03

UPM AT LEFT = -2.5068E-06 -1.0467E-05 -7.6319E-06 -6.1009E-05 -2.6656E-05 1.1245E-02 1.4853E-02 -3.2268E-02 2.2788E-03 -4.0088E-01

FLAME SPEED = 2.3717E+02 3.1362E+02 3.6576E+02 3.2174E+02 4.7275E+02 1.5612E+02 1.0011E+02 1.0781E+02 2.2795E+02

FLAME SPEED = 4.4120E+02 4.5066E+02 5.6781E+02 5.0474E+03 2.4456E+05 1.2997E+02 9.1541E+01 9.4450E+01 -2.4342E+02

FLAME SPEED = 1.4396E+04 8.4076E+03 1.9968E+06 3.2017E+08 -6.0778E+08 1.2016E+02 9.4871E+01 9.0839E+01 1.5316E+02

FLAME FRONT FROM PH10 = 2.2925E-01 TO PH10 = 1.9590E-01

FLAME FRONT FROM X = 1.9610E-02 TO X = 6.5311E-02

FLAME THICKNESS = 4.5901E-02 CM

W0 = 3.7339E-02 FLSP = 5.9206E+01 PHMEW = 2.9432E-01

SPN1 = -7.4677E-01 SPN2 = -1.1806E+00 SPEED = -7.4677E-01 OS = -1.7277E-01

T = 3.1390E-02 DT = 8.0203E-04 TOTAL STIPS = 95

NF = 143 NJ = 13

RUN TIME = 1.1129E+01

W2D AT LEFT = 9.0867E-03 2.7060E-03 2.7442E-03 1.4177E-03 1.8099E-03 4.6010E-03 1.4849E-03 2.3677E-03 0.

UPM AT LEFT = 1.3614E-07 4.2629E-07 5.6574E-07 -1.2574E-05 2.4439E-06 -3.0510E-04 -2.2904E-04 4.0097E-04 1.4274E-04 5.4524E-03

UPM2 AT LEFT = -1.0791E-05 -7.3639E-05 -9.8405E-05 8.7448E-04 -3.7392E-04 1.5002E-02 4.0314E-02 -6.7902E-02 1.1674E-02 -7.9979E-01

FLAME SPEED = 2.2109E+02 3.3001E+02 3.0495E+02 1.9344E+02 5.0891E+02 1.6442E+02 1.3393E+02 1.3715E+02 1.9403E+02

FLAME SPEED = 3.2308E+02 3.7075E+02 5.0901E+02 9.2670E+02 5.7429E+05 1.8743E+02 1.5547E+02 1.5900E+02 -4.3509E+01

FLAME SPEED = 1.3330E+04 6.2534E+03 8.2793E+05 1.5214E+07 1.1597E+09 1.9324E+02 1.6055E+02 1.5731E+02 3.5407E+01

FLAME FRONT FROM PMI\* = 2.1631E-01 TO PMI\* = 4.1426E-01

FLAME FRONT FROM X = 1.8492E-02 TO X = 7.5319E-02

FLAME THICKNESS = 5.6824E-02 CM

MO = 7.7035E-02 FLSP = 1.2215E+02 PMME = 2.8702E-01

SPM1 = -1.5407E+00 SPM2 = -5.5740E+00 SPELIM = -3.5599E+00 DS = -2.8131E+00

T = 3.9392E-02 DT = 8.0527E-04 TOTAL STIPS = 109

NF = 169 NJ = 15

RUN TIME = 1.3591E+01

M2D AT LEFT = 9.0467E-03 2.7060E-03 2.7492E-03 1.8147E-03 1.8094E-03 4.6010E-03 1.8849E-03 2.3677E-03 0.

UPM AT LEFT = 2.5742E-07 3.0457E-06 3.9378E-06 -6.4701E-05 1.0302E-05 -3.8862E-04 -6.5792E-04 9.3984E-04 1.5412E-04 1.0333E-02

UPM2 AT LEFT = -3.4347E-05 -3.8602E-04 -5.0808E-04 6.1011E-03 -1.2096E-03 1.5902E-02 7.9617E-02 -1.1364E-01 1.4161E-02 -1.0606E+00

FLAME SPEED = 2.2151E+02 2.9369E+02 2.6650E+02 1.0414E+02 4.2863E+02 1.7817E+02 1.5631E+02 1.5835E+02 1.6448E+02

FLAME SPEED = 2.4461E+02 3.3919E+02 4.1628E+02 -1.9619E+03 4.8375E+05 2.0644E+02 1.8109E+02 1.8457E+02 -6.4147E+01

FLAME SPEED = 7.1908E+03 3.1005E+03 2.5715E+05 -6.6601E+07 1.2131E+09 2.1504E+02 1.8276E+02 1.4023E+02 9.6744E+00

FLAME FRONT FROM PMI\* = 2.1631E-01 TO PMI\* = 4.3256E-01

FLAME FRONT FROM X = 1.8463E-02 TO X = 4.1804E-02

FLAME THICKNESS = 6.3344E-02 CM

MO = 1.1640E-01 FLSP = 1.8454E+02 PMME = 2.8562E-01

SPM1 = -2.3241E+00 SPM2 = -5.6343E+00 SPELIM = -3.4412E+00 DS = -1.2111E-01

T = 4.6021E-02 DT = 6.3384E-04 TOTAL STIPS = 123

NF = 140 NJ = 17

RUN TIME = 1.5322E+01

M2D AT LEFT = 9.0867E-03 2.7060E-03 2.7492E-03 1.8117E-03 1.8094E-03 4.6010E-03 1.8849E-03 2.3677E-03 0.  
UPH AT LEFT = 1.6379E-07 5.9733E-06 7.2149E-06 -1.4871E-04 1.0728E-05 -2.9831E-04 -1.0717E-03 1.2402E-03 2.5444E-04 1.0318E-02  
UPH2 AT LEFT = -2.1751E-05 -6.9877E-04 -8.4307E-04 1.4327E-02 -1.0355E-03 1.3712E-02 1.0949E-01 -1.3197E-01 -3.0135E-03 -1.0441E+00  
FLAME SPEED = 2.2124E+02 2.6829E+02 2.4199E+02 1.4614E+02 3.1293E+02 1.8574E+02 1.6963E+02 1.6967E+02 1.7904E+02  
FLAME SPEED = 2.7012E+02 3.1793E+02 3.3499E+02 -1.0871E+03 2.8065E+05 2.1727E+02 1.9836E+02 2.0087E+02 -1.9702E+02  
FLAME SPEED = 3.7140E+03 1.5632E+03 6.7018E+04 -6.9575E+06 2.4318E+07 2.3393E+02 2.0007E+02 1.9788E+02 3.7685E+00

FLAME FRONT FROM PH1 = 2.2497E-01 TO PH1 = 4.5854E-01

FLAME FRONT FROM X = 1.9091E-02 TO X = 8.8814E-02

FLAME THICKNESS = 6.9724E-02 CM

PO = 1.3188E-01 FLSP = 2.0912E+02 PHNEW = 2.9280E-01

SPN1 = -2.6376E+00 SPN2 = -2.8335E+00 SPEED = -2.7356E+00 US = 1.1454E+00

T = 5.6025E-02 DT = 1.0479E-03 TOTAL STIPS = 139

NF = 209 NJ = 20

RUN TIME = 1.7068E+01

M2D AT LEFT = 9.0867E-03 2.7060E-03 2.7492E-03 1.8117E-03 1.8094E-03 4.6010E-03 1.8849E-03 2.3677E-03 0.  
UPH AT LEFT = 9.1874E-08 2.9204E-06 4.5704E-06 -1.3637E-04 1.4728E-05 -2.2250E-04 -1.9884E-04 7.5415E-04 3.7922E-04 6.0855E-03  
UPH2 AT LEFT = -1.2384E-05 -3.6947E-04 -5.7917E-04 1.2777E-02 -1.5437E-03 9.1537E-03 1.9139E-02 -4.2614E-02 -1.6102E-02 -6.5492E-01  
FLAME SPEED = 2.3534E+02 2.3211E+02 2.3325E+02 1.9400E+02 2.7003E+02 1.9749E+02 1.9006E+02 1.8749E+02 1.8114E+02  
FLAME SPEED = 2.7364E+02 2.9605E+02 3.0843E+02 -4.3114E+02 1.4435E+05 2.3722E+02 2.2068E+02 2.2164E+02 6.7612E+02  
FLAME SPEED = 2.7364E+02 2.9605E+02 3.0843E+02 -4.3114E+02 1.4435E+05 2.3722E+02 2.2068E+02 2.2164E+02 6.7612E+02

FLAME FRONT FROM PH1\* = 2.2918E-01 TO PH1\* = 4.9329E-01

FLAME FRONT FROM X = 1.9279E-02 TO X = 9.9404E-02

FLAME THICKNESS = 8.0583E-02 CM

MO = 1.4305E-01 FLSP = 2.2683E+02 PHNEW = 2.9724E-01

SPH1 = -2.8610E+00 SPH2 = -3.3408E+00 SPELUM = -2.8610E+00 DS = -1.2543E-01

I = 7.1404E-02 UT = 1.4736E-03 TOTAL STOPS = 15M

NF = 233 NJ = 25

RUN TIME = 1.9431E+01

W20 AT LEFT = 9.0867E-03 2.7494E-03 1.8191E-03 1.8094E-03 4.6010E-03 1.8849E-03 2.3677E-03 0.

UPH AT LEFT = 1.2103E-07 2.8517E-06 3.6109E-06 -1.0434E-04 2.2447E-05 -1.9507E-04 -3.6975E-04 3.7517E-04 2.6495E-04 4.2774E-03

UPH2 AT LEFT = -1.6295E-05 -3.7184E-04 -4.8423E-04 1.0496E-02 -2.4414E-03 6.8793E-03 3.9632E-02 -4.5179E-02 -8.5144E-03 -4.6556E-01

FLAME SPELU = 2.5372E+02 2.6206E+02 2.6767E+02 1.7310E+02 2.7371E+02 2.1963E+02 2.2302E+02 2.1924E+02 2.0520E+02

FLAME SPELU = 2.8644E+02 3.0762E+02 3.1457E+02 -1.7504E+03 1.0182E+05 2.5447E+02 2.4909E+02 2.4860E+02 2.4808E+02

FLAME SPELU = 9.3608E+02 5.4421E+02 3.1791E+03 -7.0245E+05 8.1794E+05 2.9265E+02 2.5248E+02 2.5241E+02 -5.0941E+00

FLAME FRONT FROM PH1\* = 2.2925E-01 TO PH1\* = 5.0176E-01

FLAME FRONT FROM X = 1.9137E-02 TO X = 1.0250E-01

FLAME THICKNESS = 8.3358E-02 CM

MO = 1.5449E-01 FLSP = 2.4575E+02 PHNEW = 2.9264E-01

SPH1 = -3.0997E+00 SPH2 = -4.1240E+00 SPELUM = -3.0119E+00 DS = -7.5047E-01

I = 9.0143E-02 UT = 1.4736E-03 TOTAL STOPS = 170

NF = 253 NJ = 25

RUN TIME = 2.0414E+01

M2D AT LEFT = 9.0867E-03 2.7060E-03 2.7492E-03 1.8197E-03 1.8094E-03 4.6010E-03 1.8849E-03 2.3677E-03 0.  
 UPH AT LEFT = 1.3944E-07 5.7160E-06 5.9374E-06 -2.2273E-04 4.2387E-05 -1.3166E-04 -1.5909E-04 3.9561E-04 2.6359E-04 3.9809E-03  
 UPH2 AT LEFT = -1.8579E-05 -6.8406E-04 -7.1016E-04 2.1099E-02 -4.2580E-03 6.6517E-03 1.6556E-02 -4.3869E-02 -1.5466E-02 -4.5566E-01  
 FLAME SPEED = 2.7093E+02 2.6658E+02 2.6395E+02 1.8991E+02 2.8215E+02 2.4401E+02 2.5285E+02 2.5002E+02 2.6111E+02  
 FLAME SPEED = 2.8627E+02 2.8993E+02 3.0330E+02 -2.2994E+03 7.8673E+04 2.8895E+02 2.6902E+02 2.6790E+02 1.8602E+02  
 FLAME SPEED = 5.6637E+02 4.3163E+02 1.2113E+03 -3.1134E+06 3.5688E+05 3.0557E+02 2.7084E+02 2.7380E+02 -6.0684E-01

FLAME FRONT FROM PH1\* = 2.2924E-01 TO PH1\* = 5.0191E-01

FLAME FRONT FROM X = 1.8953E-02 TO X = 1.0211E-01

FLAME THICKNESS = 8.3159E-02 CM

PO = 1.6643E-01 FLSP = 2.6390E+02 PHNEW = 2.9091E-01

SPN1 = -3.3286E+00 SPN2 = -3.9057E+00 SPEED1\* = -3.6172E+00 NS = -5.2937E-03

T = 1.1127E-01 DT = 2.6007E-03 TOTAL STEPS = 189

MF = 240 NJ = 24

MIN TIME = 2.3101E+01

M2D AT LEFT = 9.0867E-03 2.7060E-03 2.7492E-03 1.8197E-03 1.8094E-03 4.6010E-03 1.8849E-03 2.3677E-03 0.  
 UPH AT LEFT = 9.7204E-08 4.2032E-06 5.3071E-06 -2.0382E-04 6.9178E-05 -7.5909E-05 -1.5980E-04 3.0913E-04 2.5142E-04 2.7671E-03  
 UPH2 AT LEFT = -1.2927E-05 -4.9446E-04 -6.2235E-04 1.8559E-02 -4.6652E-03 5.4452E-03 1.5077E-02 -3.3260E-02 -1.8026E-02 -3.3763E-01  
 FLAME SPEED = 2.9237E+02 2.2559E+02 2.6960E+02 2.5666E+02 2.9052E+02 2.6244E+02 2.6945E+02 2.7194E+02  
 FLAME SPEED = 2.8687E+02 2.5857E+02 2.8790E+02 -2.5612E+02 5.9917E+04 2.7553E+02 2.7765E+02 2.7586E+02 2.1941E+02  
 FLAME SPEED = 3.9629E+02 3.7677E+02 5.4246E+02 -3.9917E+05 2.5655E+05 2.9994E+02 2.7439E+02 2.4174E+02 -3.4135E-01

FLAME FRONT FROM PH1\* = 2.3360E-01 TO PH1\* = 4.0202E-01

FLAME FRONT FROM X = 1.9196E-02 TO X = 6.4550E-02

FLAME THICKNESS = 4.5364E-02 CM

MO = 1.7328E-01 FLSP = 2.7476E+02 PMHEW = 2.9406E-01

SPN1 = -3.4656E+00 SPN2 = -3.8352E+00 SPEED = -3.4656E+00 DS = 1.5157E-01

I = 1.3407E-01 DT = 2.4007E-03 TOTAL STIPS = 199

NF = 297 NJ = 24

RUN TIME = 2.4287E+01

W20 AT LEFT = 9.0867E-03 2.7060E-03 2.7492E-03 1.8197E-03 1.8094E-03 4.6010E-03 1.8849E-03 2.3677E-03 0.

UPH AT LEFT = 8.7589E-04 3.1231E-06 4.1576E-06 -1.0191E-04 8.8511E-05 -5.2228E-05 -2.8264E-04 2.1760E-04 1.7158E-04 2.0362E-03

UPH2 AT LEFT = -1.1677E-05 -3.7422E-04 -4.9985E-04 1.1433E-02 -6.5888E-03 4.6128E-03 2.7916E-02 -2.4007E-02 -1.3040E-02 -2.6151E-01

FLAME SPEED = 3.0079E+02 2.1841E+02 2.7280E+02 2.6717E+02 2.9434E+02 2.7352E+02 2.7621E+02 2.7790E+02 2.8322E+02

FLAME SPEED = 2.8263E+02 2.5003E+02 2.8267E+02 5.0204E+01 5.0204E+04 2.7833E+02 2.7999E+02 2.7976E+02 2.6758E+02

FLAME SPEED = 3.2633E+02 3.1574E+02 3.0523E+02 -1.0908E+05 1.8487E+05 2.8950E+02 2.8019E+02 2.8173E+02 -5.8120E-01

FLAME FRONT FROM PH1\* = 2.3354E-01 TO PH1\* = 3.8365E-01

FLAME FRONT FROM X = 1.9115E-02 TO X = 5.7870E-02

FLAME THICKNESS = 3.8754E-02 CM

MO = 1.7602E-01 FLSP = 2.7912E+02 PMHEW = 2.9453E-01

SPN1 = -3.5205E+00 SPN2 = -3.8851E+00 SPEED = -3.5205E+00 DS = -5.6910E-02

I = 1.6937E-01 DT = 2.4007E-03 TOTAL STIPS = 213

NF = 321 NJ = 24

RUN TIME = 2.5940E+01

W20 AT LEFT = 9.0867E-03 2.7060E-03 2.7492E-03 1.8197E-03 1.8094E-03 4.6010E-03 1.8849E-03 2.3677E-03 0.

UPH AT LEFT = 1.0117E-07 3.7514E-06 4.4957E-06 -1.5201E-04 1.4064E-05 -6.3938E-05 -2.7749E-04 2.2419E-04 1.6731E-04 2.0679E-03

UPH2 AT LEFT = -1.34678E-05 -6.6793E-04 -5.3823E-04 1.4415E-02 -7.1557E-03 4.6164E-03 2.7800E-02 -2.8783E-02 -1.3810E-02 -2.65520E-01

FLAME SPEED = 2.9724E+02 2.2471E+02 2.7451E+02 2.5972E+02 2.9639E+02 2.7920E+02 2.7795E+02 2.8046E+02 2.8295E+02  
 FLAME SPEED = 2.8156E+02 2.4841E+02 2.8111E+02 -2.8220E+02 5.1520E+04 2.7981E+02 2.8036E+02 2.8046E+02 2.8334E+02  
 FLAME SPEED = 2.9209E+02 2.6262E+02 2.1870E+02 -1.9914E+05 1.4745E+05 2.8100E+02 2.8040E+02 2.8023E+02 2.7016E+00  
 FLAME FRONT FROM PH1\* = 2.3358E-01 TO PH1\* = 3.9590E-01

FLAME FRONT FROM X = 1.9145E-02 TO X = 6.2759E-02

FLAME THICKNESS = 4.3614E-02 CM

MO = 1.7690E-01 FLSP = 2.8050E+02 PHNE# = 2.9295E-01

SPN1 = -3.5340E+00 SPN2 = -3.7295E+00 SPECIM = -3.6337E+00 US = -1.1321E-01

T = 2.5039E-01 DT = 7.3361E-03 TOTAL STEPS = 231

NF = 346 NJ = 31

RUN TIME = 2.8076E+01

W20 AT LEFT = 9.0847E-03 2.7060E-03 2.7492E-03 1.8197E-03 1.8094E-03 4.6010E-03 1.8849E-03 2.3677E-03 0.

UPH AT LEFT = 7.7641E-04 2.9957E-06 4.2660E-06 -1.1900E-04 7.6449E-05 -3.7055E-05 -2.9985E-04 2.1739E-04 1.5478E-04 1.9192E-03

UPH2 AT LEFT = -1.0324E-05 -3.5239E-04 -5.0182E-04 1.0720E-02 -7.2342E-03 4.2275E-03 2.9310E-02 -2.3551E-02 -1.2709E-02 -2.4592E-01

FLAME SPEED = 3.0737E+02 2.1350E+02 2.7354E+02 2.8935E+02 2.9288E+02 2.8121E+02 2.7799E+02 2.8121E+02 2.8070E+02

FLAME SPEED = 2.8136E+02 2.4861E+02 2.8218E+02 -7.5774E+01 3.7126E+04 2.8059E+02 2.8037E+02 2.8060E+02 2.8733E+02

FLAME SPEED = 2.8339E+02 2.8157E+02 1.9220E+02 -1.1633E+05 1.0940E+05 2.8005E+02 2.8039E+02 2.8059E+02 -2.1515E+01

FLAME FRONT FROM PH1\* = 2.3796E-01 TO PH1\* = 4.1429E-01

FLAME FRONT FROM X = 1.9535E-02 TO X = 6.8654E-02

FLAME THICKNESS = 4.9118E-02 CM

MO = 1.7700E-01 FLSP = 2.8047E+02 PHNE# = 2.9595E-01

SPN1 = -3.5401E+00 SPN2 = -3.5112E+00 SPECIM = -3.5401E+00 US = 9.3642E-02

T = 3.7539E-01 DT = 5.4542E-03 TOTAL STEPS = 254

NF = 391 NJ = 36

RUN TIME = 3.1410E+01

W20 AT LEFT = 9.0867E-03 2.7060E-03 1.8147E-03 1.8044E-03 4.6010E-03 1.8849E-03 2.3677E-03 0.  
UPM AT LEFT = 4.3087E-06 1.5657E-06 2.6659E-06 -3.5304E-05 4.3792E-05 -2.4572E-05 -2.0169E-04 1.3041E-04 4.7095E-05 1.3275E-03  
UPM2 AT LEFT = -5.7378E-06 -1.8641E-04 -3.1793E-04 3.1901E-03 -4.1940E-03 3.2514E-03 1.9885E-02 -1.4329E-02 -7.2934E-03 -1.7179E-01

FLAME FRONT FROM PH1\* = 2.4219E-01 TO PH1\* = 4.1427E-01

FLAME FRONT FROM K = 1.9814E-02 TO K = 6.6422E-02

FLAME THICKNESS = 4.7108E-02 CM

MO = 1.7695E-01 FLSP = 2.8058E+02 PHWEH = 3.0193E-01

SPM1 = -3.5390E+00 SPM2 = -3.5069E+00 SPEED1 = -3.5190E+00 DS = 1.0530E-03

I = 5.0000E-01 DT = 1.6201E-02 TOTAL STEPS = 272

NF = 418 NJ = 39

RUN TIME = 3.4065E+01

H  
0.4451E-14 1.5344E-10 -2.1723E-10 -7.7488E-10 -1.1824E-09 -1.1027E-09 -1.9879E-10 1.4605E-09 5.4303E-09 3.9467E-09  
-3.6848E-09 -1.2145E-08 -1.7213E-08 -1.3900E-08 2.5834E-08 3.7028E-08 9.4224E-08 4.3741E-08 -6.3010E-08 -1.9058E-07  
-2.2332E-07 -6.5632E-06 3.7408E-07 1.2034E-06 2.5050E-06 3.2936E-06 4.1972E-06 6.3043E-06 1.0943E-05 1.9081E-05  
3.1926E-05 5.0622E-05 7.6328E-05 9.9830E-05 1.2874E-04 1.6341E-04 2.0421E-04 2.5149E-04 3.0563E-04 3.6697E-04  
4.3589E-04 5.1291E-04 5.9794E-04 6.9152E-04 7.9367E-04 9.0458E-04 1.0243E-03 1.1532E-03 1.2912E-03 1.4389E-03  
1.5955E-03 1.7593E-03 1.9283E-03 2.1010E-03 2.2754E-03 2.4444E-03 2.6242E-03 2.7944E-03 2.9504E-03 3.1014E-03  
3.2477E-03 3.3720E-03 3.4902E-03 3.5943E-03 3.6837E-03 3.7454E-03 3.7810E-03 3.7944E-03 3.7845E-03 3.7501E-03  
3.6708E-03 3.4708E-03 3.4024E-03 3.3394E-03 3.2742E-03 3.2144E-03 3.1575E-03 3.1036E-03 3.0524E-03 2.9943E-03  
2.9208E-03 2.8433E-03 2.7714E-03 2.7002E-03 2.6242E-03 2.5444E-03 2.4605E-03 2.3724E-03 2.2802E-03 2.1843E-03

UM  
1.1025E-10 6.9408E-09 -3.7700E-09 -2.1492E-08 -3.7130E-08 -3.9250E-08 -1.7944E-08 3.6407E-08 1.3548E-07 1.4454E-07  
4.4745E-08 -4.4045E-08 -1.0045E-07 -2.0045E-07 -3.0045E-07 -4.0045E-07 -5.0045E-07 -6.0045E-07 -7.0045E-07 -8.0045E-07

1.0227E-06 3.1152E-06 7.7174E-06 1.5638E-05 2.7684E-05 4.5460E-05 7.9510E-05 1.2741E-04 1.8996E-04 2.6512E-04  
3.5249E-04 4.5124E-04 5.6057E-04 6.3045E-04 6.8012E-04 7.1215E-04 7.2910E-04 7.353E-04 7.2803E-04 7.1515E-04  
6.9747E-04 6.7894E-04 6.6271E-04 6.4980E-04 6.4137E-04 6.3850E-04 6.3505E-04 6.311E-04 6.2751E-04 6.2403E-04  
1.4924E-03 1.2767E-03 1.5576E-03 1.5694E-03 1.5925E-03 1.5411E-03 1.5119E-03 1.4747E-03 1.4345E-03 1.3943E-03  
1.3573E-03 1.3264E-03 1.3044E-03 1.2850E-03 1.2733E-03 1.2608E-03 1.2449E-03 1.2266E-03 1.2174E-03 1.2079E-03  
1.2833E-03 1.2998E-03 1.2919E-03 1.2998E-03 1.3132E-03 1.3145E-03 1.3230E-03 1.3264E-03 1.3280E-03 1.3329E-03  
1.3393E-03 1.3472E-03 1.3557E-03 1.3638E-03 1.3707E-03 1.3755E-03 1.3773E-03 1.3773E-03 1.3773E-03 1.3773E-03

0

1.0376E-10 1.1730E-08 6.7114E-09 3.7470E-08 6.4373E-08 8.8552E-08 1.5863E-07 5.7733E-08 2.2353E-07 2.7054E-07  
1.8801E-07 6.3188E-08 5.3349E-08 2.3575E-07 7.0760E-07 1.5863E-07 2.9791E-06 3.7466E-06 4.1689E-06 4.9831E-06  
6.9344E-06 1.0766E-05 1.7221E-05 2.7042E-05 4.0972E-05 5.5172E-05 7.4494E-05 1.0280E-04 1.3144E-04 1.6850E-04  
2.1160E-04 2.6080E-04 3.1640E-04 3.5763E-04 3.9792E-04 4.3675E-04 4.7358E-04 5.0741E-04 5.3919E-04 5.6691E-04  
5.9055E-04 6.1055E-04 6.2968E-04 6.5092E-04 6.7252E-04 7.1164E-04 7.5708E-04 8.1053E-04 8.9298E-04 9.9085E-04  
1.1046E-03 1.2404E-03 1.3803E-03 1.5224E-03 1.6607E-03 1.7894E-03 1.9020E-03 1.9403E-03 2.0398E-03 2.0584E-03  
1.0467E-03 1.2404E-03 1.3803E-03 1.5224E-03 1.6607E-03 1.7894E-03 1.9020E-03 1.9403E-03 2.0398E-03 2.0584E-03  
7.2617E-04 6.1290E-04 5.2643E-04 4.3642E-04 3.6849E-04 3.1830E-04 2.8111E-04 2.5077E-04 2.4514E-04 2.3608E-04  
2.2943E-04 2.2103E-04 2.1400E-04 2.0811E-04 2.0314E-04 1.9808E-04 1.9506E-04 1.9150E-04 1.8797E-04 1.8317E-04  
1.7896E-04 1.7535E-04 1.7237E-04 1.7002E-04 1.6832E-04 1.6724E-04 1.6694E-04 1.6694E-04 1.6694E-04 1.6694E-04

M02

2.1401E-10 2.1290E-07 9.3423E-08 2.1863E-07 5.8302E-07 8.5950E-07 1.2056E-05 5.0839E-07 2.3971E-07 5.5285E-07  
2.4280E-08 1.3715E-06 3.3684E-06 5.8951E-06 8.8310E-06 1.2056E-05 1.5450E-05 2.1539E-05 3.7805E-05 6.8544E-05  
1.1406E-04 1.9065E-04 2.9061E-04 4.2223E-04 6.9801E-04 7.2696E-04 8.6569E-04 9.955E-04 1.0942E-03 1.1671E-03  
1.1479E-03 1.1483E-03 1.0357E-03 9.2899E-04 8.2631E-04 7.2900E-04 6.3838E-04 5.575E-04 4.825E-04 4.1974E-04  
3.6907E-04 3.2768E-04 2.9170E-04 2.6050E-04 2.3368E-04 2.1044E-04 1.9031E-04 1.7270E-04 1.5701E-04 1.4274E-04  
1.2490E-04 1.1808E-04 1.0730E-04 9.7398E-05 8.8198E-05 7.9588E-05 7.1415E-05 6.3671E-05 5.6433E-05 4.9724E-05  
4.3525E-05 3.7832E-05 3.2646E-05 2.7466E-05 2.3790E-05 1.8724E-05 1.4578E-05 1.1980E-05 8.589E-06 6.5445E-06  
4.9933E-06 3.8321E-06 2.9572E-06 2.0118E-06 1.3172E-06 8.3182E-07 5.4143E-08 5.4700E-08 5.3900E-08 5.1073E-08  
1.7200E-07 1.2472E-07 9.2013E-08 7.1245E-08 5.9773E-08 4.9545E-08 4.2869E-08 4.2737E-08 4.2737E-08 4.2737E-08  
4.8640E-08 4.6711E-08 4.5190E-08 4.4065E-08 4.3303E-08 4.2869E-08 4.2737E-08 4.2737E-08 4.2737E-08 4.2737E-08

M202

2.2049E-10 2.5113E-07 7.6061E-08 3.2423E-07 7.6506E-07 1.0377E-06 2.5241E-06 4.4929E-07 1.0972E-06 1.5931E-06  
9.0745E-07 3.0915E-07 1.4063E-06 1.7334E-06 6.4012E-06 1.0377E-06 2.5241E-06 4.4929E-07 1.0972E-06 1.5931E-06  
1.3136E-05 2.4272E-05 4.6625E-05 8.3412E-05 1.3945E-04 2.0300E-04 2.9624E-04 4.0743E-04 5.2883E-04 6.4969E-04  
7.6087E-04 9.5202E-04 9.1599E-04 9.3437E-04 9.4547E-04 9.3644E-04 9.1360E-04 8.7448E-04 8.2982E-04 7.7194E-04  
5.1609E-05 3.0860E-05 1.7269E-05 9.2345E-06 5.1575E-06 3.4378E-06 2.4752E-06 1.5343E-06 1.323E-06 1.2098E-06  
4.0327E-07 3.7132E-07 4.0755E-07 4.4883E-07 4.3204E-07 3.4900E-07 2.8672E-07 2.4179E-07 2.1080E-07 1.9031E-07  
1.7692E-07 1.6721E-07 1.5776E-07 1.4503E-07 1.3601E-07 1.3004E-07 1.2648E-07 1.2470E-07 1.2405E-07 1.2389E-07  
1.2358E-07 1.2280E-07 1.2216E-07 1.2112E-07 1.2112E-07 1.2065E-07 1.2014E-07 1.1962E-07 1.1899E-07 1.1806E-07  
1.1724E-07 1.1654E-07 1.1595E-07 1.1549E-07 1.1515E-07 1.1495E-07 1.1488E-07 1.1488E-07 1.1488E-07 1.1488E-07

M2

6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02  
6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02  
6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02  
6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02  
6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02 6.4451E-02  
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M2



[illegible]

FLAME FRONT FROM PH10 = 2.6219E-01 TO PH10 = 4.1627E-01

FLAME FRONT FROM  $\lambda = 1.9414E-02$  TO  $\lambda = 6.6924E-02$

FLAME THICKNESS =  $4.7110E-02$  CM

	FLAME THICKNESS = 4.7110E-02 CM
9	97
M	1.00000000E+00 5.21031710E+04 9.08674670E-03
UM	1.70000000E+01 5.5020874E+02 2.70604230E-03
U	2.74924000E-03 3.72206442E+03 2.74924000E-03
M2	1.60000000E+01 1.51456505E+02 1.81967200E-03
M2	3.30000000E+01 1.51456505E+02 1.81967200E-03
M2	3.40000000E+01 -9.56829781E+02 1.80941920E-03
M2	2.00000000E+00 -1.59507953E-01 4.60049146E-03
U2	3.20000000E+01 -3.56890803E-02 1.88489399E-03
U2	1.80000000E+01 -3.21070709E-03 2.36766026E-03
M20	1.80000000E+01 -3.05183200E-02 0.

[illegible]

1.00000000E+00  
MO = 1.7698E-01  
FLSP = 2.803E+02  
PMNEW = 3.014JE-01  
-J.539570E+00  
1.019380E-01

[illegible]

1.142723E-03	8.575691E-04	4.618628E-04	3.022893E-04	1.846218E-04	1.337297E-04	8.547893E-05	5.862451E-05
3.114448E-05	1.577175E-05	4.597169E-06	1.507748E-06	7.283767E-07	1.007329E-07	5.479287E-08	4.935581E-08
4.313488E-08	4.273245E-08	-8.181571E-07	1.206643E-06	7.730431E-08	1.042506E-05	5.637685E-05	2.651095E-04
2.204929E-10	1.638016E-07	8.1294410E-04	5.763647E-04	2.162042E-04	5.972726E-05	4.444189E-06	1.084060E-06
7.918157E-04	9.455837E-04	1.736816E-07	1.384435E-07	1.239794E-07	1.2233490E-07	1.200039E-07	1.174837E-07
4.227233E-07	3.039779E-07	1.148776E-07	6.484747E-02	7.478865E-02	6.468762E-02	6.437222E-02	6.390176E-02
1.150745E-07	1.148776E-07	6.484747E-02	5.761871E-02	5.415119E-02	5.097553E-02	4.545208E-02	4.227400E-02
6.485089E-02	6.485074E-02	5.959232E-02	3.416574E-02	3.391701E-02	3.413461E-02	3.4444837E-02	3.468553E-02
6.281797E-02	6.161351E-02	3.416574E-02	2.178921E-01	2.178869E-01	2.178125E-01	2.1773569E-01	2.158923E-01
3.447931E-02	3.500996E-02	2.179023E-01	1.938710E-01	1.628783E-01	1.428174E-01	1.087389E-01	7.756356E-02
2.178982E-01	2.178982E-01	1.938710E-01	2.237501E-03	3.529965E-04	1.604470E-04	8.996373E-05	8.411963E-05
2.111397E-01	2.049389E-01	6.639131E-03	6.679335E-06	1.718319E-05	9.269714E-05	4.667241E-04	1.816660E-03
4.280195E-02	2.210425E-02	-1.756189E-06	4.225573E-02	5.673452E-02	9.008757E-02	1.275462E-01	1.605500E-01
7.639565E-05	7.588760E-05	2.848159E-02	2.410771E-01	2.431674E-01	2.433655E-01	2.434309E-01	2.434373E-01
1.167315E-08	2.300260E-07	2.361270E-01	2.980245E-01	2.992196E-01	3.024931E-01	3.169305E-01	3.468703E-01
6.992510E-03	1.458776E-02	2.980245E-01	6.275526E-01	7.287487E-01	9.879669E-01	1.130709E+00	1.220536E+00
1.971313E-01	2.192041E-01	2.980245E-01	1.430418E+00	1.463114E+00	1.529452E+00	1.585941E+00	1.592574E+00
2.434170E-01	2.434128E-01	2.980245E-01	2.749240E-03	1.819672E-03	4.600991E-03	1.844894E-03	2.367660E-03
2.980000E-01	2.979990E-01	2.980245E-01					
4.251998E-01	5.071887E-01	6.275526E-01					
1.314973E+00	1.372683E+00	1.430418E+00					
1.6284770E+00	1.632729E+00	1.632729E+00					
4.0866761E-03	2.706042E-03	2.749240E-03					

INDEX = 0

# GLOSSARY

- $c_{pk}$  = specific heat of species  $k$ ,  $\text{cal-g}^{-1} - \text{K}^{-1}$ .
- $c_p$  = specific heat of the mixture,  $\text{cal-g}^{-1} - \text{K}^{-1}$ .
- $D_{km}$  = diffusion coefficient of species  $k$  in the mixture,  $\text{cm}^2 - \text{s}^{-1}$ .
- $h_k$  = specific enthalpy of species  $k$ ,  $\text{cal-g}^{-1}$ .
- $k_j$  = rate constant for reaction  $j$  in centimeter - mole - second units.
- $m_o$  = mass flux of the mixture through the origin,  $\text{g-cm}^{-2} - \text{s}^{-1}$ .
- $M_k$  = molecular weight of species  $k$ ,  $\text{g-mole}^{-1}$ .
- $N$  = number of chemical species (also number of PDE's).
- $NO$  = number of ODE's.
- $p$  = pressure, atmos.
- $r_j$  = rate of reaction  $j$ ,  $\text{mole-cm}^{-1} - \text{s}^{-1}$ .
- $R$  = gas constant =  $82.05 \text{ cm}^3 - \text{atoms-mole}^{-1} - \text{K}^{-1}$ .
- $R_k$  = rate of production of species  $k$ ,  $\text{mole-cm}^{-3} - \text{s}^{-1}$ .
- $S_v$  = velocity of the flame relative to the unburned mixture,  $\text{cm-s}^{-1}$ .
- $\hat{t}$  = temporal coordinate, s. ( $\hat{t}$ ,  $\hat{x}$  coordinate system).
- $\tilde{t}$  = temporal coordinate, s. ( $\tilde{t}$ ,  $\tilde{\psi}$  coordinate system).
- $t$  = non dimensional temporal coordinate, with respect to  $t_\infty$  ( $t$ ,  $\psi$  coordinate system).
- $\hat{T}$  = temperature of the mixture, K.
- $\tilde{T}$  = temperature of the mixture, K.
- $T$  = non dimensional temperature of the mixture, with respect to  $T_\infty$ .
- $T_U$  = temperature of the unburned mixture.

# GLOSSARY (continued)

- $T_B$  = temperature of the burned mixture (adiabatic temperature).
- $v$  = fluid velocity of the mixture,  $\text{cm-s}^{-1}$ .
- $V_k$  = diffusion velocity of species  $k$ ,  $\text{cm-s}^{-1}$ .
- $\hat{x}$  = spatial coordinate, cm.
- $Y_k$  = mass fraction of species  $k$ .
- $Y_{kU}$  = mass fraction of species  $k$  in the unburned mixture.
- $Y_{kB}$  = mass fraction of species  $k$  in the burned mixture.
- $\epsilon_k$  = mass flux fraction of species  $k$ .
- $\lambda$  = heat conductivity of the mixture,  $\text{cal-cm}^{-1}\text{-s}^{-1}\text{-K}^{-1}$ .
- $\rho$  = density of the mixture,  $\text{g-cm}^{-3}$ .
- $\tilde{\psi}$  = transformed distance coordinate,  $\text{g-cm}^{-2}$ .
- $\psi$  = non dimensional transformed distance coordinate, with respect to  $\psi_\infty$ .

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